

FAST ION MASS SPECTROMETRY AND CHARGED PARTICLE
SPECTROGRAPHY INVESTIGATIONS OF TRANSVERSE ION ACCELERATION AND
BEAM-PLASMA INTERACTIONS

by

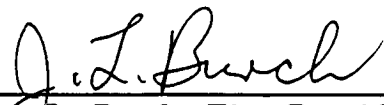
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FINAL REPORT
SwRI Project No. 15-7435
NASA Contract No. NAS5-27430

Prepared for
NASA Goddard Space Flight Center
Wallops Flight Facility
Wallops Island, VA

January 1987

Approved:


James L. Burch, Vice President
Instrumentation and Space Research

(NASA-CR-168348) FAST ION MASS SPECTROMETRY
AND CHARGED PARTICLE SPECTROGRAPHY
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Final Report (Southwest Research Inst.)

N87-22817

Unclas
G3/25 0071334

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1. SCIENCE OBJECTIVES

The principle scientific objective of this project is an investigation of ion acceleration transverse to the magnetic field in the topside ionosphere. Transverse acceleration is believed to be responsible for the upward-moving conical ion distributions (or "conics") commonly observed along auroral field lines at altitudes from several hundred to several thousand kilometers. Since these conics are observed in conjunction with ion cyclotron wave activity, the current theoretical understanding is that the ions are heated transverse to the magnetic field by VLF waves at harmonics of their gyrofrequencies, and then accelerated upward along the field by the magnetic mirror force. Such a mechanism is clearly important in the determination of the ultimate source of the magnetospheric plasma population (i.e. a terrestrial vs. a solar wind source).

Of primary concern in this investigation is the extent of these conic events in space and time. Theoretical predictions (e.g. Dusenbery and Lyons, 1981) indicate very rapid initial heating rates, depending on the ion species. These same theories (in concurrence with observations) predict that the events will occur within a narrow vertical region of only a few hundred kilometers. Thus an instrument with very high spatial and temporal resolution was required for this investigation; further, since different heating rates were predicted for different ions, it was necessary to obtain composition as well as velocity space distributions.

The FIMS instrument was designed to meet these criteria. To facilitate rapid scans, measurements were limited to those energies ($<2\text{keV}$) and ions (O^+ , NO^+ , H^+ , He^+) predicted to dominate the ion conic events (see Klumpar, 1979). High spatial resolution was further enhanced by the low speed and high telemetry rate of the sounding rocket (as compared with those of a satellite). The complete measurement objectives are given below:

Energy Range.....	1 eV/q to 2 keV/q
Ion Species.....	H^+ , He^+ , O^+ , NO^+
Pitch-Angle Resolution.....	$\Delta\alpha = 1^\circ$
Pitch-Angle Range for E/q Measurements.....	$\alpha = 0^\circ - 180^\circ$
Pitch-Angle Range for M/q Measurements.....	$\alpha = 80^\circ - 130^\circ$
Time Resolution for complete (E/q, α) Distribution.....	1.1 s
Time Resolution for complete (E/q, M/q, α) Distribution.....	6.6 s

2. INSTRUMENT DESCRIPTION

2.1 The Analyzers

Plate 1 is a photograph of the FIMS with an MCP detector mounted on the rocket deck plate.

The FIMS instrument consists of two pairs of spherical section conducting plates ($R_{in} = 28.9\text{mm}$, $AR_{in} = 3.02\text{mm}$; $R_{out} = 37.6\text{mm}$, $AR_{out} = 3.88\text{mm}$) acting as a dual-channel energy filter, followed by a cylindrical dual-channel ($R_{in} = 73.7\text{mm}$, $AR_{in} = 7.4\text{mm}$; $R_{out} = 82.2\text{mm}$, $AR_{out} = 8.2\text{mm}$) ExB mass analyzer, and two channeltron detectors. Figure 2-1 is a schematic view of the instrument. The entrance housing provides a baffle for off-angle trajectories. The electrostatic plates are noryl coated with conducting paint and covered with lamp blacking to reduce scattering. The magnet is SmCo with a field strength of 1900 gauss. The ExB analyzer operates at a bias of -815V with respect to the electrostatic analyzer.

The flight detector consisted of two (2) Amperex B413-BL channeltrons with a grounded grid in front of them. The biasing network and amplifier circuit (supplied by Mullard Space Science Labs) used with them are shown in Figure 2-2.

The detector and pre-amp section of Figure 2-2 consists of two (2) Amptek, Inc., Model A111 Hybrid Charge Sensitive Pre-Amplifier/Discriminator and Bias Networks packaged on a printed circuit board. A second circuit board houses an Amptek D400 Quad 8-bit Binary Counter and a 74HC244 Bus Driver.

An aluminum housing provides mounting for the two boards, as well as the two channeltrons. A 15-pin sub D connector provides the interface for the power and data lines.

2.2 Power Supplies

The programmable power supply (PPS) developed for the Fast Ion Mass Spectrometer is a new and totally different design from that used on previous programs, such as the Centaur I Sounding rocket. To provide a greater number of voltage steps and the versatility of programming for different instruments, two CMOS UV erasable PROMs were used as a lookup table addressed by the Central Electronics Package (CEP). The PROM output data provided input to a 10-bit D/A converter which performs as a staircase generator used to control the driver of the high-voltage transformer stage, as well as the control for the dynamic high-voltage shunt regulator.

Through a network of high-value resistors, a reference voltage is fed back to the control section to form a closed-loop system for better voltage regulation. Figure 2-3 is a block diagram of the PPS.

Isolation of the power supply secondary voltages is provided by transformer coupling in the input DC/DC converter, which provides the necessary digital and analog supply voltages. In the case of the mass PPS where the high-voltage return is at a float potential of as much as 3 kV, an additional P.C. card is provided which contains opto-isolators for the data and control lines, and a V/F - F/V circuit for the analog current monitors.

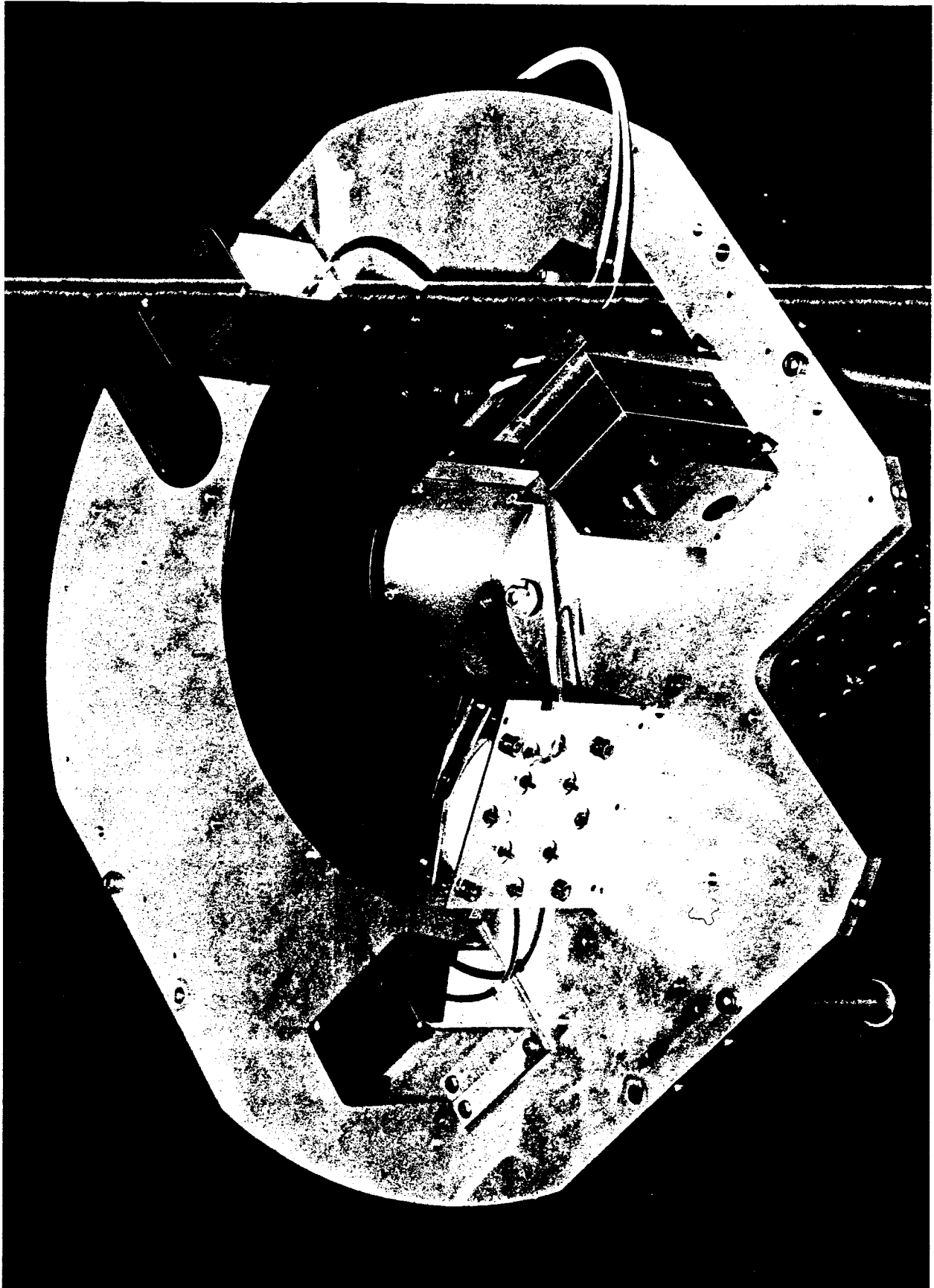


Plate 1. The FIMS Flight Unit in its Laboratory Configuration.

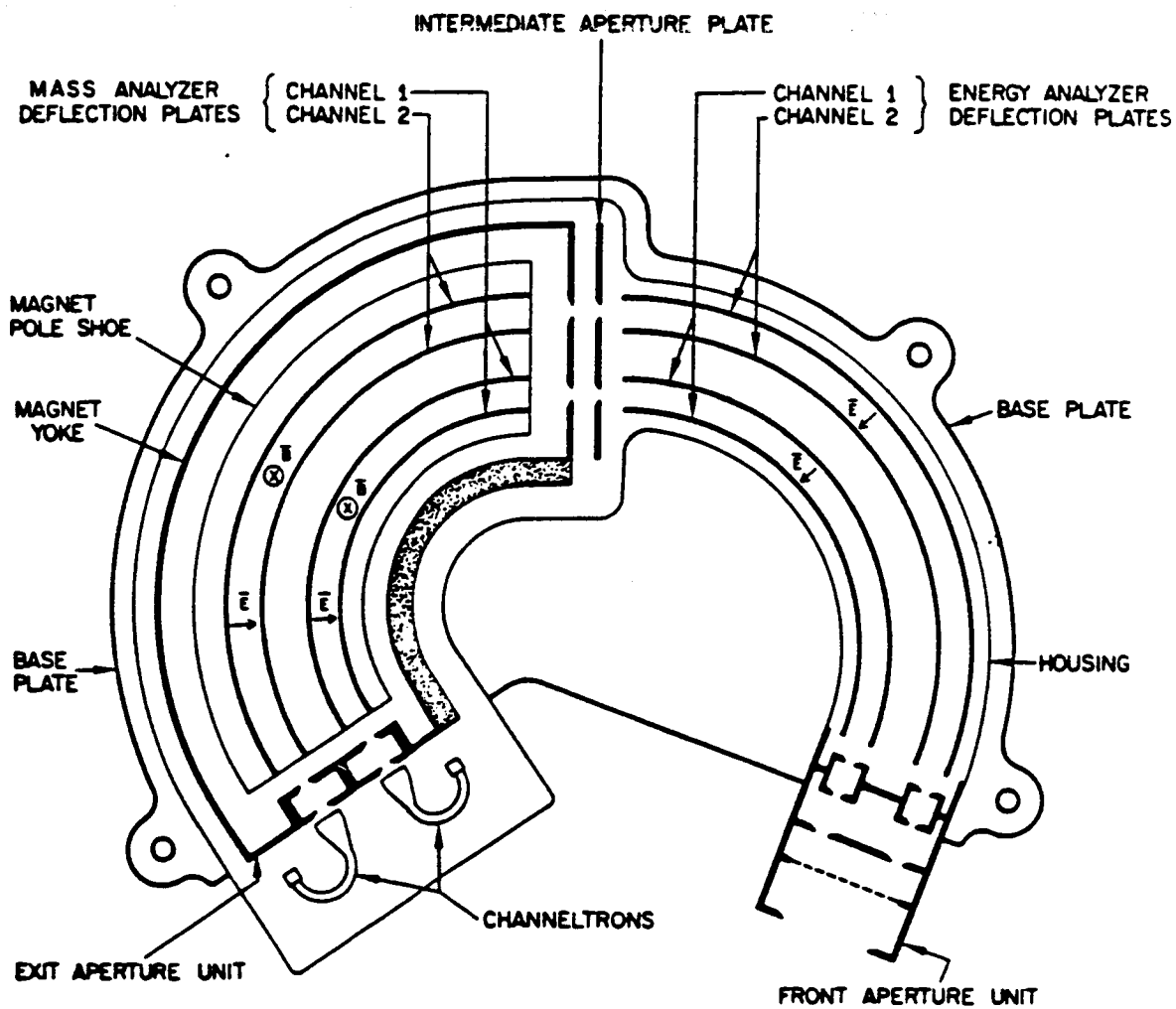


Figure 2-1 SCHEMATIC PRESENTATION OF THE DUAL-CHANNEL FAST ION MASS SPECTROMETER (FIMS).

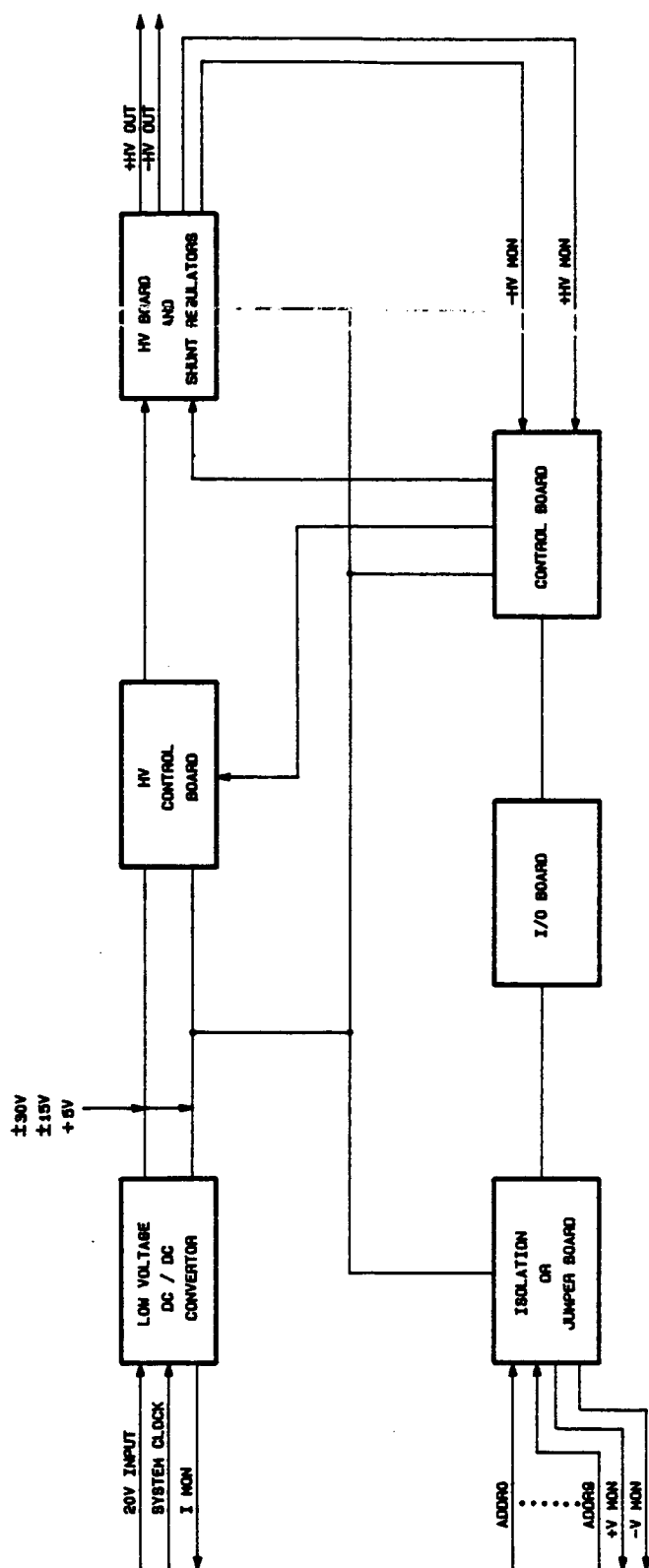


Figure 2-3 PPS Block Diagram

The fabrication of the power supply consisted of five separate epoxy glass P.C. boards and a single motherboard. The boards are divided into functional blocks to minimize circuit interconnects on the motherboard. For isolation, separate high-voltage connectors are used on the high-voltage sub-board to the motherboard where the high value resistor networks are located. The five boards are separated as follows: I/O board, Power Supply board, Control board, H.V. Control board, and H.V. board. In the isolated supply for the mass analyzer, an additional Isolation board is installed. A jumper board is installed for non-isolated operation. Figure 2-4 shows the component layout for the motherboard, and Figures 2-5 through 2-10 show the component layout for the six sub-boards.

The power supply is packaged in an aluminum housing, measuring 7 in. long x 4 in. wide x 2.5 in. deep. A 25-pin sub D connector provides the interconnections for the low-voltage control and supply. The high-voltage leads are routed through an epoxy glass insulator board and connected directly to the motherboard, eliminating exposed high-voltage terminals.

2.3 Central Electronics Package

The FIMS analyzer is controlled and monitored by a Central Electronics Package (CEP) shown in block diagram as Figure 2.3-1. The CEP is responsible for the generation of all Program Power Supply (PPS) commands as well as the acquisition of science data from the analyzer's detector assembly. Data acquired from the analyzer is formatted and relayed to the rocket's pulse code modulation (p.c.m.) telemetry subsystem at the appropriate time.

A 16 bit microprocessor with associated clock, memory and input/output circuitry is employed within the CEP. Circuitry for the CEP is contained on 4 plug-in printed wiring/stitchweld circuit boards all of which are housed in the single CEP enclosure. A detailed description of each of the 4 boards is contained in paragraphs to follow.

2.3.1 Central Processing Unit

Figures 2.3-2 through 2.3-6 are schematic diagrams of the FIMS CEP central processing unit (CPU). As can be seen in Figure 2.3-2, the CPU is controlled by an 80C86 microprocessor operating in the "minimum" configuration (without co-processors). The 80C86's clock and bus controller circuits can also be seen in Figure 2.3-2.

Since the 80C86 employs a multiplexed address/data bus it is necessary to de-multiplex the bus before it can be used to communicate with memory and I/O devices. Figures 2.3-3 and 2.3-4 show the manner in which the CPU bus is de-multiplexed within the FIMS CEP. The components used by the CEP are all complementary metal oxide semiconductor (CMOS) except for interface drivers.

In order to minimize the amount of circuitry needed by each of the 4 circuit boards for address decoding, a centralized I/O device decoding system, shown in Figure 2.3-6, is used by the CEP. To detect an I/O address, the system's 11 most significant address bits are compared to a preset values by 54HC688 octal comparators. When a true comparison is found, the system's

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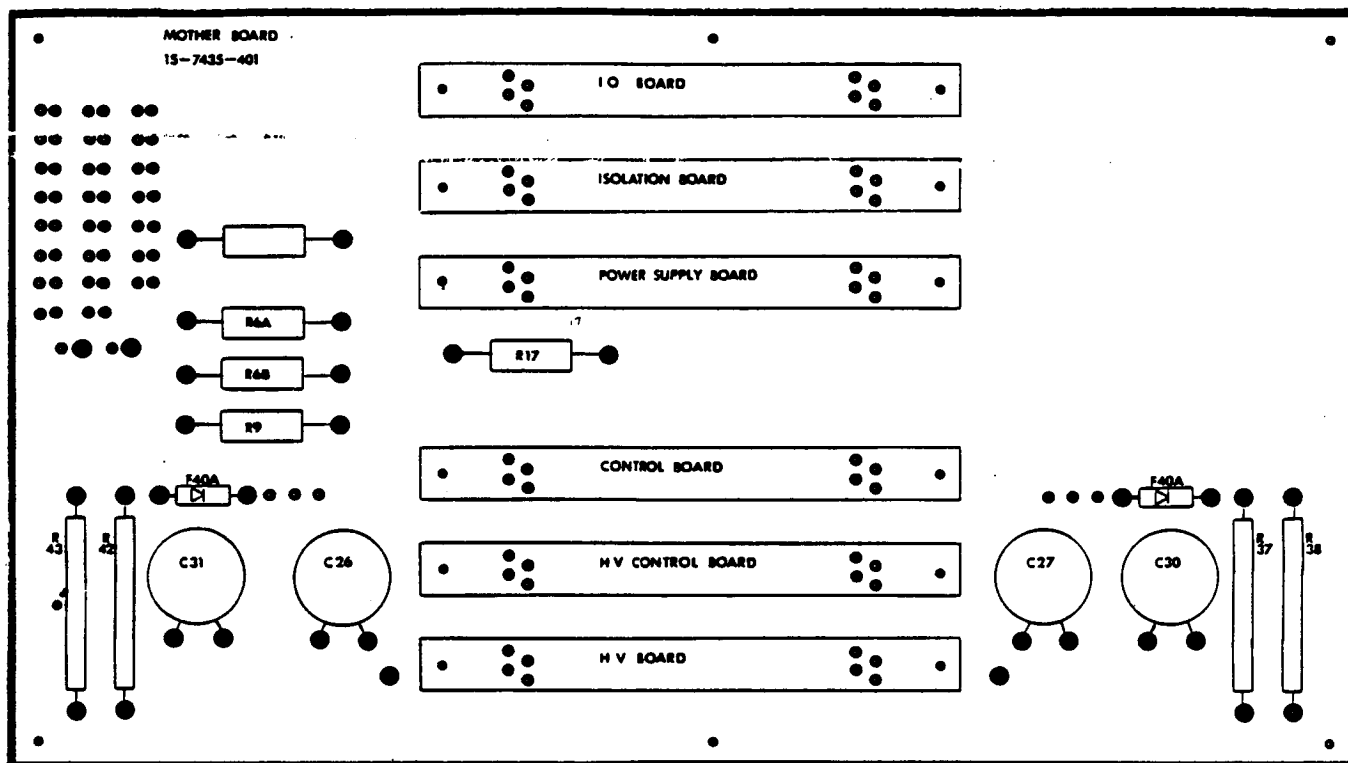


Figure 2-4 Component Layout of Motherboard

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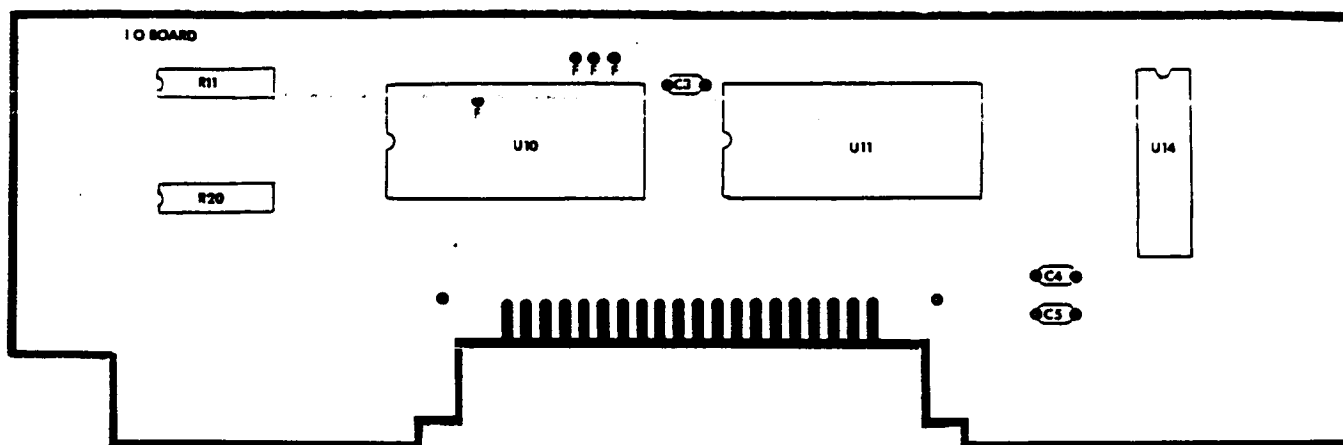


Figure 2-5 Component Layout of the I/O Board

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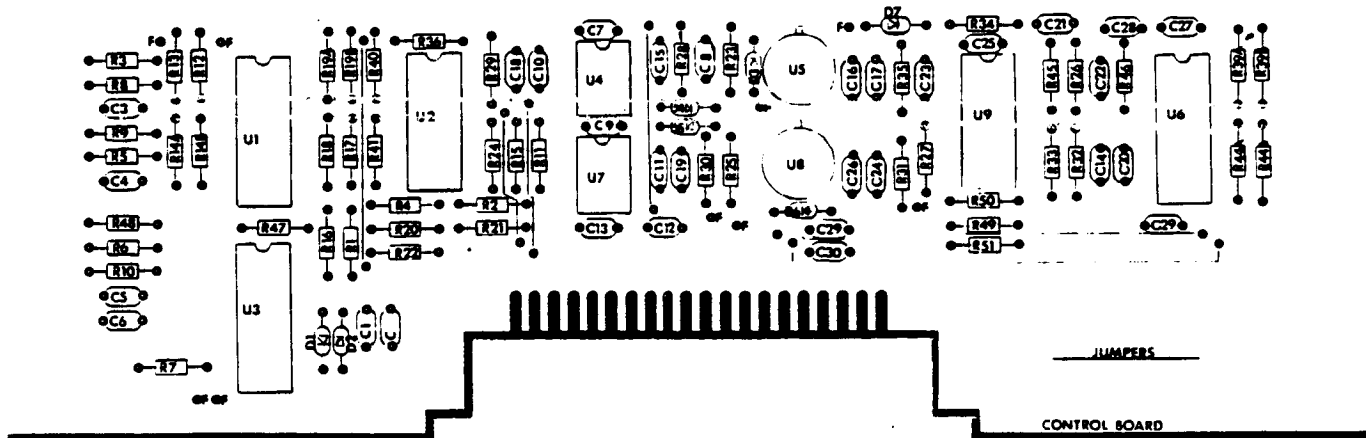


Figure 2-6 Component Layout of the Control Board

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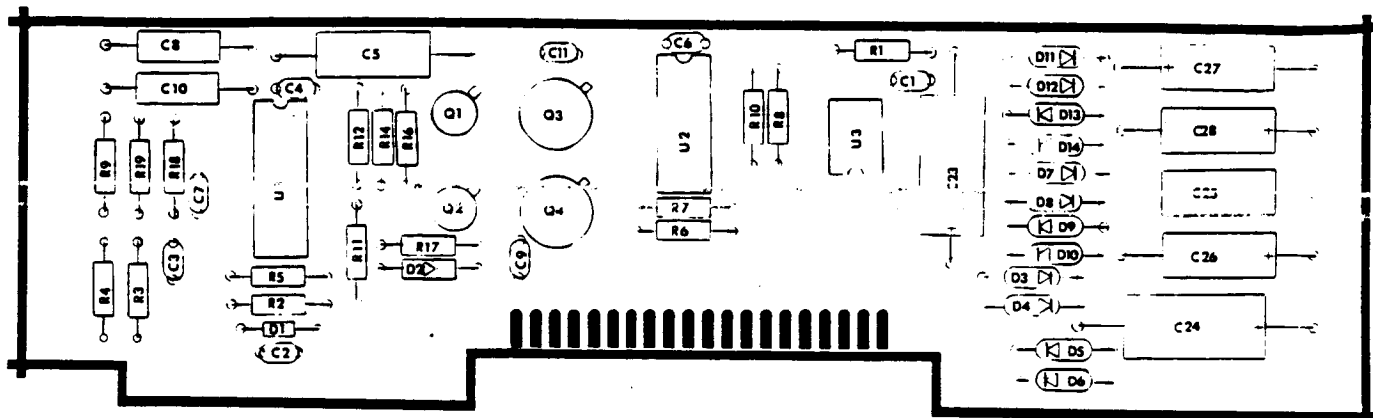


Figure 2-7 Component Layout of the H. V. Control Board

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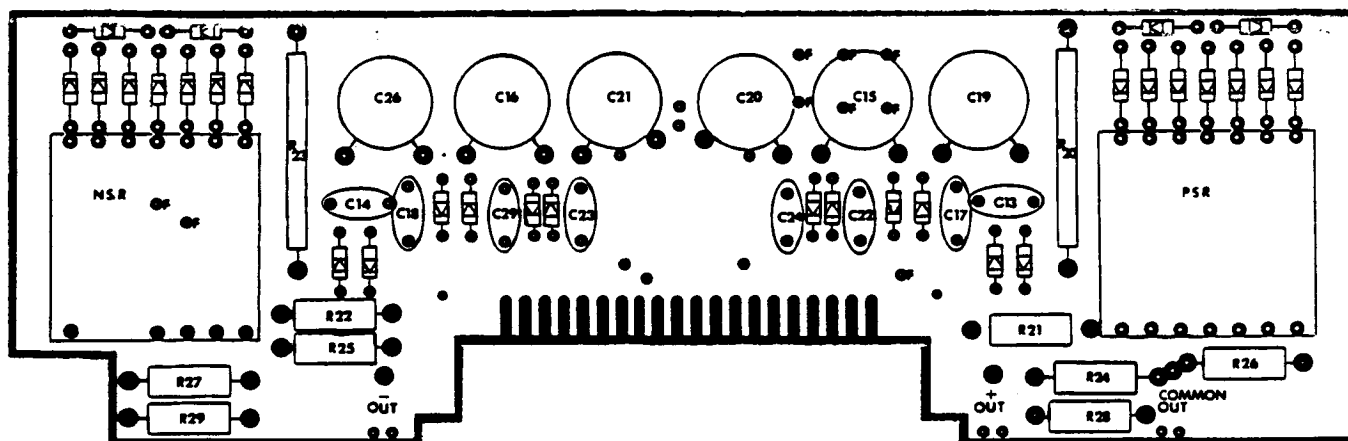


Figure 2-8 Component Layout of the H. V. Board

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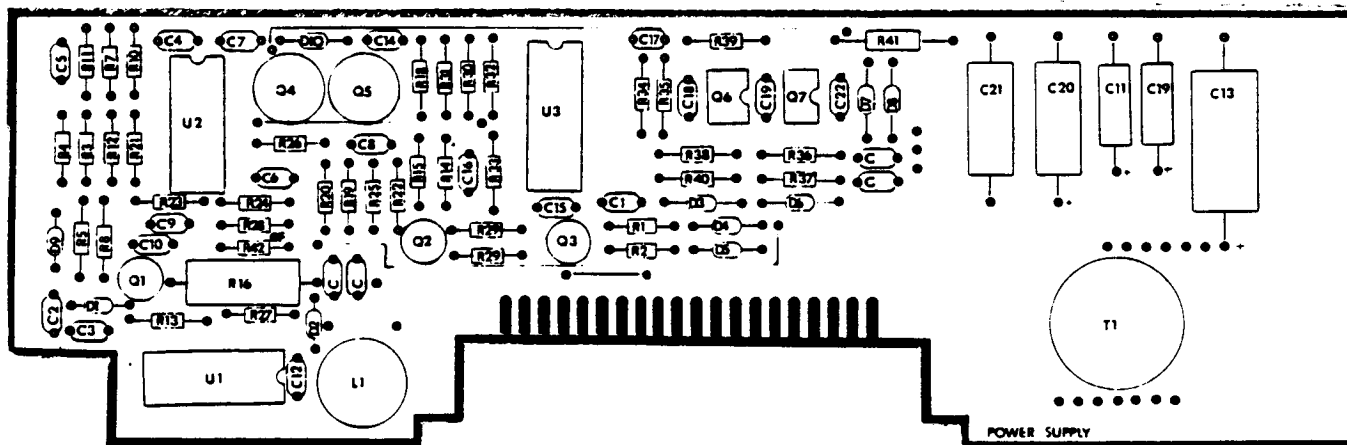


Figure 2-9 Component Layout of the Power Supply Board

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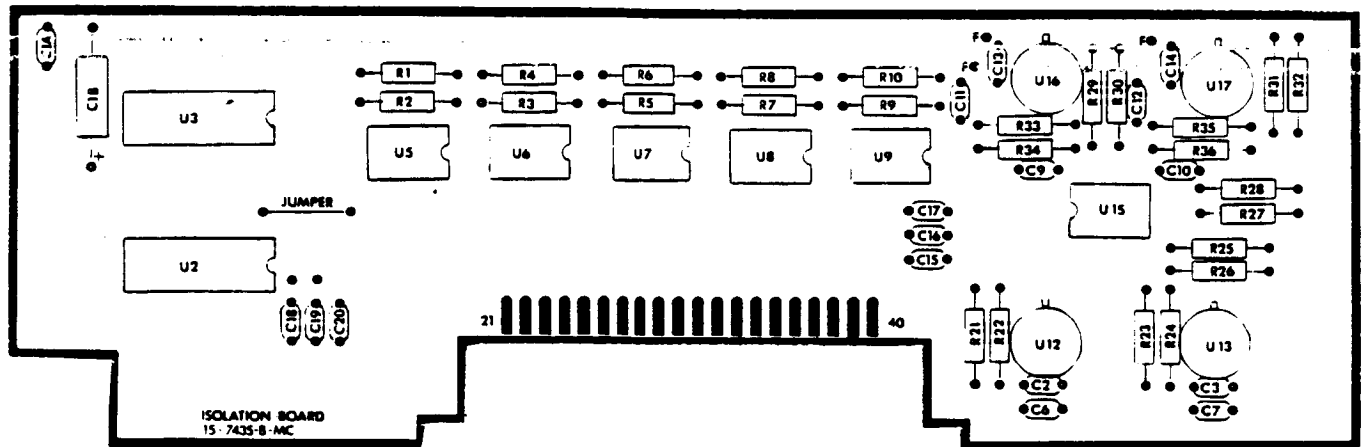


Figure 2-10 Component Layout of the Isolation Board

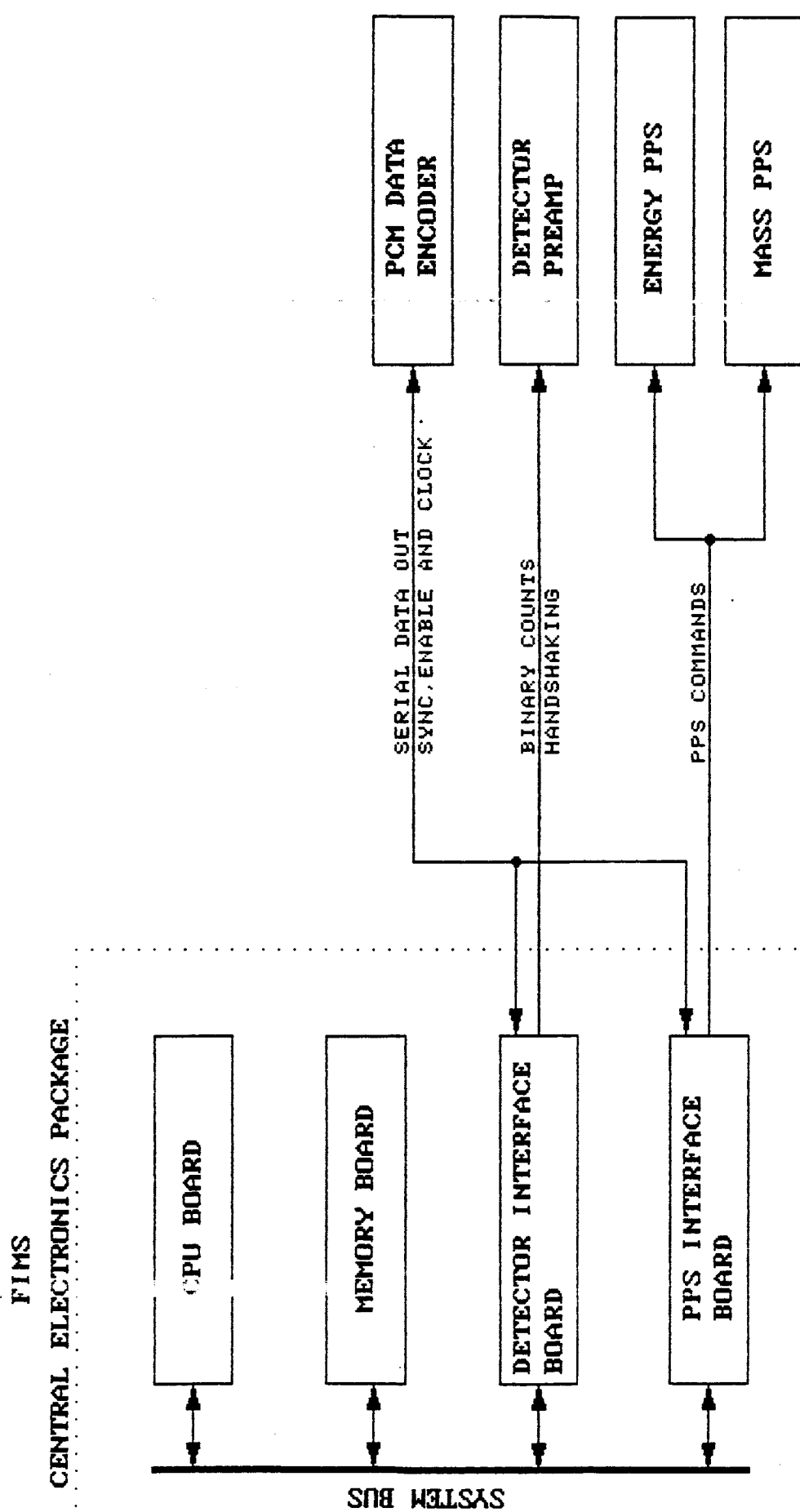
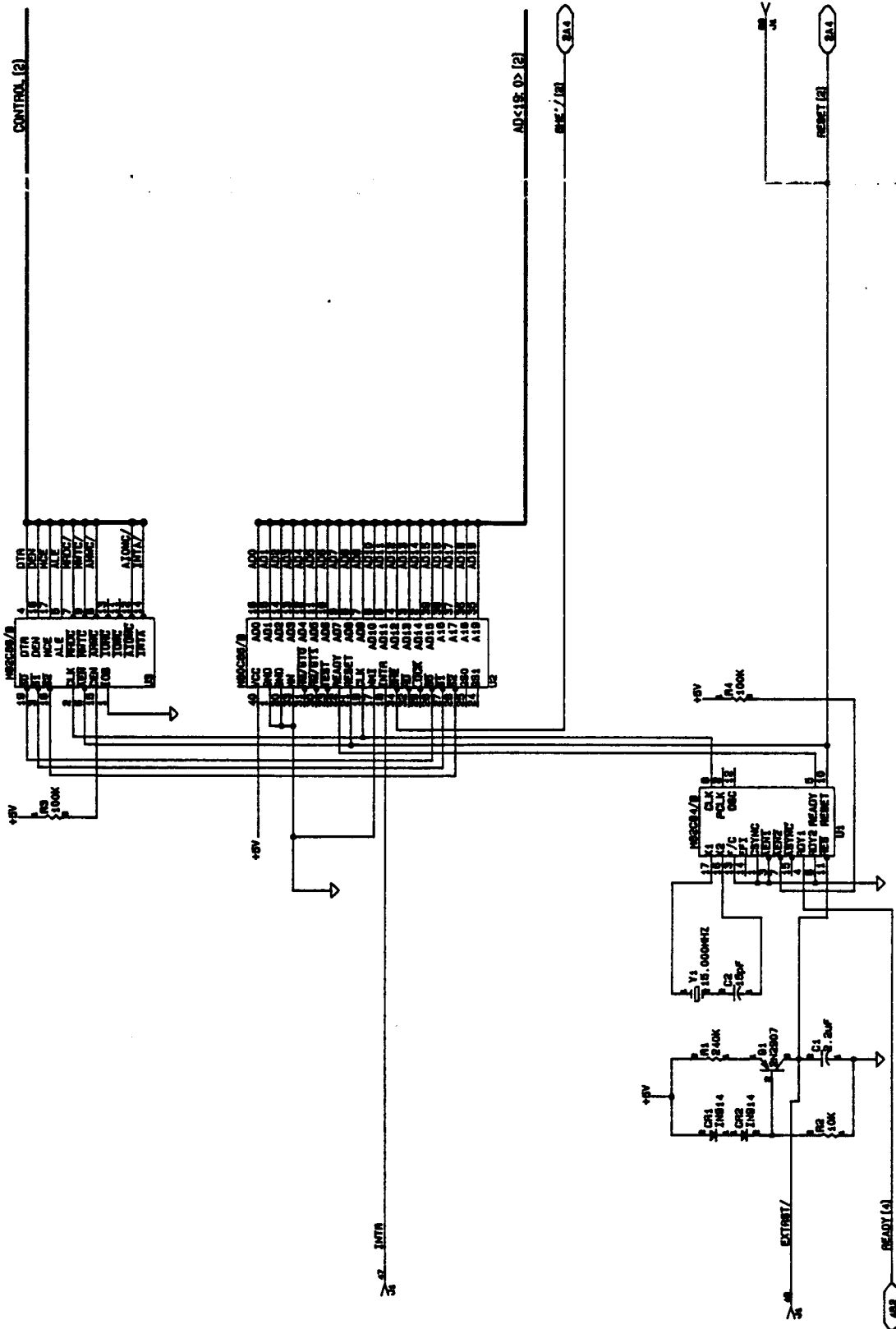


Figure 2.3-1 FIMS CEP Block Diagram

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FINS CENTRAL PROCESSOR CARD

Figure 2.3-2

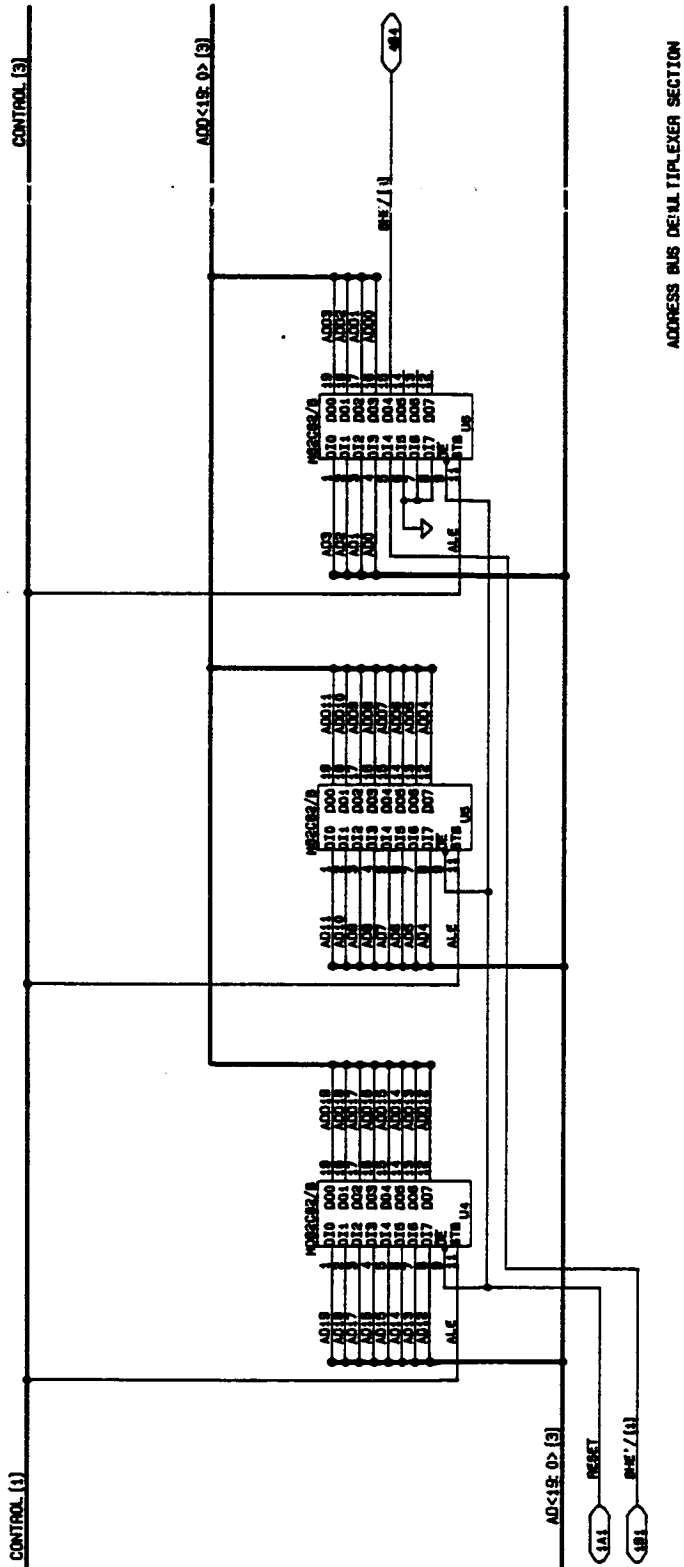


Figure 2.3-3

FMS CENTRAL PROCESSOR CARD

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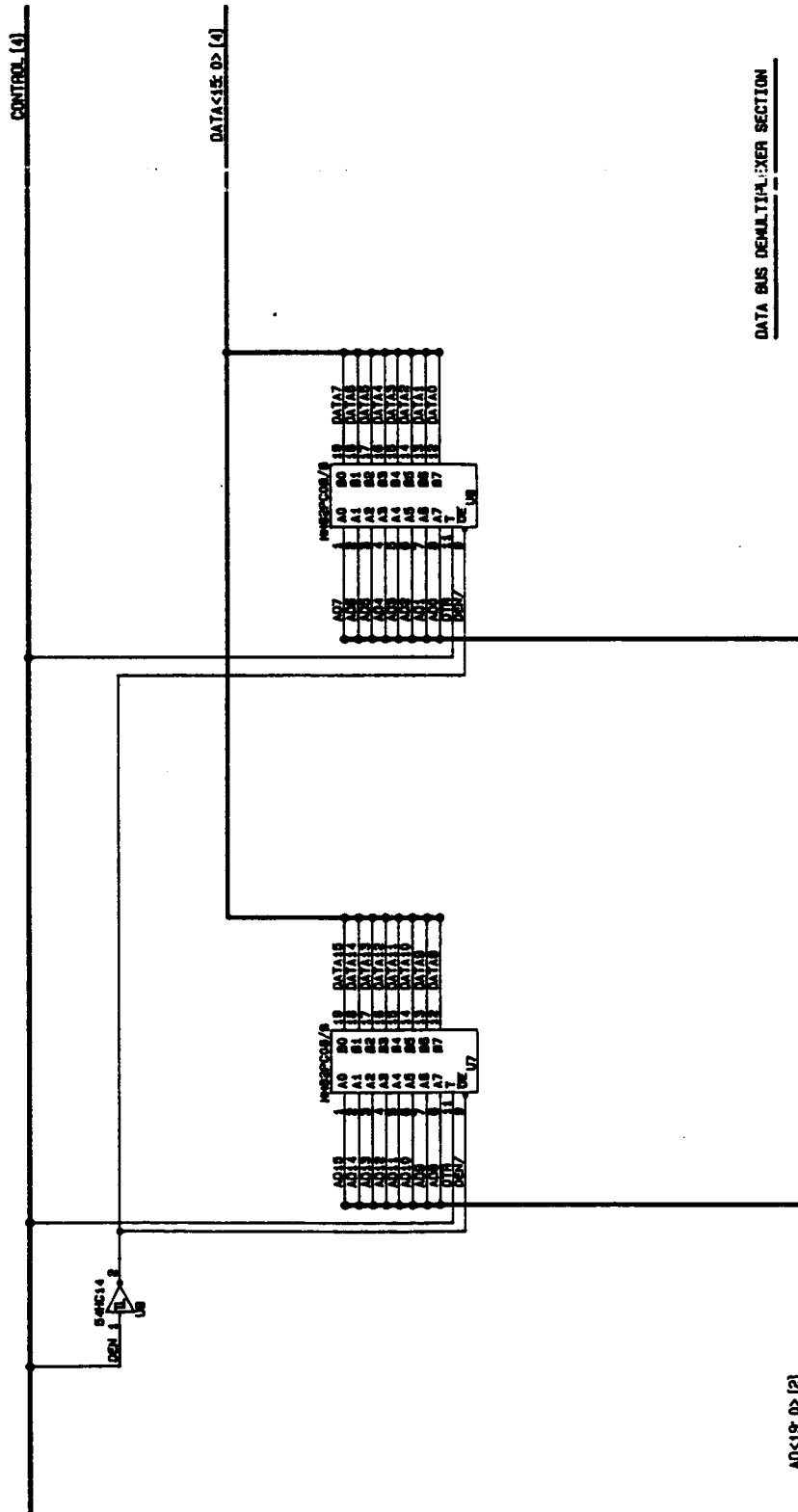


Figure 2.3-4

FMS CENTRAL PROCESSOR CARD

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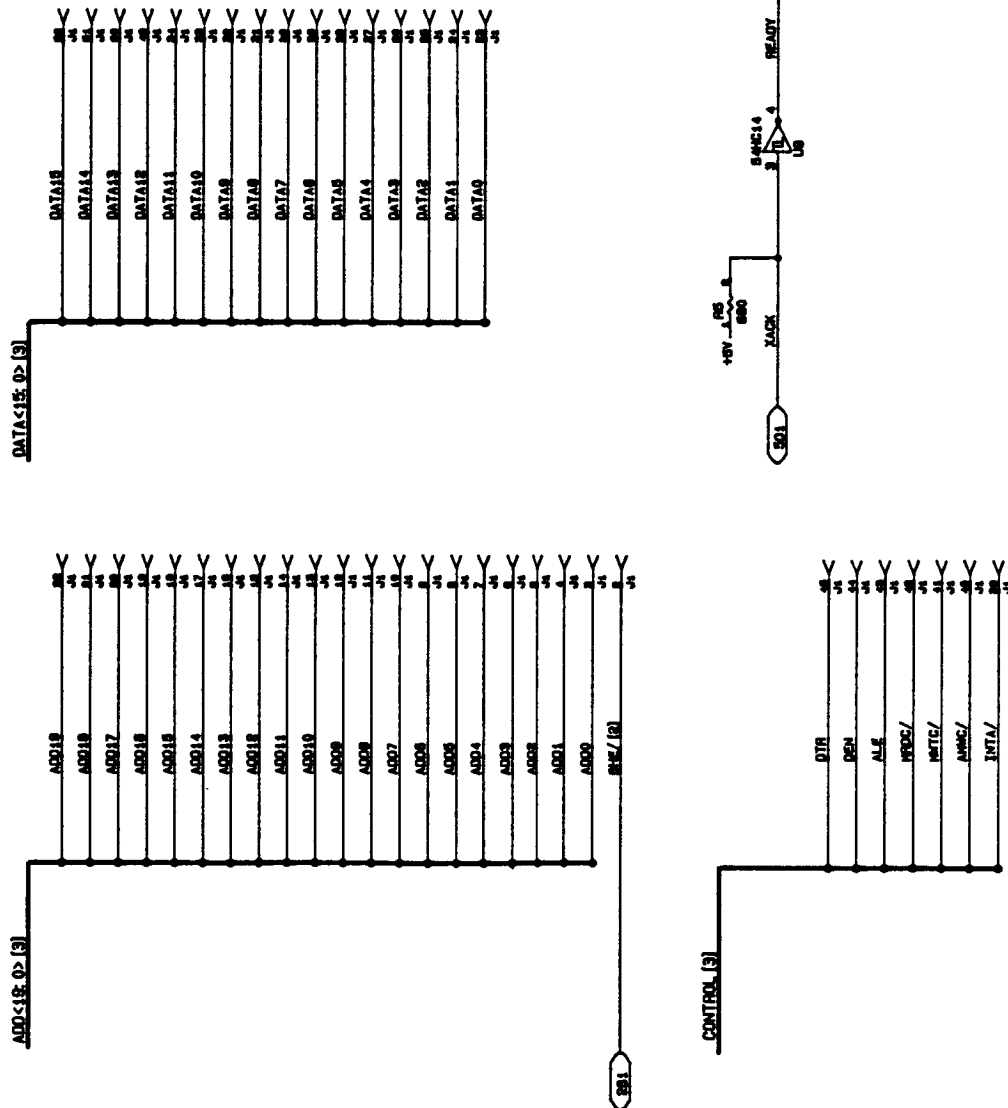


Figure 2.3-5

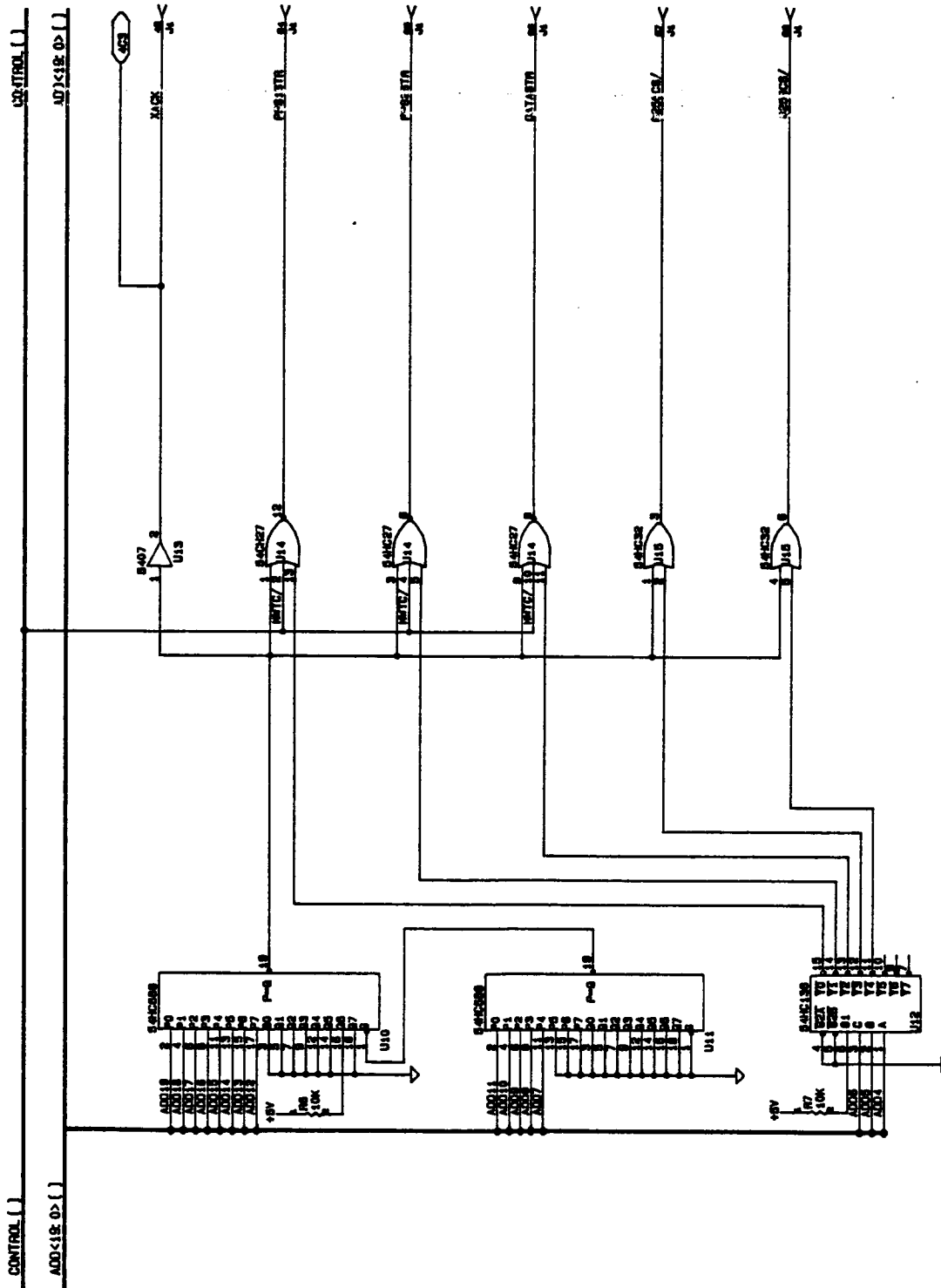
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FINS CENTRAL PROCESSOR CARD

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Figure 2.3-6



address bits add4-add6 are decoded in order to generate unique select lines for the individual I/O devices. Decoded addresses are then qualified against memory read and write control signal to produce the various strobe and enable signals needed for operation by the other 3 boards. Figure 2.3-6 shows the manner in which these several control signals are generated.

2.3.2 Control and Data Memory

Figures 2.3-7 through 2.3-9 show the circuitry used to provide the FIMS CEP with both control and data memory. As a media for storing software instructions, CMOS u.v. erasable/programmable read only memory (EPROM) devices are used. Between the two 27C64 EPROMs used in the CEP, a total of 16k bytes of program storage space is made available.

For data memory and program stack operations, CMOS static random access memory (RAM) is used. Figures 2.3-8 shows the circuitry used to produce a total of 4k bytes of ram for the CEP. Address recognition for the memory board is managed with 54HC688s as seen in Figures 2.3-7 and 2.3-8. Because of the short operating time of the FIMS instrument no attempt has been made to implement error detection/correction on control or data memory.

2.3.3 Program Power Supply (PPS) Interfaces

Figures 2.3-10 through 2.3-14 show the circuitry used to provide an interface between the CEP and the PPS's. Commands for the PPSs are latched in CMOS octal latches as shown in Figures 2.3-11 and 2.3-12. The strobe signals used by the latches to actually trap the PPS command words off of the system bus are produced on the CPU board itself as described earlier. Reference is made to the signal labeled "PPSiSTR" in Figure 2.3-11 as an example of a PPS command strobe signal.

To provide electrical drive capability to the octal latches storing the 10 bit command words used by each of the 2 PPS's, CMOS hex inverters are used. Examples of these interface buffers can be seen in Figures 2.3-11 and 2.3-12. It should be noticed that the logical interface to both of the PPS's is through the 10 least significant bits of the CPU's 16 bit data bus.

In addition to providing a logical and physical interface to the PPS's, the PPS interface board also provides command monitoring capability. Figures 2.3-13 and 2.3-14 show the digital circuitry used to latch the PPS command words into parallel/serial shift registers. The same strobe signal used to latch the PPS command into the appropriate output buffer is also used to latch the command into the input section of a 10 bit shift register.

The shift registers used for the 2 PPS command interfaces can be clocked out by the rocket's p.c.m. telemetry interface as required. Attention is called to the fact that interfaces to the rocket's telemetry system are true differential with all output signals driven by high current line drivers and all inputs optically coupled.

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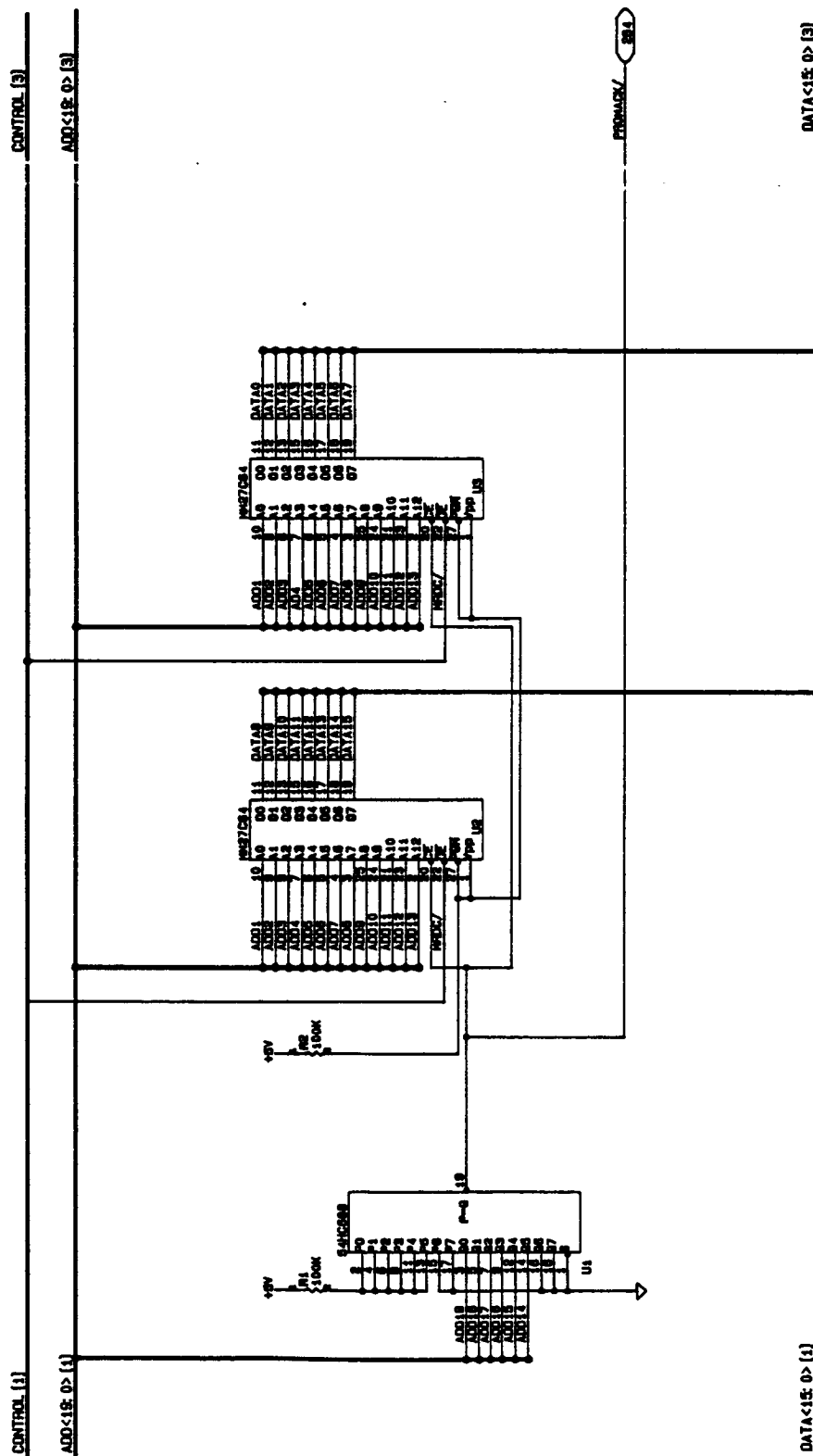


Figure 2.3-7

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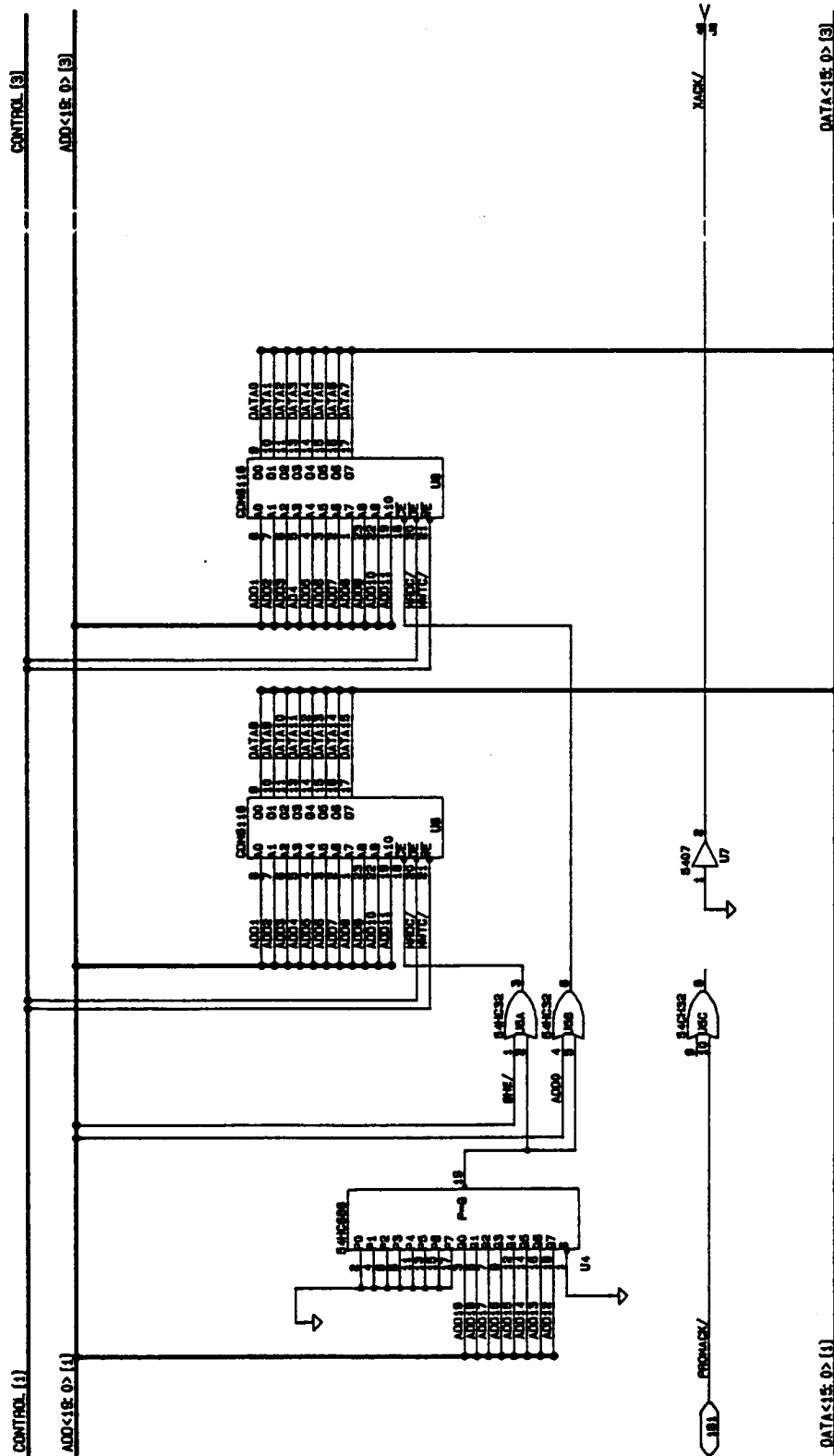


Figure 2.3-8

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FMS PROCESSOR MEMORY BOARD

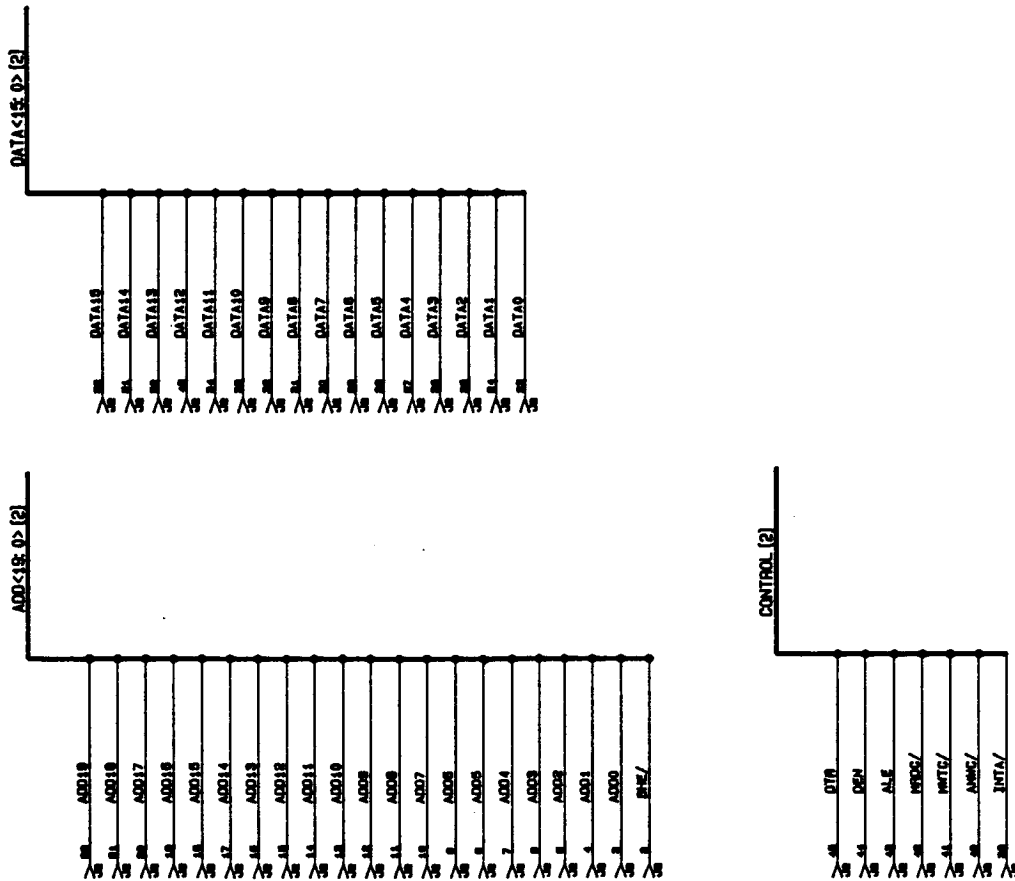
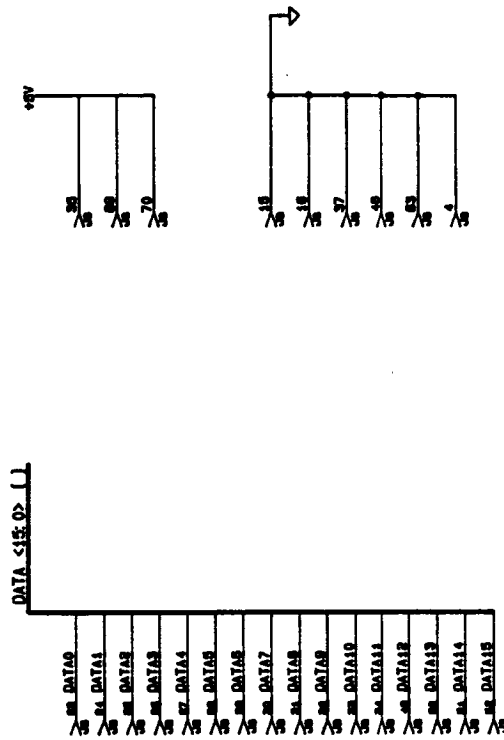


Figure 2.3-9



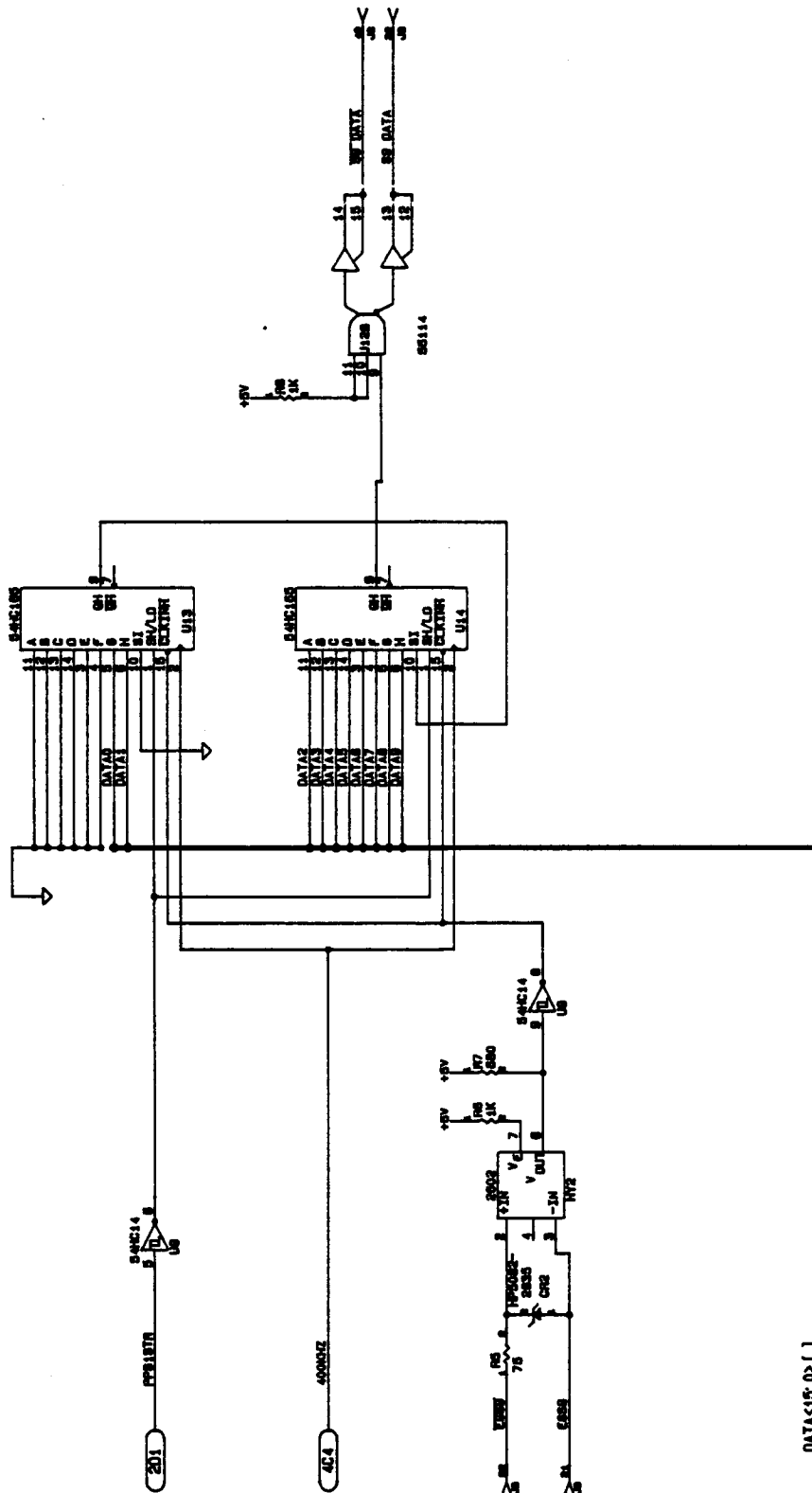
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Figure 2.3-10

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Figure 2.3-13



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Figure 2.3-14

2.3.4 Detector Interface

Shown on Figures 2.3-15 through 2.3-19 are the electrical schematics of the interfaces between the CEP and the analyzer's detector subsystem. Two Amperex B413-BL channel electron multipliers (CEM) (Section 2.1) with associated preamplifiers and a 2 channel CMOS hybridized binary counter are used in the detector subsystem. Ions exiting the E X B analyzer will be detected by the CEMs producing pulses which will in turn be counted by the binary counter mentioned above. The CEP acquires detector counts from this hybridized counter through an 8 bit parallel interface shown in Figures 2.3-14 and 2.3-15. A handshaking systems is employed to assure proper data transfer between the CEP and the detector subsystem. Figure 2.3-19 is a timing diagram of this handshaking.

The 80C86 microprocessor communicates with the detector interface through an 82C55 programmable peripheral interface (PPI) device. The handshaking scheme described above is carried out using port c output bits from the 82C55. Although somewhat slow, this approach to handshaking uses very few components.

2.3.5 Software Operation

As mentioned earlier, the FIMS instrument is controlled and monitored by a 16 bit CMOS microprocessor. A simplified flowchart of the software operation of the microprocessor is shown in Figure 2.3-20. The controller is completely interrupt driven, depending on timing interrupts generated by telemetry interface circuitry for operation. Minor frame (0.8 ms) and major frame (25.6 ms) rate interrupts synchronize the operation of the instrument to the data acquisition rate of the telemetry system.

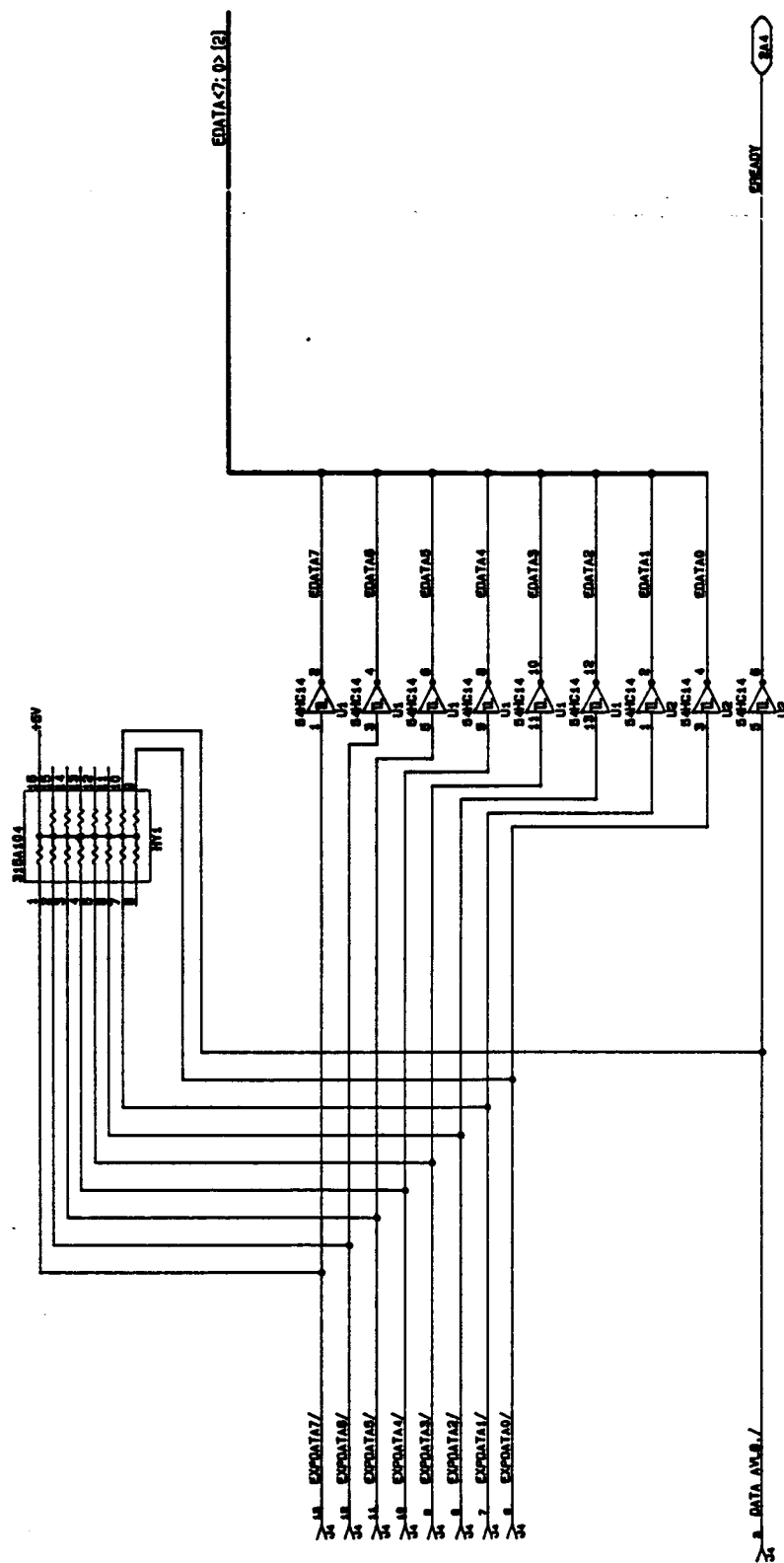
As each major frame rate interrupt is received, the microprocessor runs a software task which builds a pointer into a table of commands used by the energy analyzer's PPS. As seen in the flowchart, the process of building and transmitting energy PPS commands continues until a complete energy sweep ranging from 1ev/q to 2115ev/q has been completed. Normally, an energy sweep is completed in 1.1 seconds. Table 2.3-1 shows the energy levels visited during a normal energy sweep.

At the completion of an energy sweep a new ion species is selected (via mass PPS commands) for measurement and the energy analysis begun again. Table 2.3-2 shows the atomic masses of the species examined and the order in which they are sampled. When all species have been analyzed in the order shown in Table 2.3-2 the software recycles and begins the "normal sweep" again.

Because of the cross coupling between the energy/charge and mass/charge analyzers it is necessary to make minor corrections to the mass PPS setting for each new energy command. It is therefore oversimplified to think that in normal operation only the energy PPS is stepped each major frame.

Reference has been made to a "normal sweep" in the paragraphs above in order to differentiate between the standard sweep and the fine mass resolution sweep which is also programmed into the 80C86 software. The phrase

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FM3 PROCESSOR DETECTOR INTERFACE CARD

Figure 2.3-15

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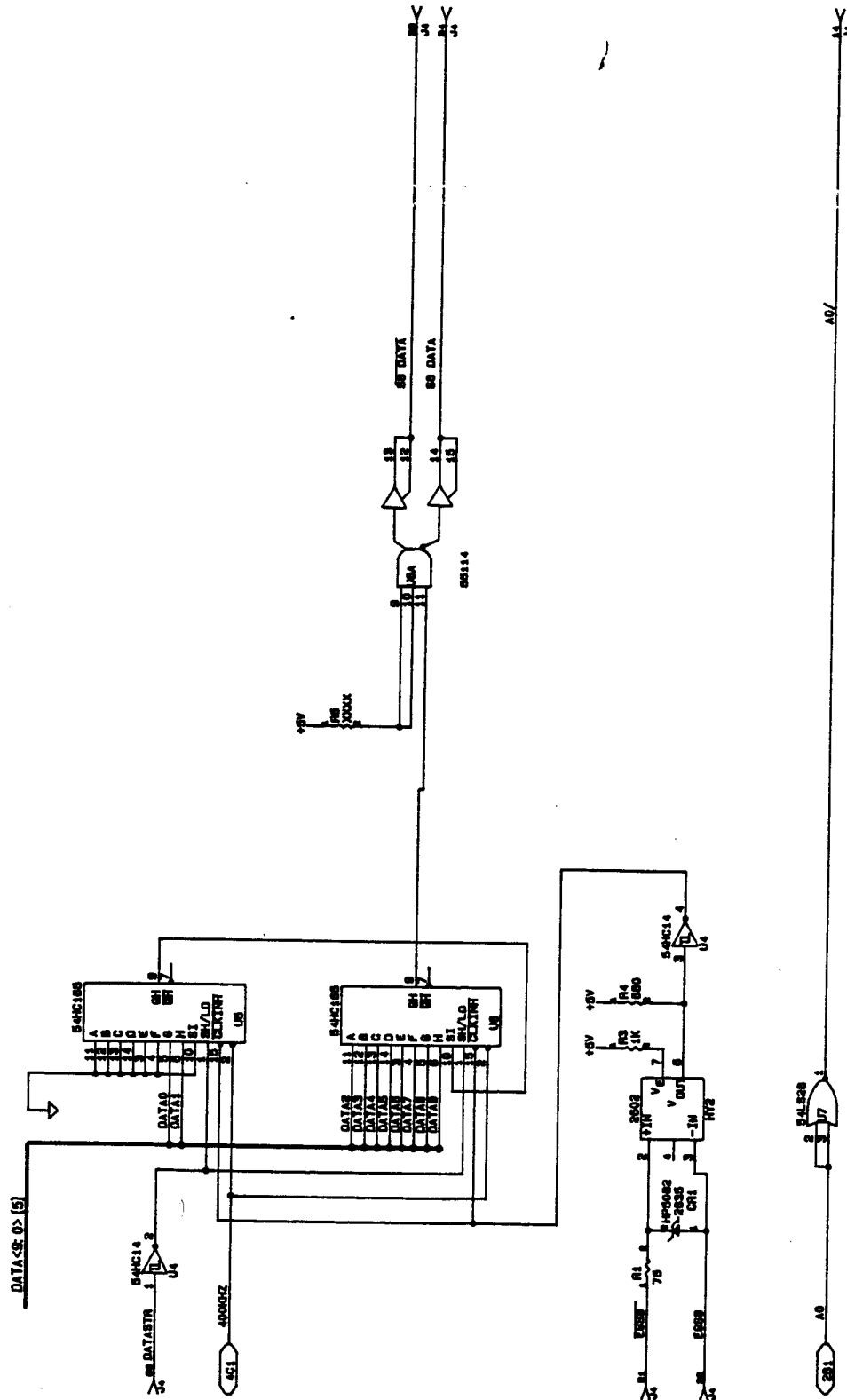
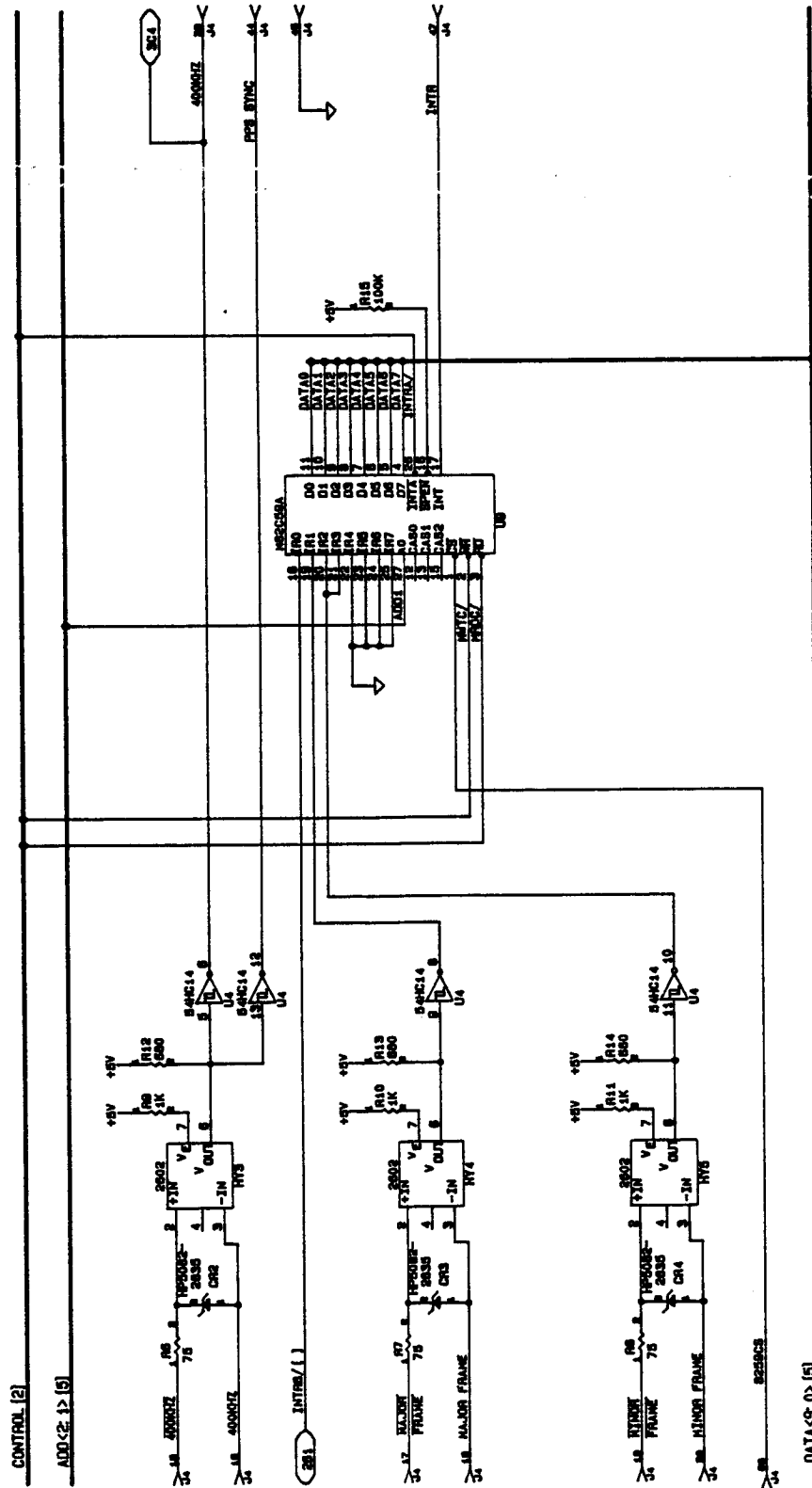


Figure 2.3-17



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Figure 2.3-18

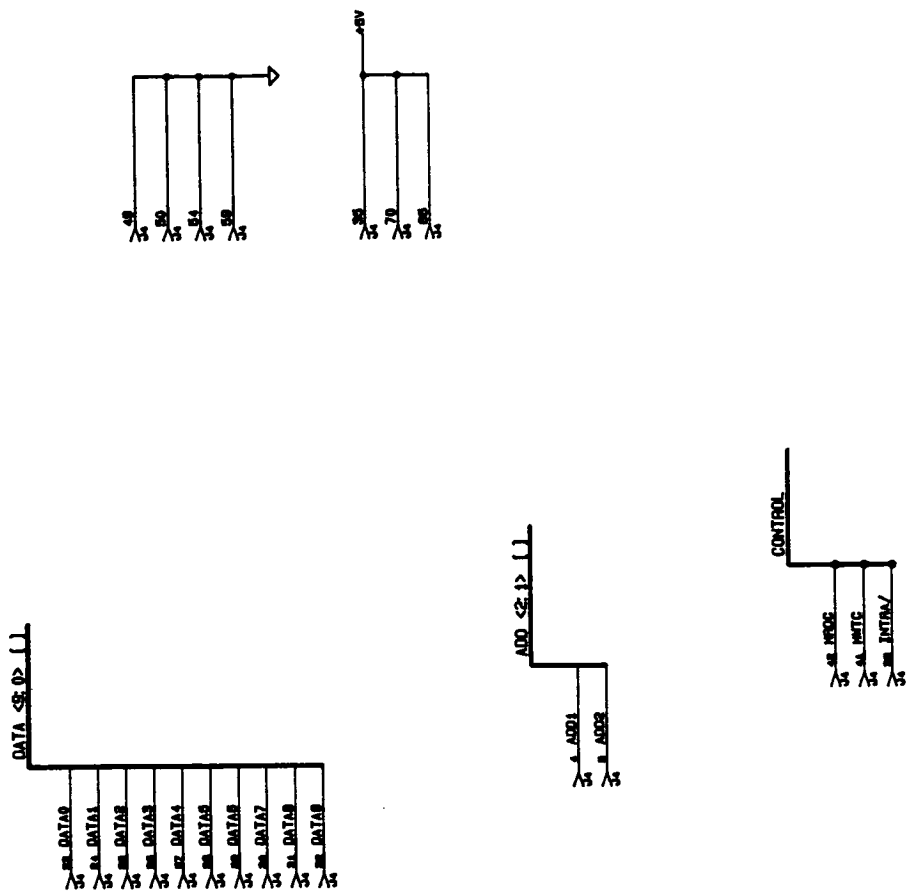


Figure 2.3-19

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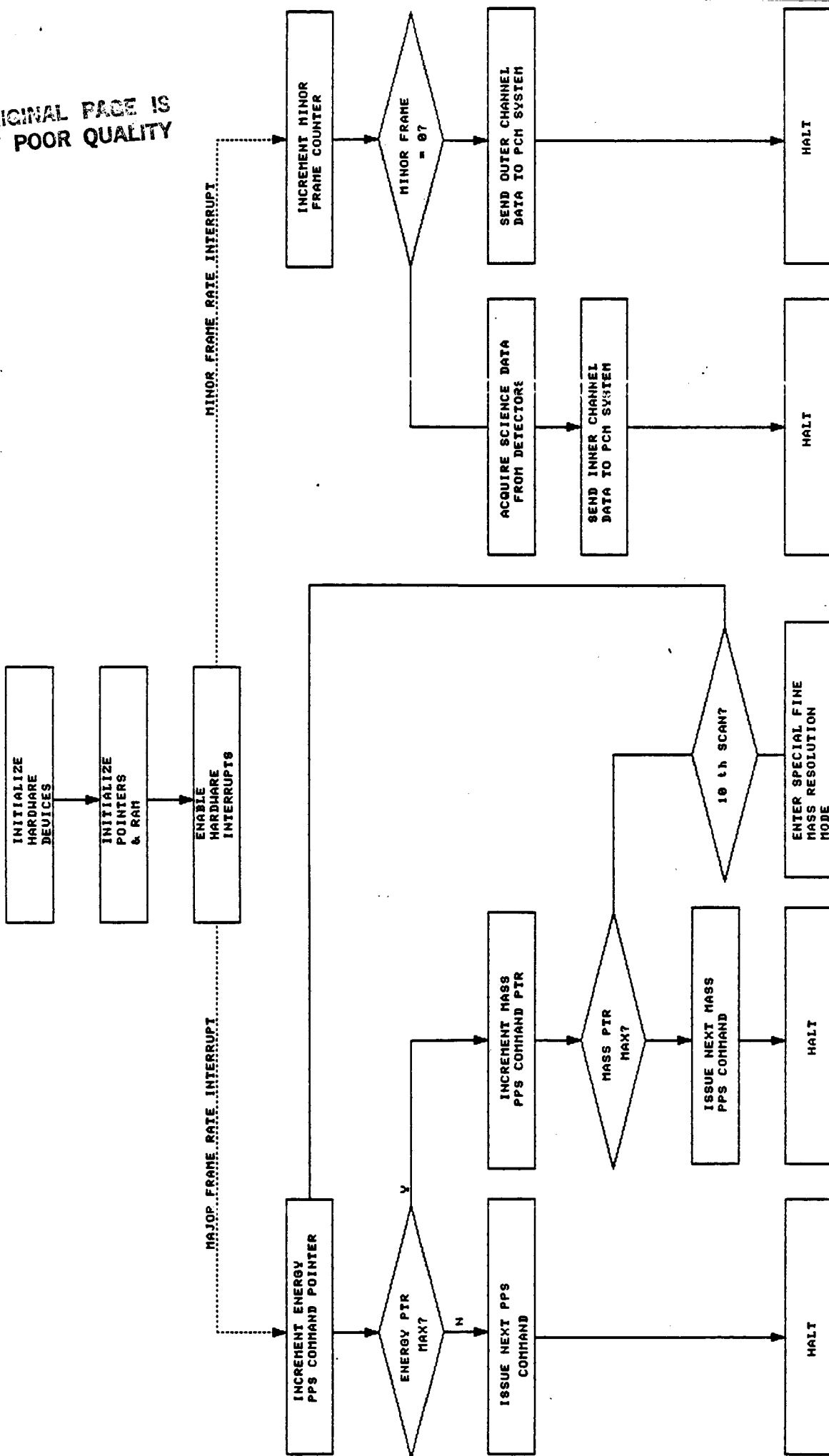


Figure 2.3-20. FIMS Flight Software Flow Chart

TABLE 2.3-1 ENERGY/CHARGE VALUES SAMPLED BY FIMS

STEP NUMBER	ENERGY/CHARGE SAMPLED (ev)
1	1.00
2	1.20
3	1.44
4	1.73
5	2.07
6	2.49
7	2.98
8	3.58
9	4.30
10	5.16
11	6.19
12	7.43
13	8.91
14	10.70
15	12.83
16	15.40
17	18.48
18	22.18
19	26.61
20	31.94
21	38.32
22	45.99
23	55.18
24	66.22
25	79.47
26	95.36
27	114.43
28	137.32
29	164.78
30	197.73
31	237.28
32	284.74
33	341.69
34	410.02
35	492.03
36	590.43
37	708.52
38	850.22
39	1020.27
40	1224.27
41	1469.18
42	1763.02
43	2115.62

TABLE 2.3-2 SPECIES SAMPLED BY FIMS

SPECIES	AMU
NO+	30
O+	16
H+	1
NO+	30
O+	16
He+	4

"normal sweep" is used to indicate an energy/mass set of commands which visits each of the ion species listed in Table 2.3-2 at each of the energy levels shown in Table 2.3-1. A normal sweep thus requires 6.6 seconds to complete. When 9 normal sweeps have been completed the FIMS software is programmed to enter the fine mass resolution mode. In this mode the energy PPS is held fixed for 9 major frame times while the mass PPS is commanded to the 4 closest command settings below and the 5 closest setting above the optimum mass PPS command for each species. In other words, the energy PPS is fixed and the mass PPS sweeps the 9 closest command settings to the optimal for the selected species.

The total number of major frames required to complete the fine mass scan mode is $42 \text{ (energy steps)} \times 9 \text{ (mass steps/energy steps)} \times 6 \text{ (species)}$ yielding a product of 2268 commands issued over 58.06 seconds. At the completion of the fine mass scan, the FIMS software resumes the normal scan mode.

A complete set of software listings for the FIMS CEP is contained in Appendix B of this document.

3. LABORATORY CALIBRATION

The FIMS instrument was first calibrated with laboratory electronics and detectors at the SwRI Ion Calibration Facility using hydrogen (H_2^+) and nitrogen (N_2^+) ions at energies from 100eV to 2 keV to test the inner and outer channels respectively. Contour plots, such as Figure 3-1, were used to confirm the analyzer constants for the two sections over the entire energy range. The plot shows voltage on the electrostatic analyzer (x axis) vs. voltage on the electric field plates in the ExB analyzer (y axis) vs. counts (shown as contours) for 2 keV N_2^+ . These data give an average dE/E of about 10%. Mass resolution can be demonstrated by taking the separation in voltage applied to the ExB analyzer between masses of interest compared to the spread in voltage for an individual mass. All ions of interest are clearly resolved.

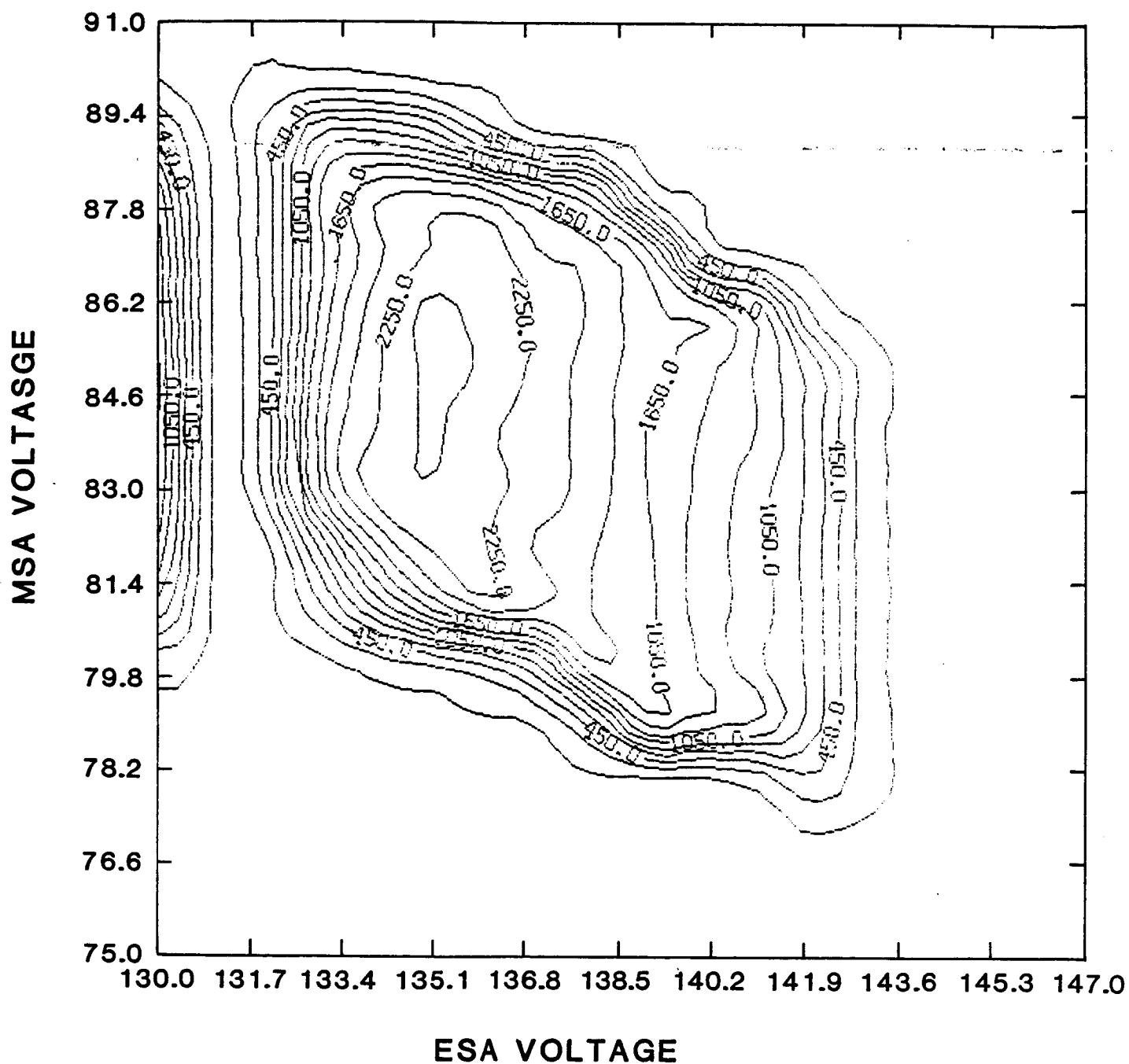
Scans of azimuthal and elevation throughput were also made to confirm the angular range of the instrument. Figure 3-2 shows the acceptance to be $\pm 3^\circ$ in azimuth and $\pm 12^\circ$ in elevation. Appendix C is a collection of plots of lab data.

A microchannel plate (MCP) detector was used in order to study exit z position of the particle trajectories vs. incoming ϕ angle, for reference in future missions in which use of an MCP might allow such correlations to be recorded yielding additional information about the pitch angle dependence of the conic events. These data are shown in Figure 3-3, plotted against $\tan\theta$ (solid line) which is the expected acceptance. We hope to investigate the discrepancy further with the next FIMS.

Finally, the flight power supplies (PPSs) and central electronics package (CEP) were integrated with the analyzer and the complete instrument was calibrated using an SC-1 Spacecraft Computer to simulate the rocket's communication buss. Figure 3-4 is a block diagram of the preflight verification configuration. Appendix D contains a table showing a listing of power supply settings for various energies and masses. Instrument performance in the final configuration was confirmed using several input ions.

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INNER CHANNEL H₂ 2 KeV



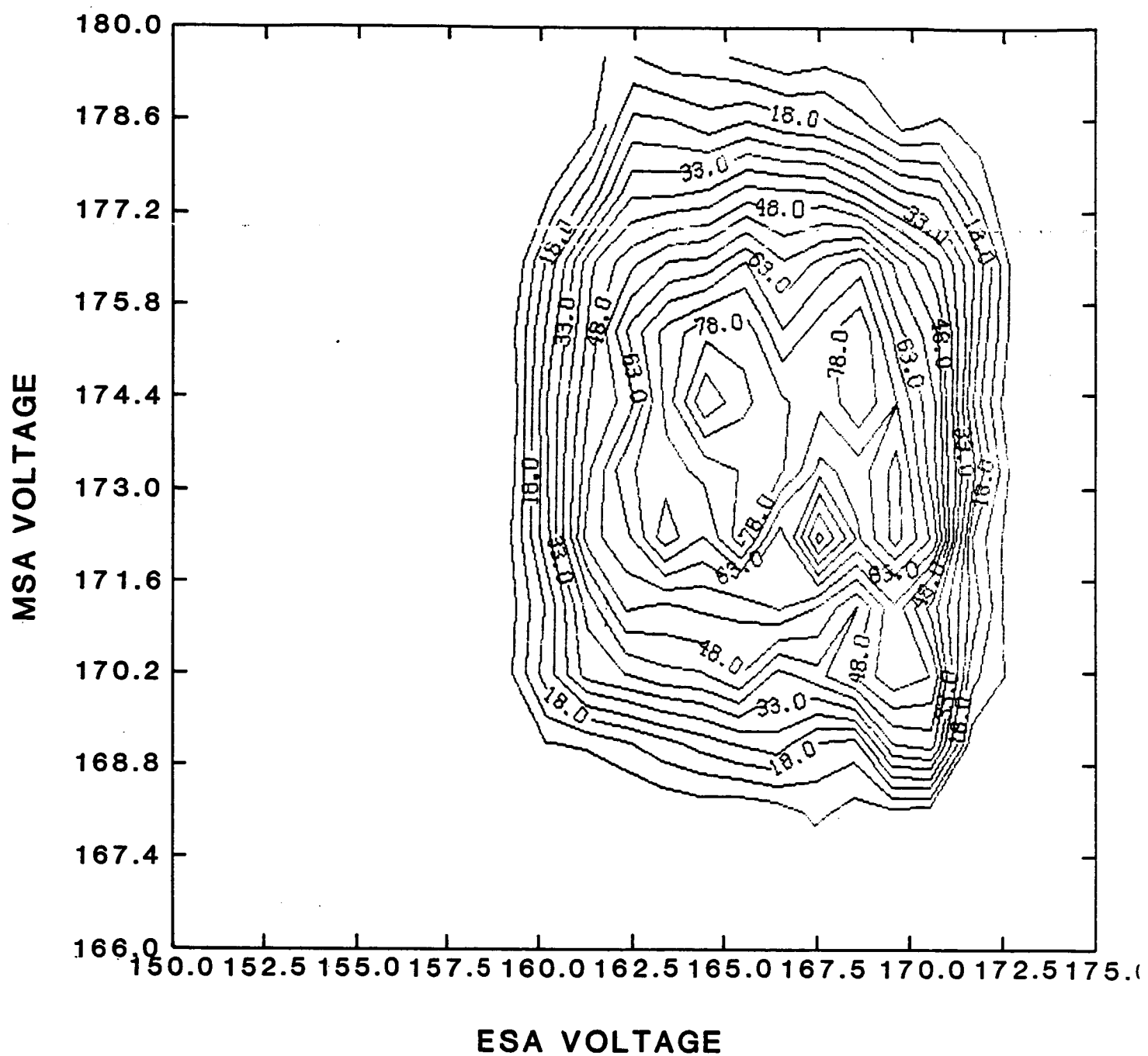
OUTER CHANNEL N₂ 2 KeV

Figure 3-1b

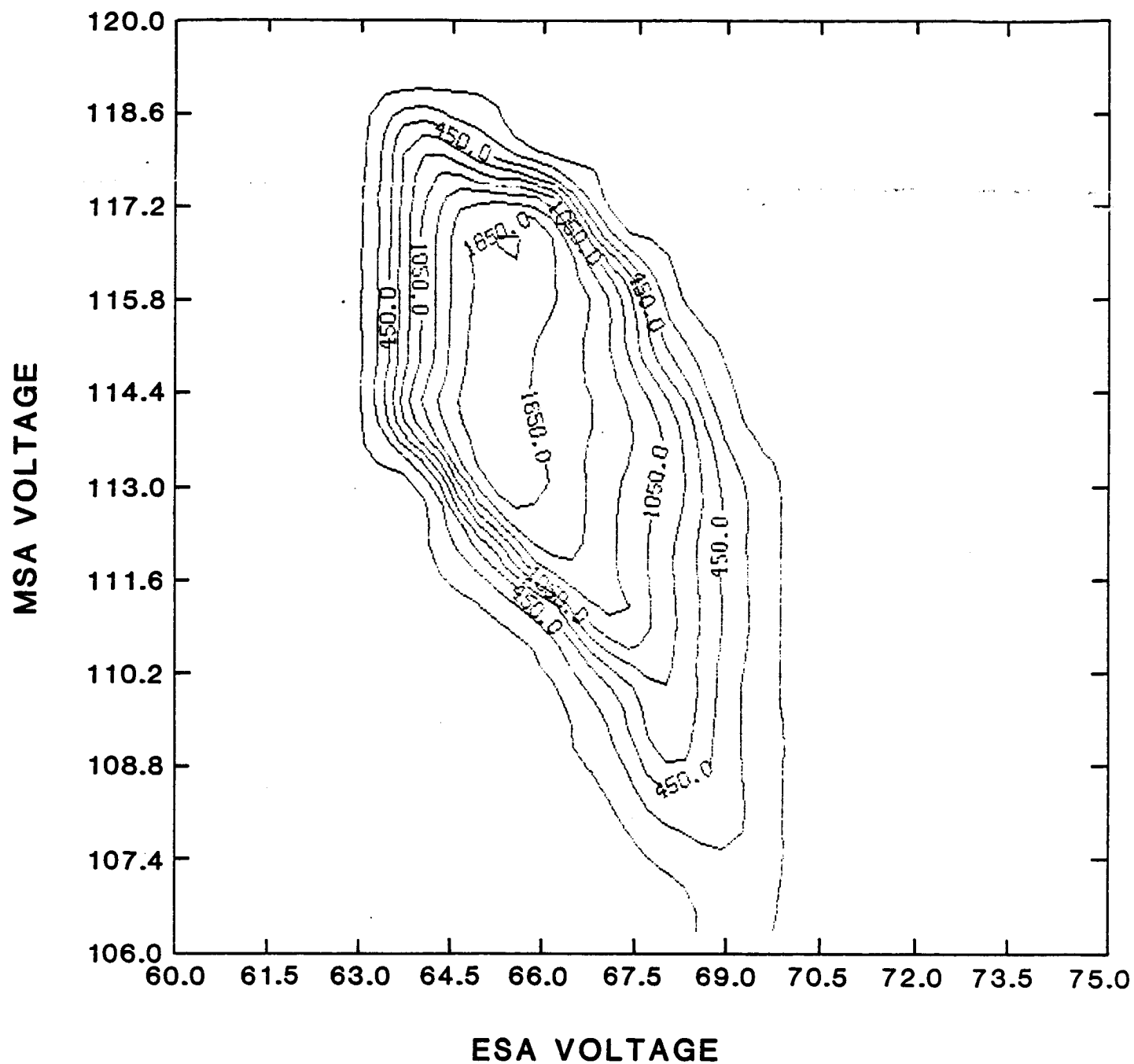
INNER CHANNEL H₂ 1 KeV

Figure 3-1c

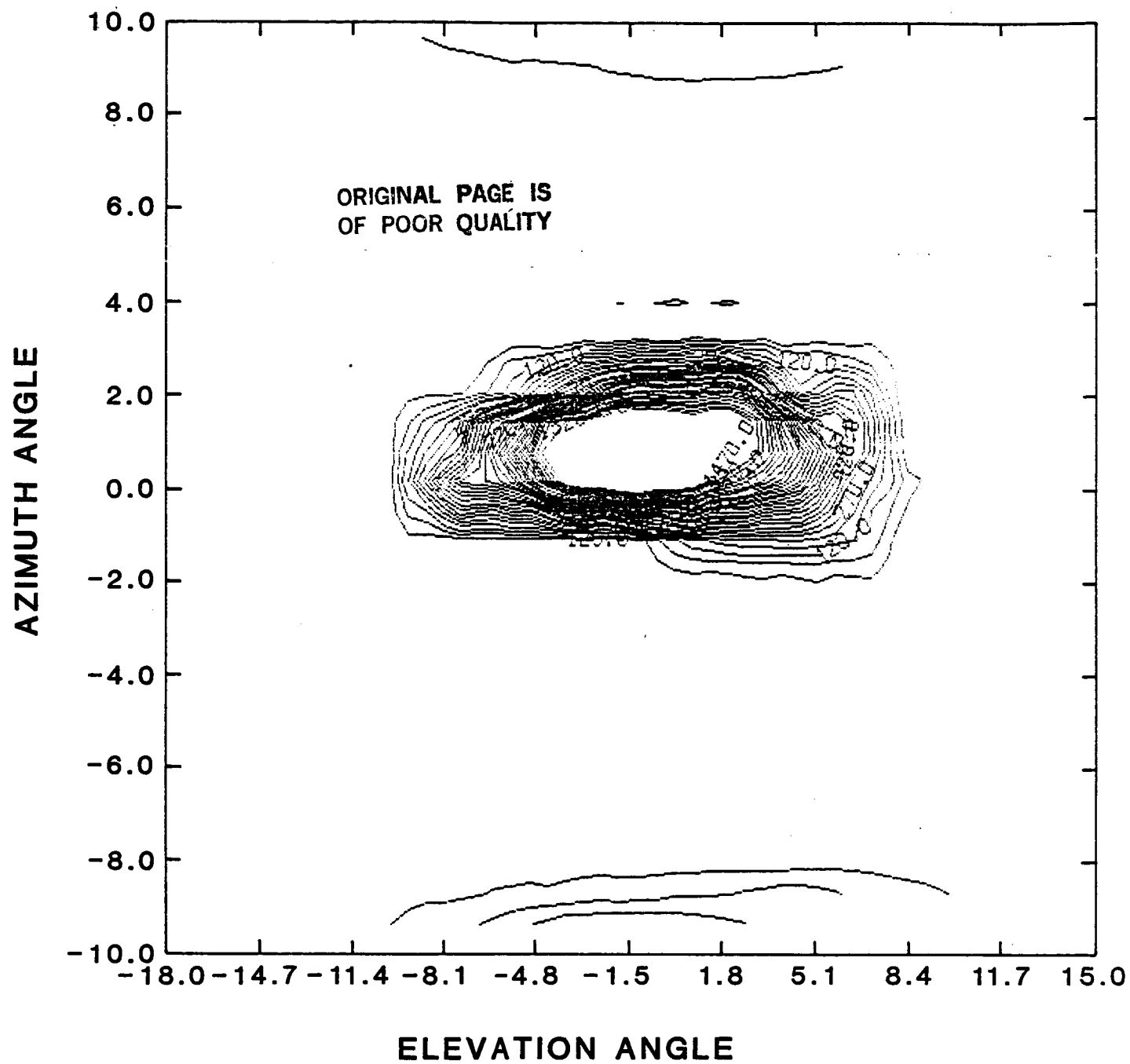
INNER CHANNEL H₂ 1 KeV

Figure 3-2a

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ANGLE SCAN N_2 2 KeV

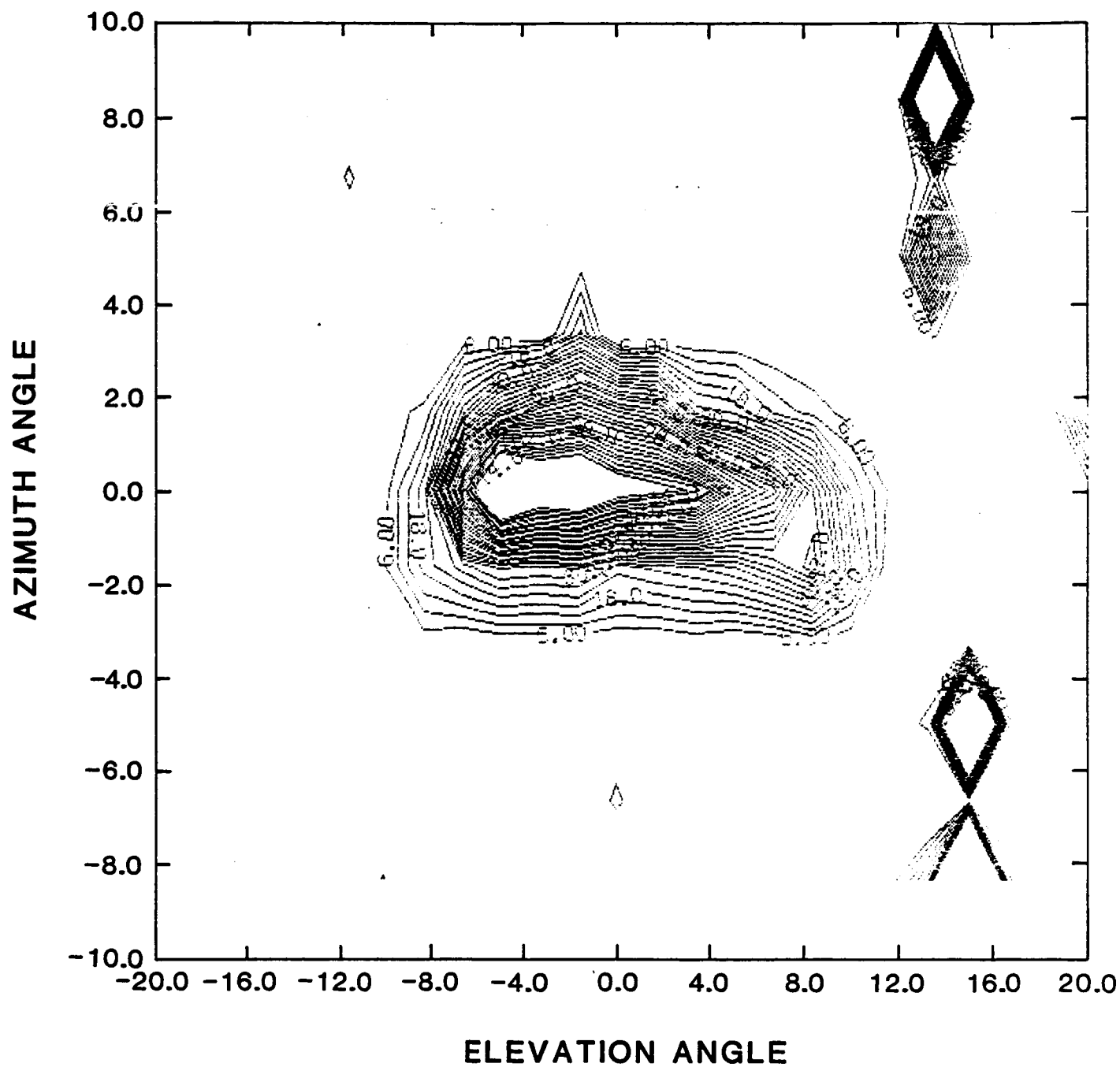


Figure 3-2b

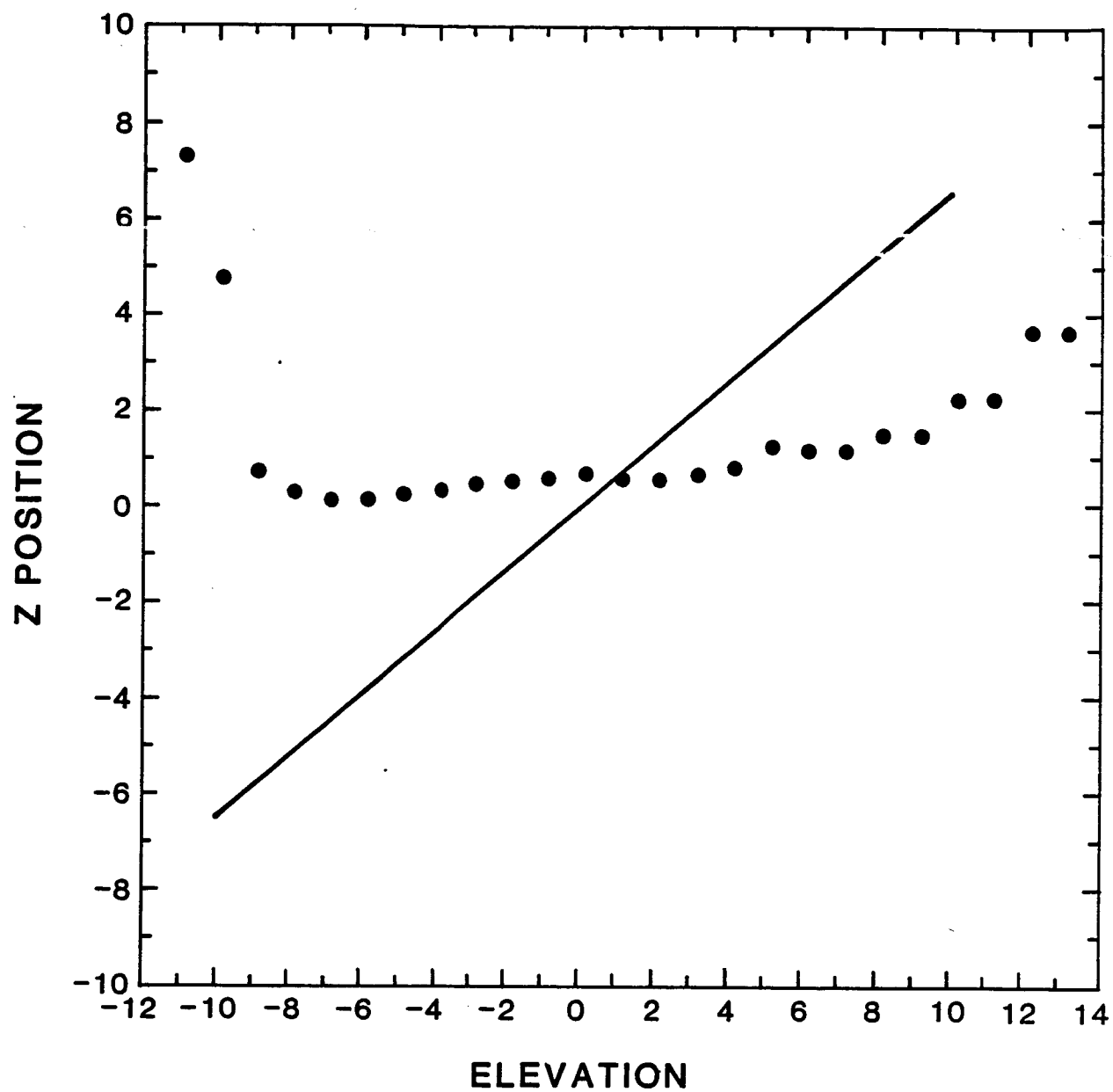


Figure 3-3

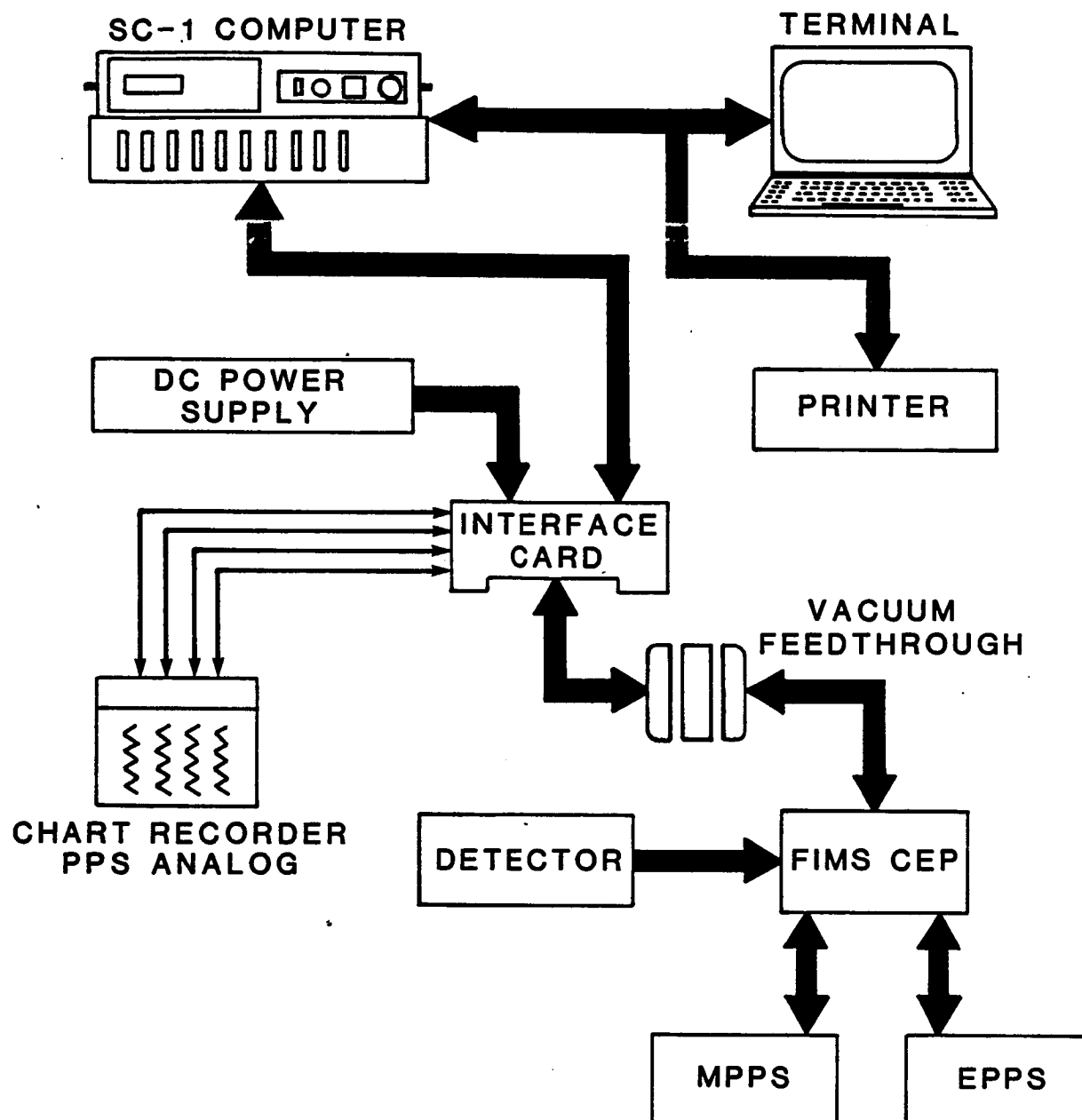


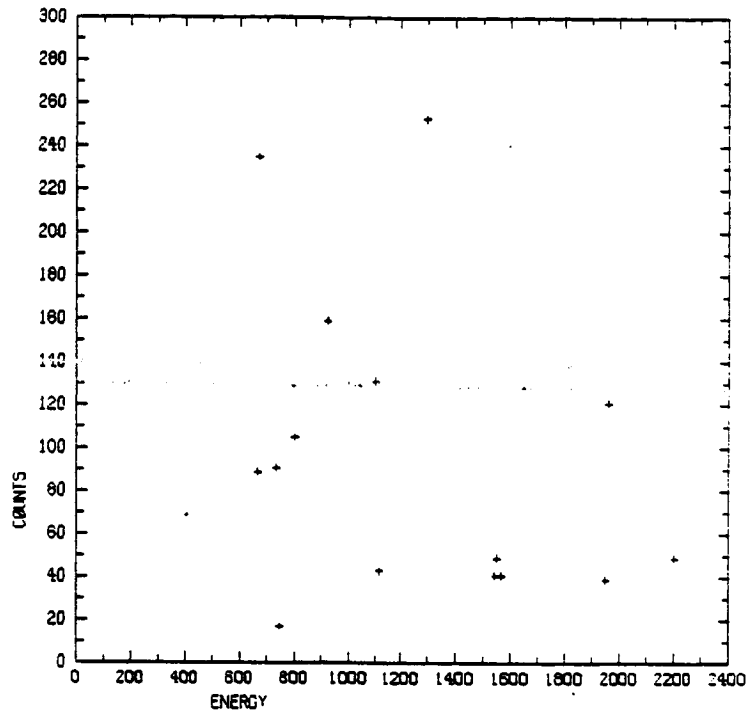
Figure 3-4

4. SCIENCE RESULTS

In January of 1986 the Centaur II sounding Rocket was launched from the Andoya Range in Norway. Due to a mechanical failure, the nose cone of the rocket was never completely released, blocking the view of the scientific payload and resulting in an inappropriate and shortened trajectory. Telemetry tapes recorded at the station in Andoya, and also by the NASA Wallops portable tacking station onsite, were studied extensively in the hope that some data might be retrieved; however, no clear evidence of mass peaks or a mass/energy correlation could be found.

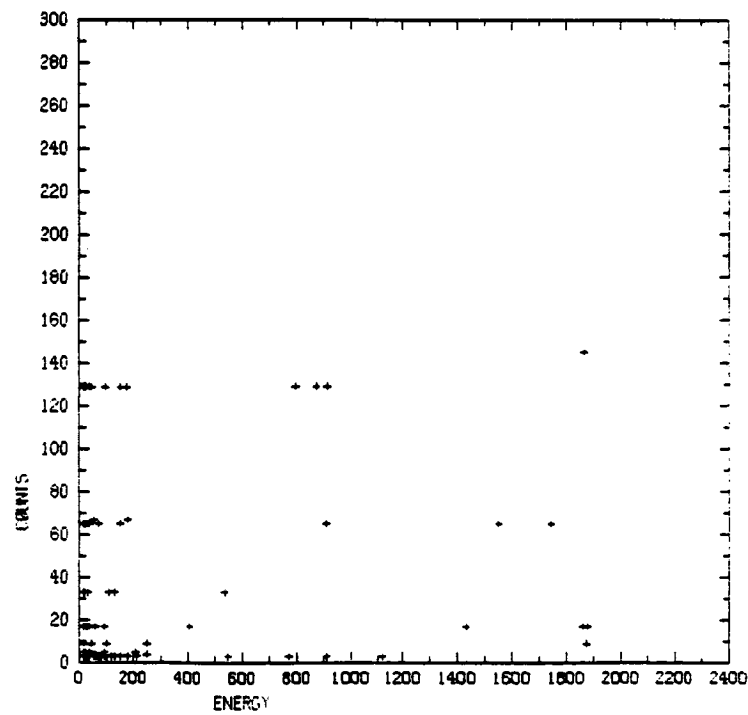
Figure 4-1 shows data from the original Andoya tapes; note the anomalous counts at powers of 2 (2, 4, 16, etc.). Figures 4-2 through 4-6 show data from the tapes supplied by Wallops. Figure 4-2 shows data from the first tape supplied by Wallops, selected for masses in the NO^+ range in (a) and the O^+ range in (b). Figure 4-3 shows the same data eliminating the time period in which the high voltage was turned on. Figure 4-4a shows data from the instrument inner channel and Figure 4-4b shows the same data eliminating the time interval for high-voltage turn-on. Figures 4-5a and 4-5b show data for the O^+ range and the NO^+ range, respectively, from the second Wallops tape. Finally, Figures 4-6a and 4-6b show data from the inner channel selected on the H_2^+ , H^+ range and on the NO^+ , O^+ range, respectively, from Wallops tape 2. Figure 4-6 clearly indicates that these counts are due to noise, since data are identical with the voltages set in the high-mass range (incorrect for the inner channel) and with voltages set in the low mass range (proper setting for this channel).

In summary, in all the outer channel data we see a noise pattern occurring at powers of two, and in all the inner channel data we see a constant (noise) count rate around 30-60 counts.



Masses Between 0
and 8.

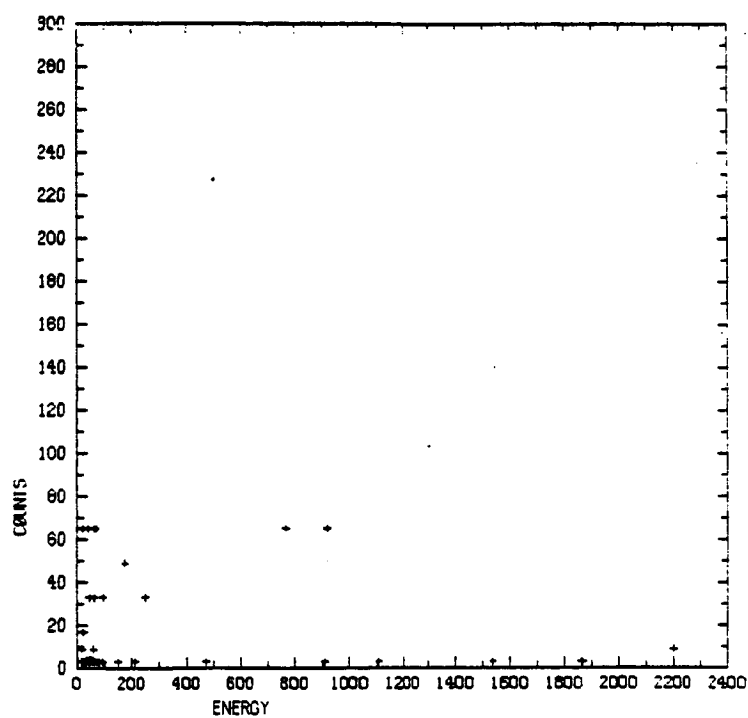
(a)



Masses Between 12
and 20.

(b)

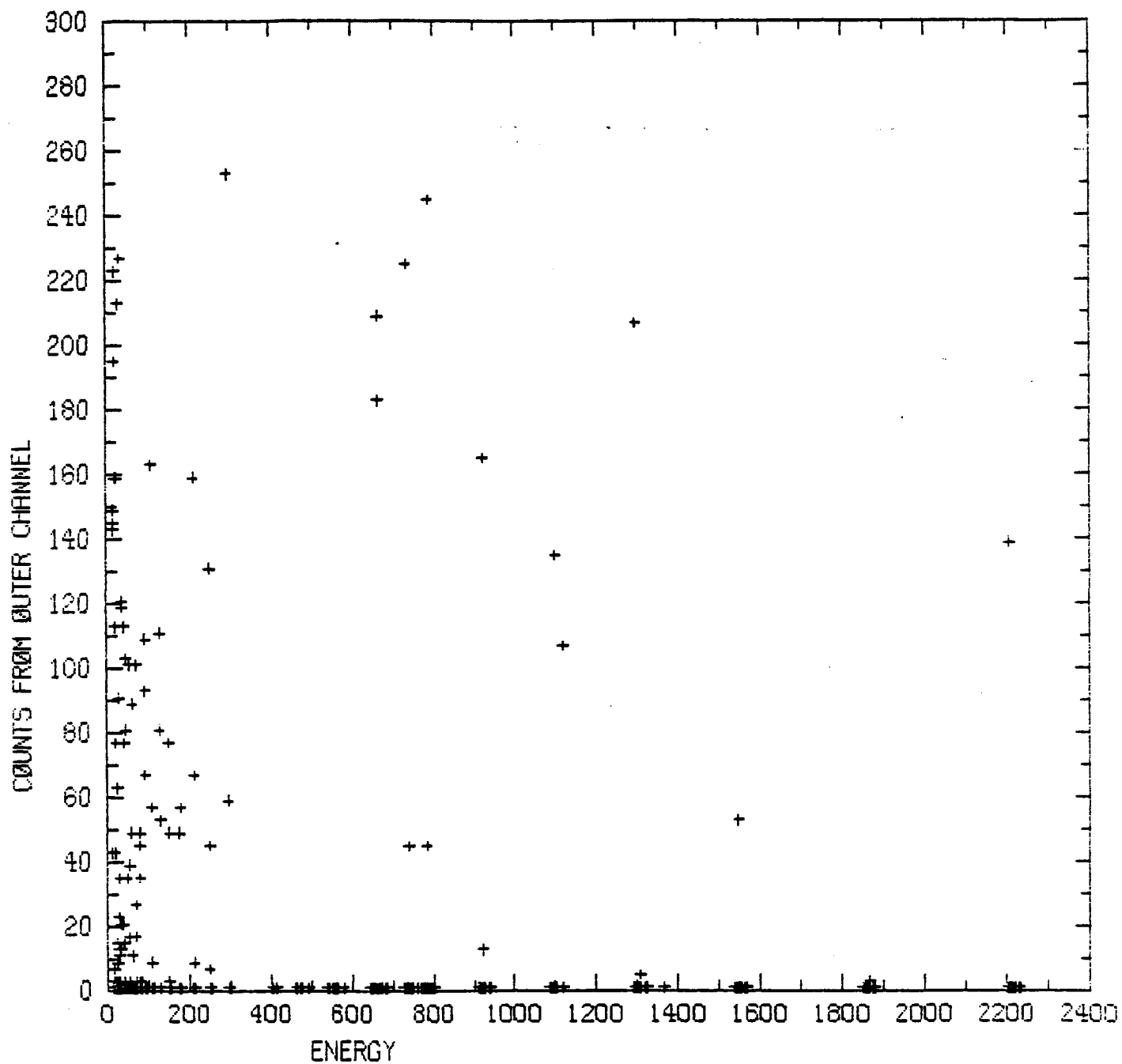
Figure 4-1



Masses Between 20
and 40.

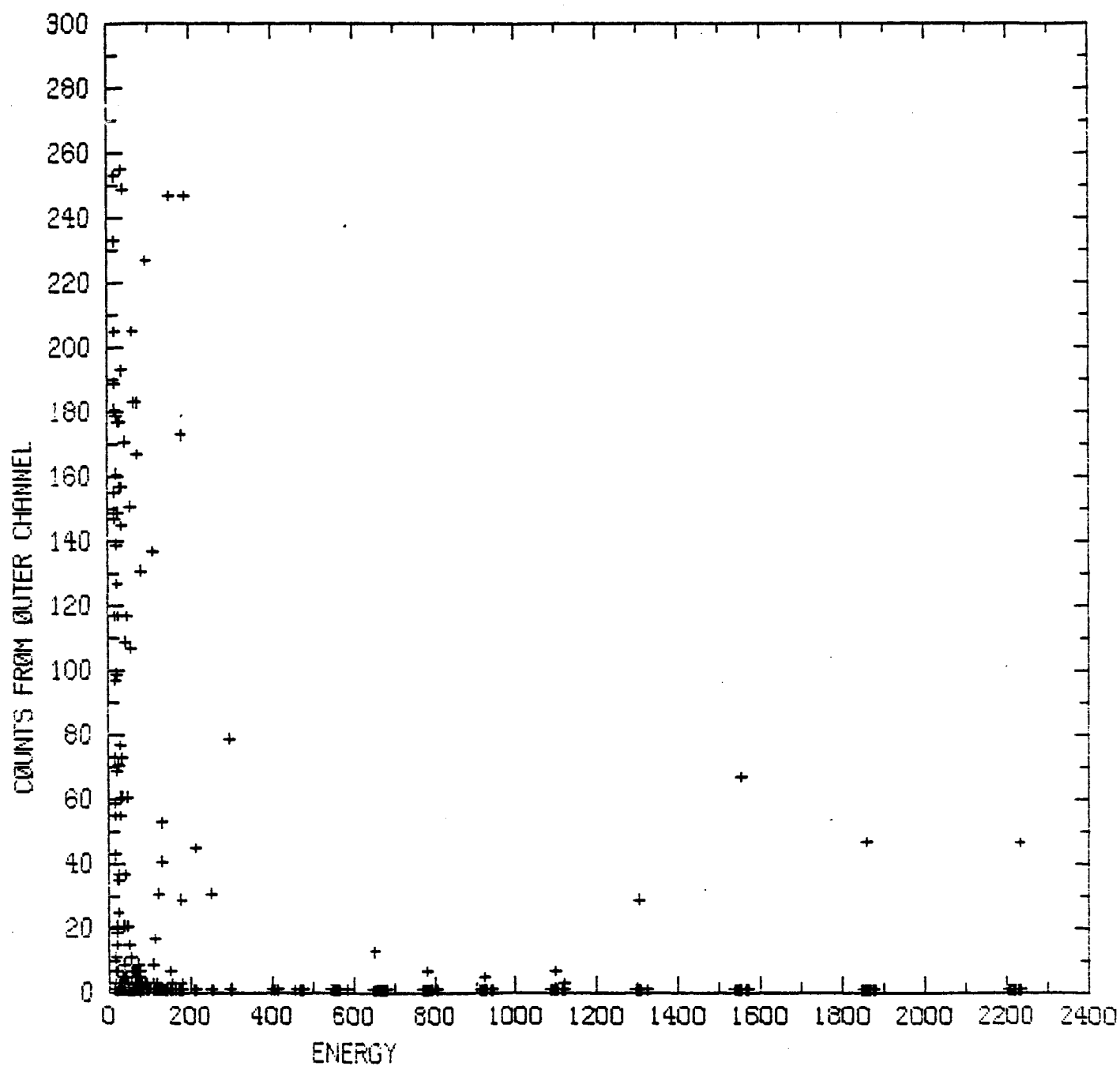
(c)

Figure 4-1



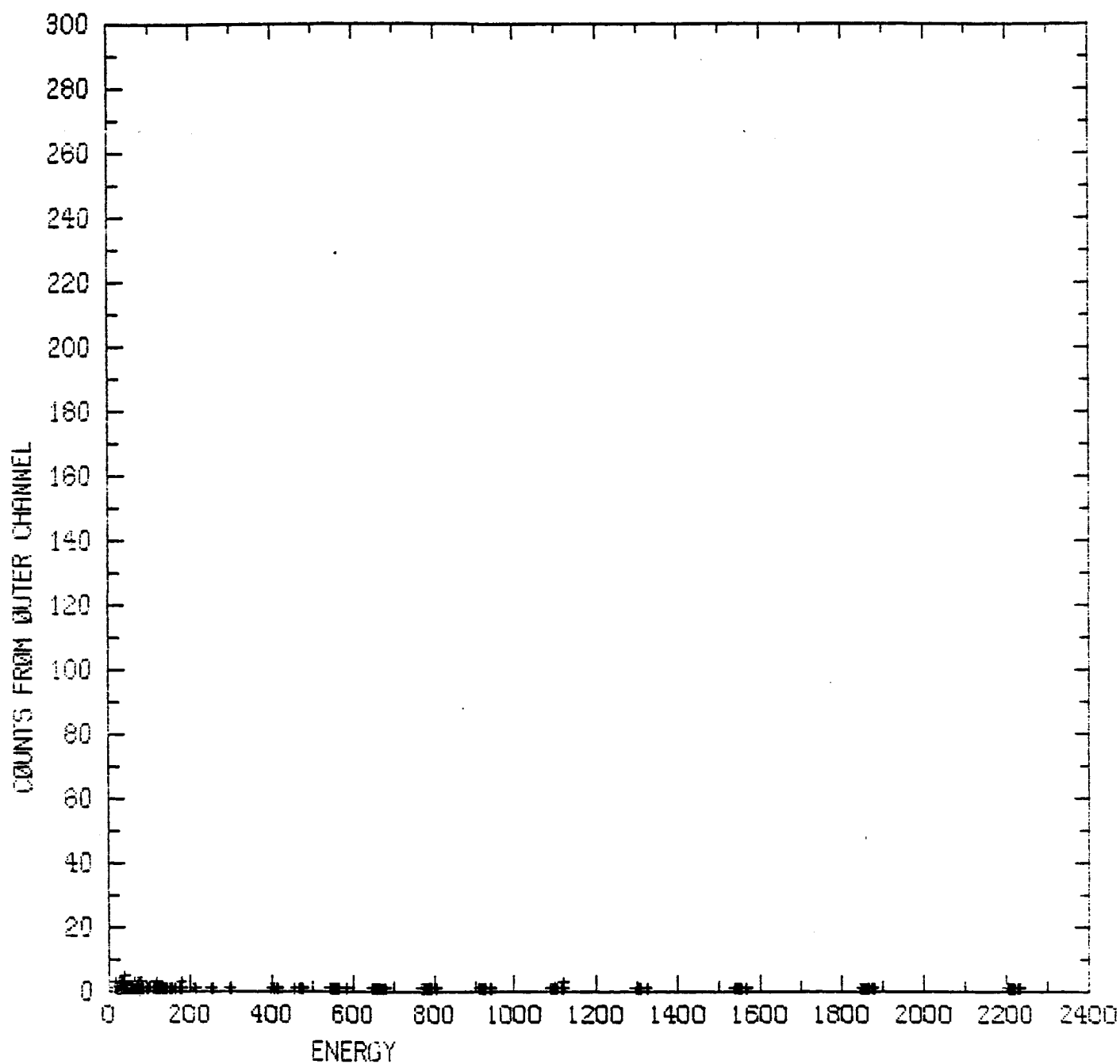
MASSES BETWEEN 20 AND 40 5000 TO 12000 WALLØPS TAPE 1

Figure 4-2a



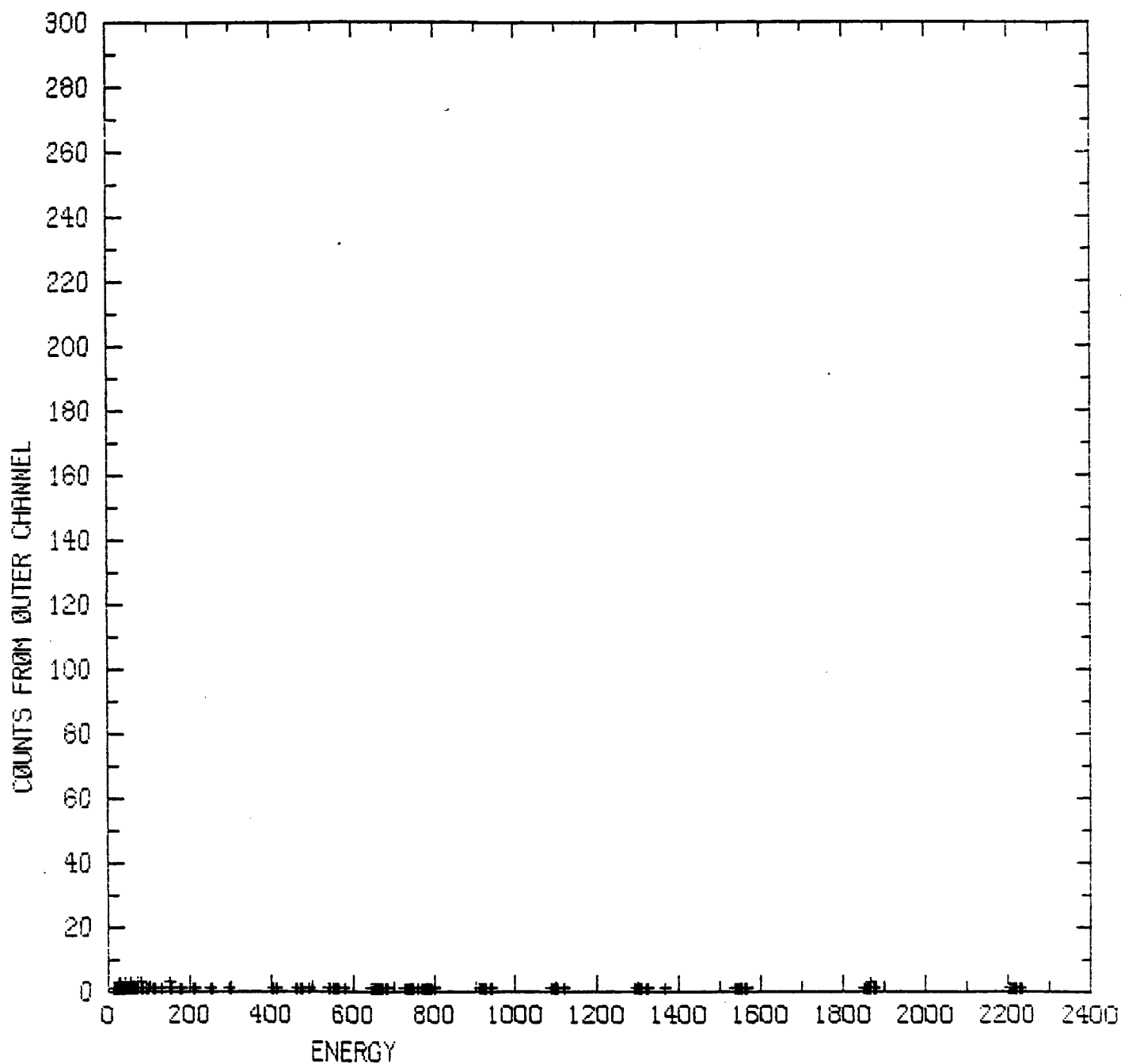
MASSES BETWEEN 12 AND 20 5000 TO 12000 WALLØPS TAPE 1

Figure 4-2b



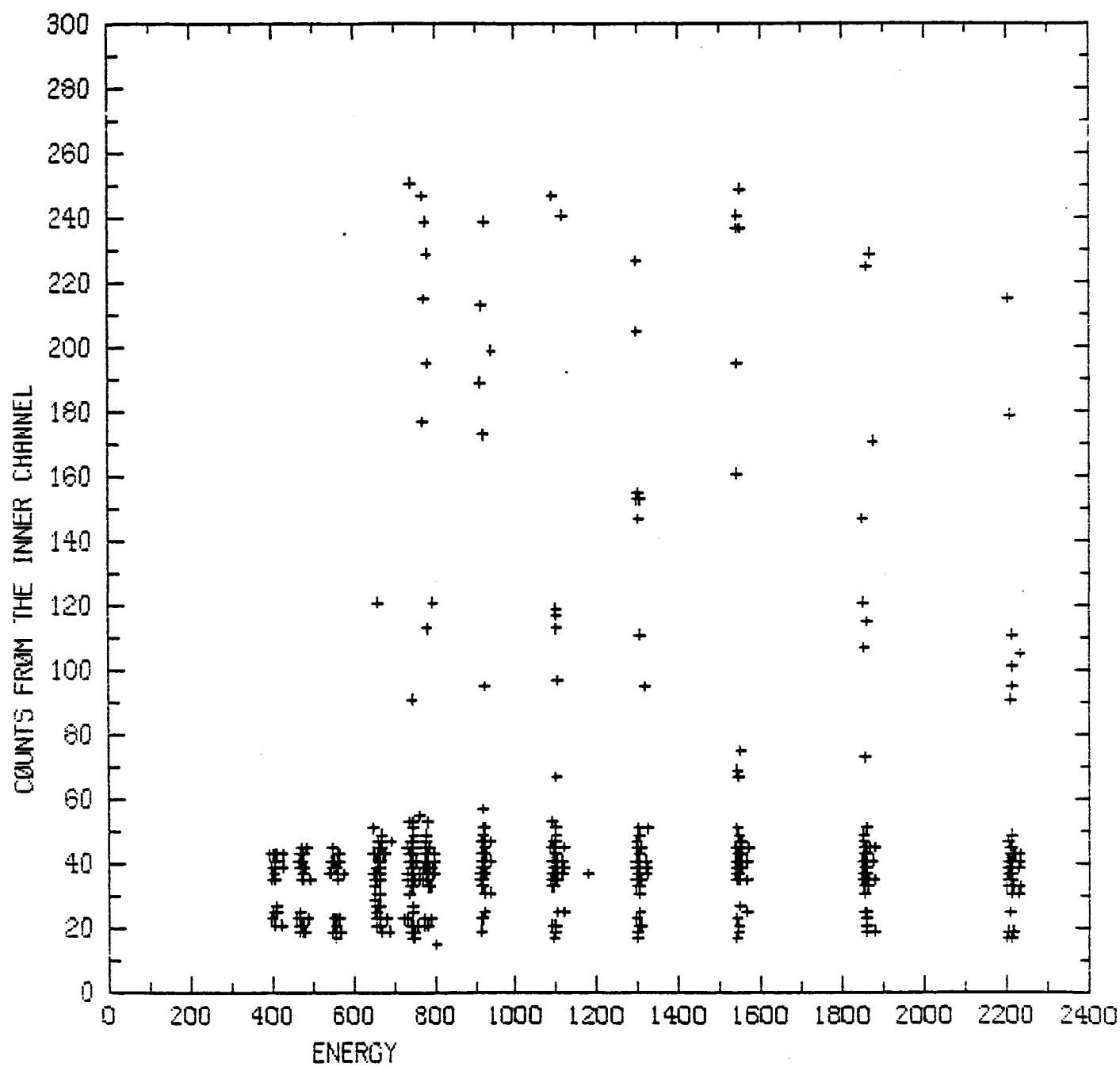
MASSES BETWEEN 12 AND 20 6000 TO 12000 WALLØPS TAPE 1

Figure 4-3a



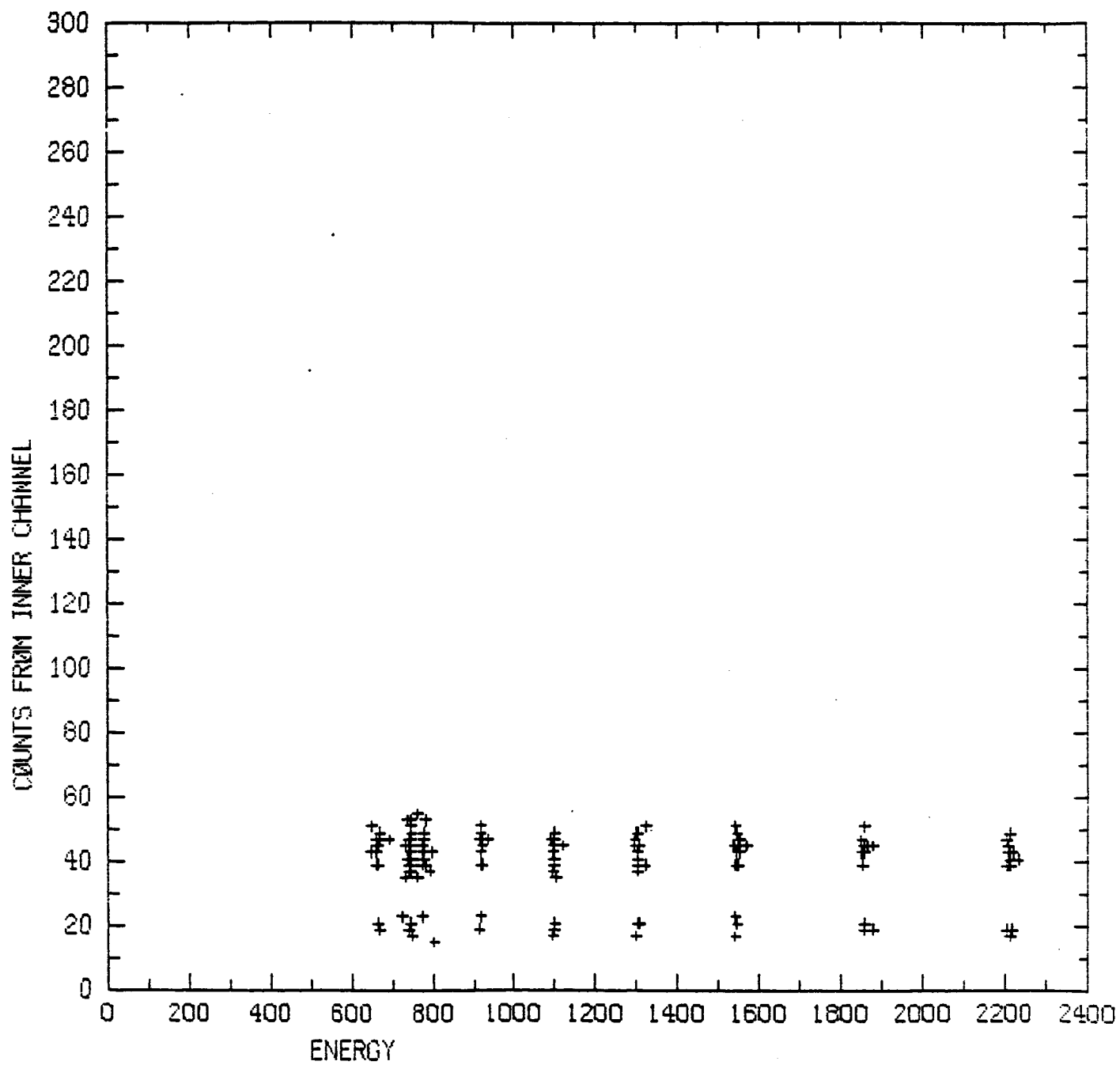
MASSES BETWEEN 20 AND 40 6000 TO 12000 WALLØPS TAPE 1

Figure 4-3b



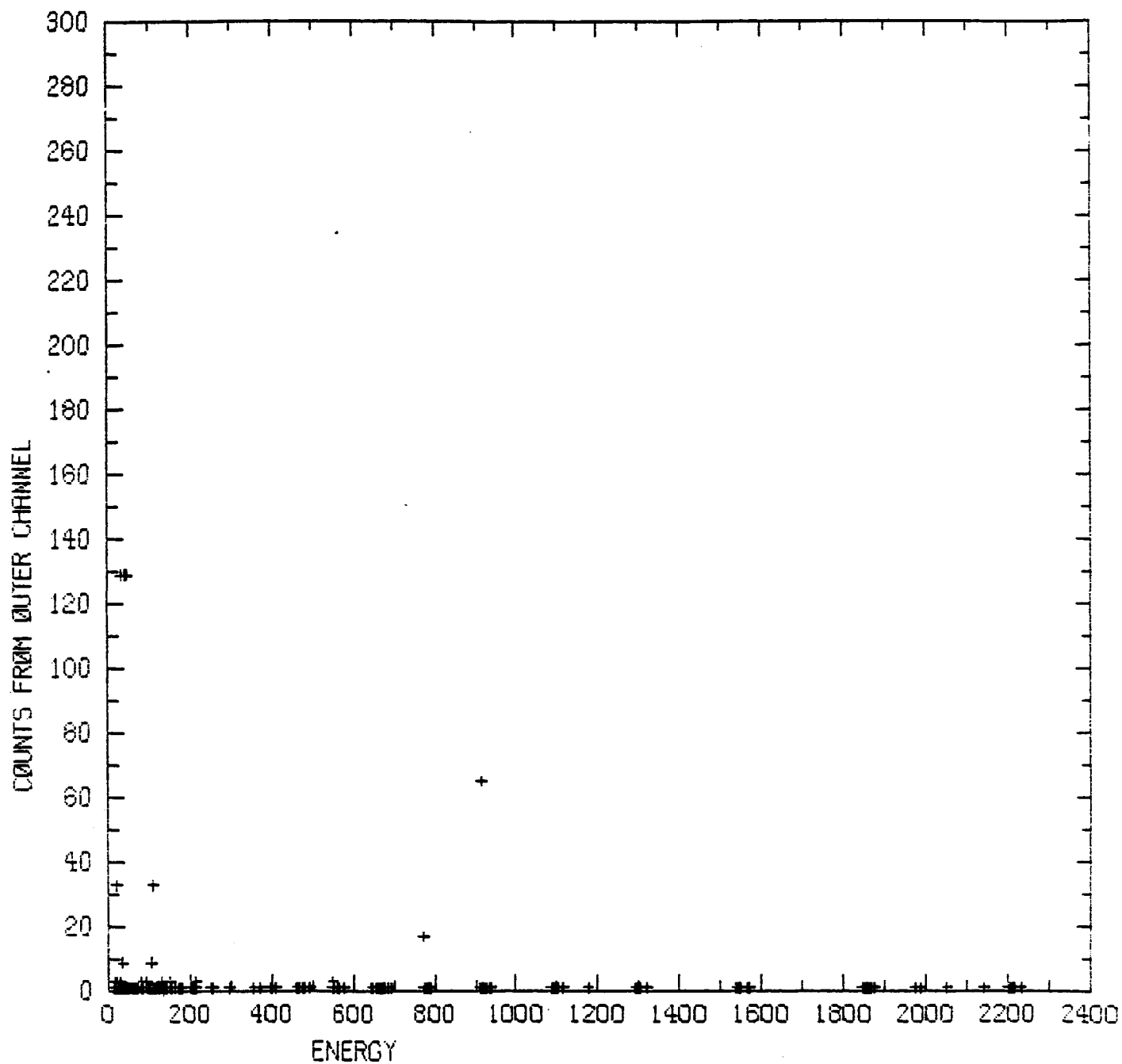
WALLØPS TAPE 1 MASSES LESS THAN 10

Figure 4-4a



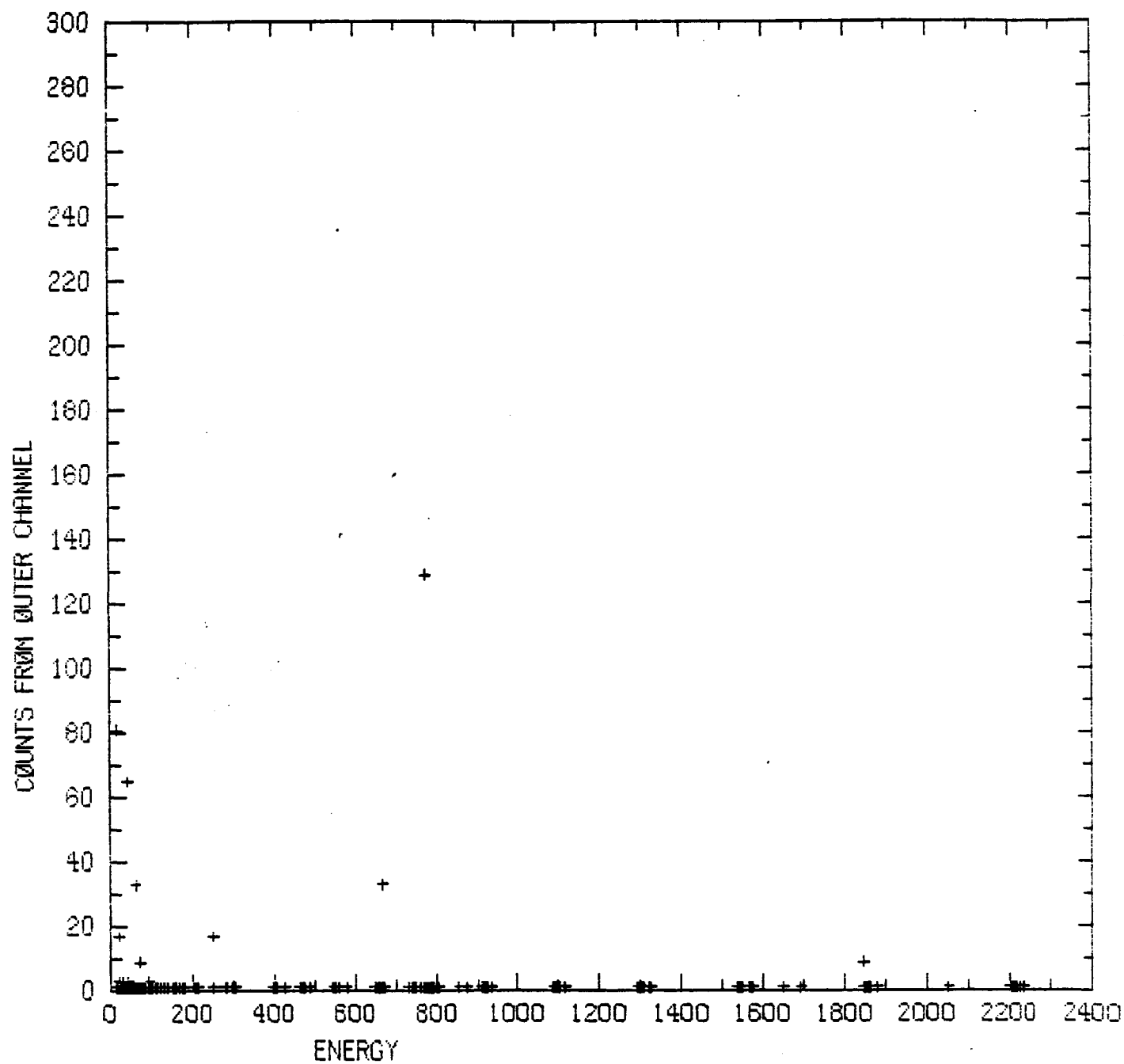
MASSSES LESS THAN 10 6000 TO 12000 WALLØPS TAPE 1

Figure 4-4b



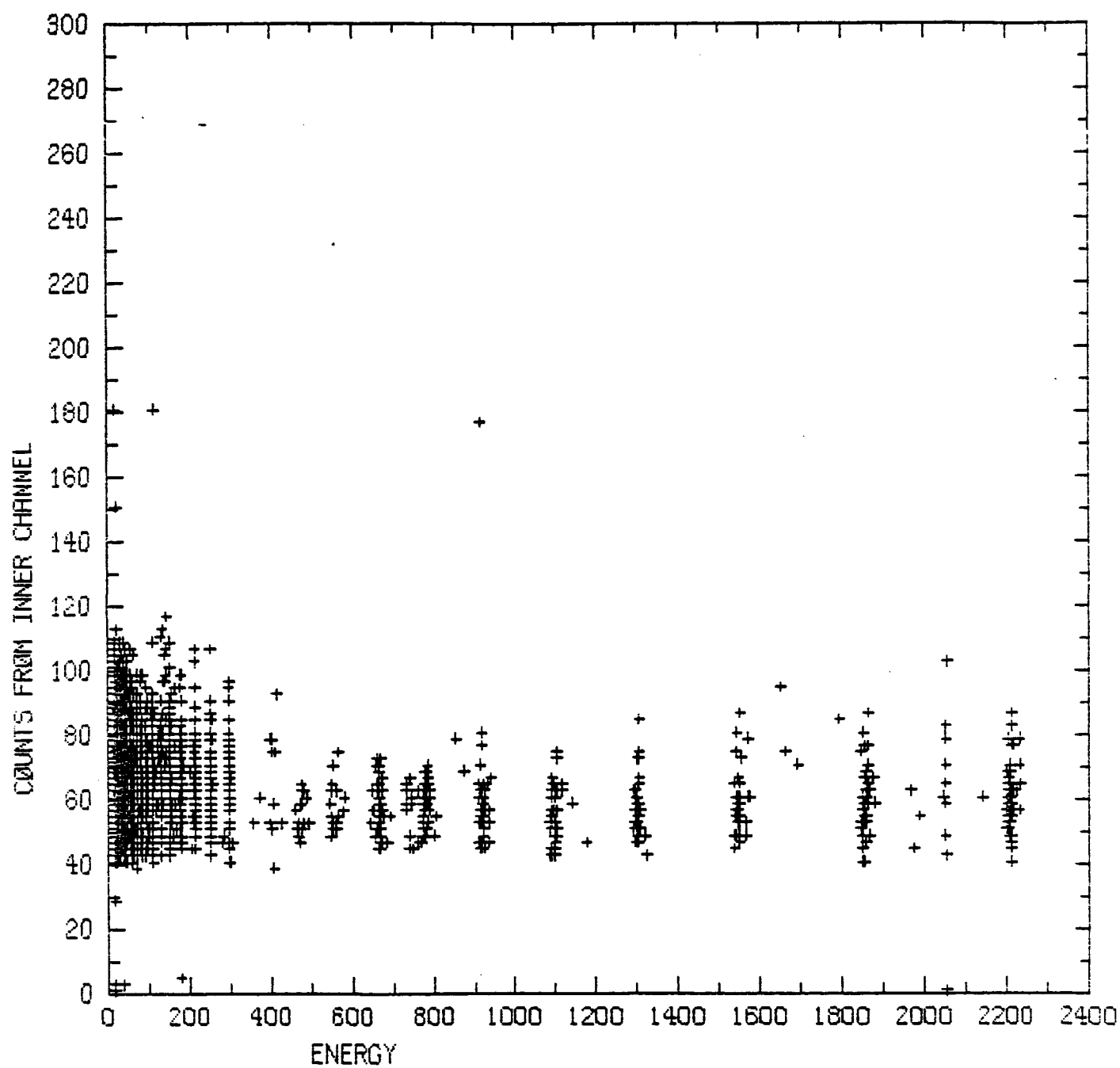
MASSSES BETWEEN 12 AND 20 1 TØ 8000 WALLØPS TAPE 2

Figure 4-5a



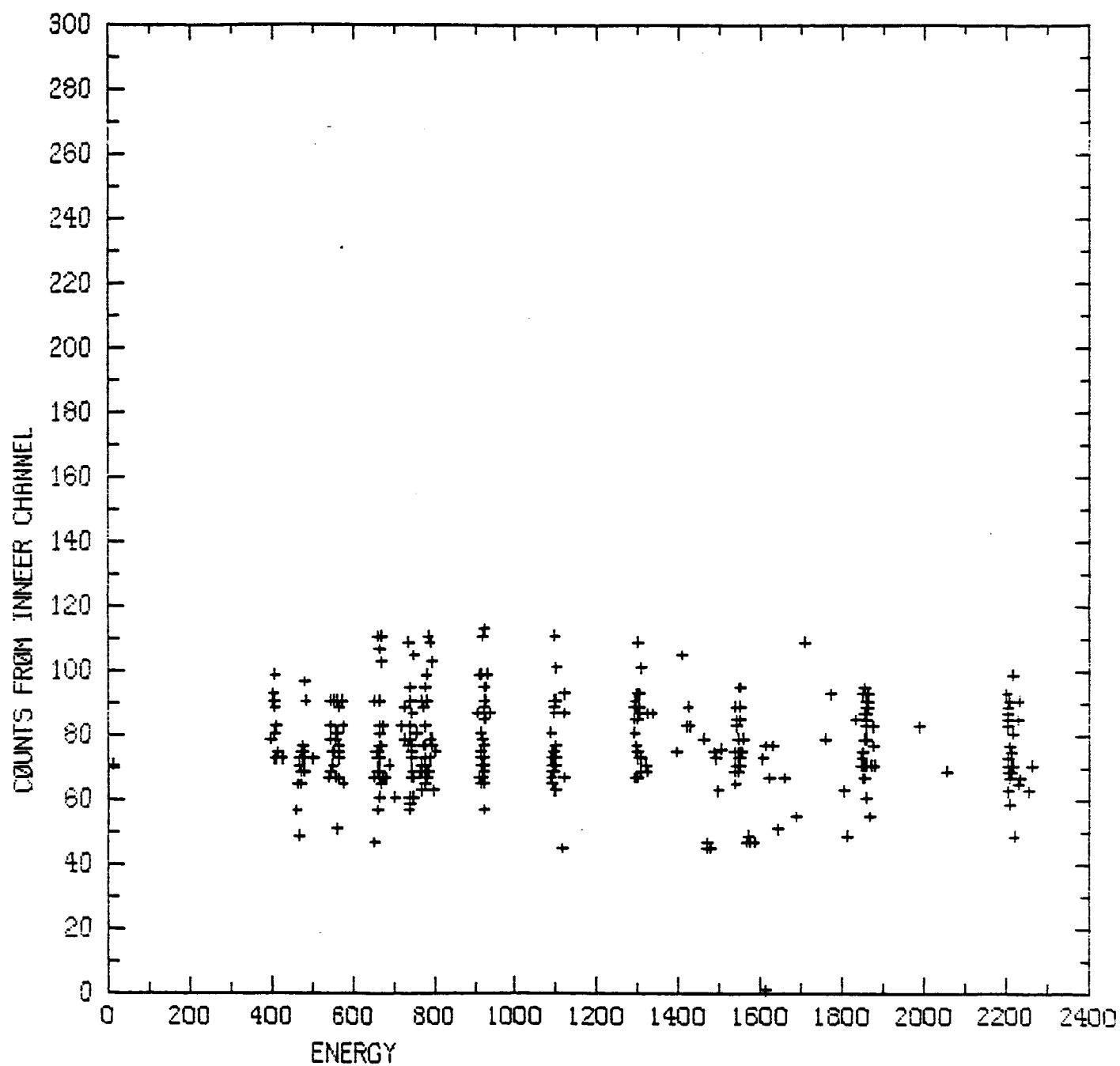
MASSSES BETWEEN 20 AND 40 1 TO 8000 WALLØPS TAPE 2

Figure 4-5b



MASSSES BETWEEN 10 AND 40 WALLØPS TAPE 2

Figure 4-6a



MASSSES LESS THAN 10 1 TO 8000 WALLØPS TAPE 2

Figure 4-6b

5. REFERENCES

Dusenbery, P.B. and Lyons L.R., "Generation of Ion-Conic Distribution by Upgoing Ionospheric Electrons", Journal of Geophysical Research, Vol.86, No.A9, pp. 7627-7638, September 1, 1981.

Klumpar, D.M., "Transversely Accelerated Ions: An Ionospheric Source of Hot Magnetospheric Ions", Journal of Geophysical Research, Vol.84, No. A8, pp. 4229-4237, August 1, 1979.

APPENDIX A
FIMS C Data Logbook

SECTION 1 / IDENTIFICATION DATA

Section 1: (1) List assembly/component name, inspection report (IR) number, MSFC specification number with revision and EO's, MSFC vendor part number and revision, serial number and engineering parts list (EPL) number. (2) Information on sub-assemblies/major components that are serialized or time and cycle sensitive should be recorded when included as a part of the main assembly. (3) All data entered in this section should reflect the as-built configuration and shall not be updated. (Changes to the as-built configuration shall be recorded in Section 3, Life History.) (4) Quality will certify by stamping and dating the bottom right hand corner of the last page of Section 1 to signify that the identification data accurately reflects the as-built configuration of the article at the time of layout/build-up inspection.

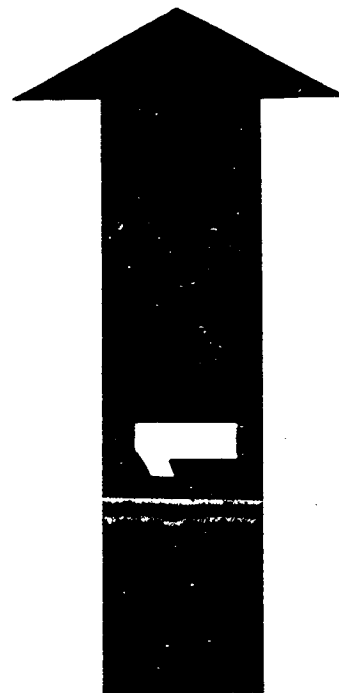
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SECTION 1 / IDENTIFICATION DATA									
ASSEMBLY/COMPONENT NAME		SPEC. NO.	MSFC PART NUMBER		VENDOR PART NUMBER				
CALIBRATION ASSEMBLY		50M26270	40M26271		EA-65120				
TELEMETRY - MODEL 2		REV. B	REV. A	EO's 1,2,3	REV. C				
		EO's 1,2	SER. NO. 060	EPL NO.	EPL 65120-100				
SUBASSEMBLY NAME	IR NO.	PART NO.	REV	S/N	EO's	EPL NO.	DAR's/MRD's		
CONTROL PANEL	AP-302	50M12347	A	006	NONE	-1018	AP-0096		
POWER SUPPLY	BA-302	50M12368	C	004	1,2	-101	NONE		
WIRING HARNESS	AZ-021	50M12396	NC	N/A	1	-101A	NONE		
(ALL ENTRIES SUBJECT TO SAMPLE)									
5/20/71									

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SECTION 2 / SPECIAL INSTRUCTIONS

Section 2: (1) List special safety precautions required to prevent death or injury to personnel and/or damage to the hardware. (2) List special tests, replacement, and service requirements. (3) Factors which would disqualify a component for flight or restrict its use should be listed here. Factors to be considered should include limiting aspects of environmental exposure, etc. (4) Design Engineering will specify the parameters that define a cycle. (5) Any other special instructions of any nature should be entered here.

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SECTION 3 / LIFE HISTORY

ORGANIZATION	EVENT		DATE	SUMMARY OF CHRONOLOGICAL EVENTS TEST NO./TEST DESC./NATURE & DESC OF MALFUNCTIONS/ ACTION TAKEN TO CORRECT/REPAIRS/REPLACEMENTS/REWORKS/ ADJUSTMENTS/MAINTENANCE/SHIPPED/RECEIVED/ENVIRONMENT/ETC.	RUNNING TIME/CYCLE		STAMP ON INITIALS
	NO.	SUBJECT			START	STOP	
PROD	1	Run-In	6-10-71	POST MANUFACTURING RUN-IN TEST PER SPEC. RM-889	09:40:09:50	09:10:00:10	JB
RM CO.							
QUAL							
PROD.	2	Accept. Test	6-11-71	ACCEPTANCE FUNCTIONAL TEST PER RM-870 SWITCH S8 FAILED OPEN. REF. DR 61366.	08:00:09:50	07:59:03:03	JB
RM CO.							
QUAL							
PROD.	3	Rework	6-11-71	REPLACED TOGGLE SWITCH S8 P/N 307843 ON CONTROL PANEL (REF. WO 8856)	--	--	JB
RM CO.							
QUAL							
PROD.	4	Accept. Test	6-13-71	ACCEPTANCE FUNCTIONAL TEST PER RM 870.	08:00:13:00	06:30:06:30	JB
RM CO.							
QUAL							
PROD.	5	Final Cleaning	6-15-71	CLEANED AND PACKAGED PER SPECIFICATION RM 816.	--	--	JB
RM CO.							
QUAL							
PROD.	6	Ship	6-16-71	SHIP TO MSFC (Ref. SO-11110)	--	--	JB
RM CO.							
QUAL							
PROD.	7	Receiving Inspection	6-29-71	VISUAL AND DIMENSIONAL.	--	--	JB
RM CO.							
QUAL							
PROD.	8	A.C.E. Functional	7-4-71	ACCEPTANCE FUNCTIONAL TEST PER ATP 40 M88571-A	10:00:13:00	09:00:00:00	JB
RM CO.							
QUAL							
PROD.	9	Transfer	7-10-71	ROUTED TO SEE-PE FOR STORAGE.	--	--	JB
RM CO.							
QUAL							
PROD.	10	Storage	7-10-71	RECEIVED AT SR-88-7/10/71	--	--	JB
RM CO.							
QUAL							
PROD.	11	Assignment	8/1/71	WITHDRAWN FROM STORAGE ON W.O. 11G-8871 ROUTED TO BLDG. 4708.	--	--	JB
RM CO.							
QUAL							
PROD.	12	Installation	8/3/71	INSTALLED ON 60M10016 UNIT S/N 000 PER TPS-ATM-FLT-0431	--	--	JB
RM CO.							
QUAL							
PROD.	13	Check-out	8/14/71	POST MANUFACTURING CHECKOUT PER FLT-TCP-H-70009	09:15:15:30	06:05:14:30	JB
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SECTION 3 / LIFE HISTORY

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SECTION 3 / LIFE HISTORY

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SECTION 4 / TEST DATA

Section 4: (1) The variable test data required to be recorded will be specified by Design Engineering in S Section 2. (2) Under "Description of Test" enter test procedure title or step and identification of the parameter being tested. (3) Identify the test procedure by number and revision level in the appropriate block. (4) Under "Test Limits" indicate the measurement limits for the test being conducted. (5) Under "Test Results" record the actual test measurement or satisfactory/unsatisfactory. (6) The test conductor's initials shall be noted in the column provided. (7) Normally, when tests are performed, detailed description of test, limits and results need not be recorded herein. Reference to the test procedure and data should be adequate, i.e., "Test performed in accordance with FLT-TCP-H-70009".

(8) The wire list E.O. level which is used to test electrical assemblies shall be recorded under description of test if applicable.

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RECEIVED 7 MRS LAYN K'ERDEN JORDAN

CHAN 1	CHAN 2	CHAN 3	CHAN 4	CHAN 5	CHAN 6	CHAN 7	CHAN 7
S10 FIMS MASS PPS STEP DATA	A41 FIMS MASS PPS + V MONITOR	A42 FIMS MASS PPS - Y MONITOR	S8 FIMS DATA	S9 FIMS PPS STEP DATA	A43 FIMS ENERGY PPS + V MONITOR	A44 FIMS ENERGY PPS - Y MONITOR	A4Z FIMS +5V MONITOR

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DATE/COMPONENT/EVENT	DESCRIPTION
28 JUNE 1985 CEP - CPU CARD ONLY LOW VOLTAGE APPLIED.	SUPPLYING 5V + GND TO J5 OF CEP (PINS 5+6) MONITORED CURRENT WITH HP CLIP-ON DC MILLIAMMETER NOTICE A PECULIAR POWER-ON CURRENT, SURGING ABOVE 200mA AND SETTLING TO 1.15mA OVER A 45SEC PERIOD. MONITORING 5V LINES WITH SCOPE. SEE QUOTE A BIT OF NOISE WITH SCOPE (VOLT RIPLE) ON 5V BUS CLOSEST TO 82C84/B CLOCK GENERATOR CHIP. 5V LINE NEAR CONNECTOR IS PRETTY CLEAN
28 JUNE CEP + CPU + MEMORY CARDS LOW VOLTAGE APPLIED	240 mA @ 5V HALF A VOLT RIPLE ON 5V BUSES (SCOPE)
5 AUG. 85 POWER APPLIED TO CPU + MEMORY CPU REMOVED TO ALLOW DCE PLUG-IN	240 mA @ 5V NO READY LINE - NO RAM READ/WRITE GREIVOUS NOISE PROBLEM ON 5V LINE OF CPU CARD 82PC08 DEVICES MAY OVER HEATED - PRESUMED DEAD HIGH RYTEREM + RAM NOT TRIED TO BEND ICE ABLE TO TALK TO MEMORY - PROBLEMS WITH WIRED CURRENT + NOISE ON V. + GROUND JUST NOT PRESENT! NO EXPLANATION 82PC08s REPLACED CPU MEM. + DET I/F CARDS EQUAL \approx 500mA ON + 5V + 5VIRTN LINES.
11 SEPT. 85 POWER APPLIED TO CPU, MEMORY, + DETECTOR I/F. (CPU REMOVED CH. VERY FIRST APPLICATION OF POWER TO DETECTOR BOARD.	8255 I/F DEVICE ON DET I/F CARD IS UN-RESPONSIVE TO EMULATOR. NO "RESET" SIGNAL APPEARS TO BE SUPPLIED TO THE DEVICE - PIN 69 ON CARD EDGE IS OPEN. RESET IS WIRED TO THE CHIP ITSELF. HY-4 PINS 2 + 3 - SEEM TO HAVE ONE SENSE HY-5 " " : SEEM TO HAVE OPPOSIT SENSE HY-5 - 3 MASTER FRAME - SHOULD BE HY-4 - 3 HY-5 - 4 MASTER FRAME IS IS CORL → EGS8 HAS CHOS REPERED HY-5 - 2 MINOR FRAME MINOR FRAME NOT HOME HY-5 - 2 FOUR 700KHz SEEMS BACKWARDS HY-5 -
HY-4	ORIGINAL OF POOR
HY-4	PAGE IS QUALITY
PPS I/F CARD	DATA BIT 3 IS HELD LOW - 06-8282 BAD, REPLACED. 540000 - 1194118 TAPITS BOUTITS WIRED BACKWARDS

DATE/COMPONENT/EVENT	DESCRIPTION
12 SEPT 85 FULL CEP.	NO MISWIRED SIGNALS IN CEP; TEST CABLE
16 SEPT 85 PROCESSOR PPS BOARD	SERIAL DATA NOT CLOCKING OUT ON S9 BUT FOR DATA BITS 8+9. FOUND DRAWING WAVE SPECIFYING PINS 5+6 OR U13 AND 11,12,13,14,3+4 OF U14 AS BEING GROUNDLED WHEN IN FACT THOSE WERE LIVE DATA LINES → CORRECTED WIRING OF U13 & U14 WITH SOLDER JUMPLERS JOINING DATA BUSES ON U10 & U11. S9 DATA IS FUNCTIONAL.
17 SEPT 85 PROCESSOR PPS BOARD — FIND CEP CONNECTOR J3+J4	CEP HAVING NOW TESTED (SAND BOARDS) FOR POWER SUPPLY & SIGNAL INTEGRITY ON THE ACTUAL SOUNDING ROCKET BUS.
11 11 11	ANALOGUE MONITOR SIGNALS FOR (PCM SLOTS) A41-A50 NOW VERIFIED USING 0-3V P-P TRIANGLE WAVE. AN EXTRA SIGNAL, A54, WAS VERIFIED (PIN 44), PER BILL GIBSON'S REQUEST.
	100 KHZ, E658, MAIN FRAME, MAIN FRAME ON J4 & E659 & E6510 ON J3 FROM THE ROCKET BUS.

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DATE/COMPONENT/EVENT

DESCRIPTION

13 OCT 85
FIMS ANALYSER/DETECTOR

ION CALIBRATION CONFIRMED FOR FIMS ~~CEM~~ CEM
DETECTOR.

H₂ IS SEEN IN BOTH CHANNELS

Error found & corrected in CEP flight code. MD800066 faulted & replaced.
Subtracted installed in Jan lab. Mass PPS faulted. UH4010 board.
One detector analogue board, U9+U10 changed to C80P from 100A
of the precision cal showed very low sensitivity
of Gas monitor in detector changed back to 100A.

Actual pickup output voltage = -316TV. before adjustment
Float V₀ = 100.3. before adjustment.

Amplifier B413B-101 Channels in FIMS, not connected for above
measurements.

Inner channel is "IN 1" to digital board
Outer channel is "IN 2" to digital board.

Decision made to adjust pickup to -3000V and adjust divider
ratio to get 815V float.

Voltage divider of 50M \pm 0.5% in series with a 30M \pm 1% and a
50M \pm 1% in parallel to give a measured reference voltage
of -815.0V for -2990V in CEM bias at this
potential. CEM NOT CONNECTED FOR THIS MEASUREMENT.

The following absolute voltage measurements were made directly on output of PPS.
(E Demand) E+ E- M+ M- (M Demand) (Error)

1.2% (85.01) 86.2 86.4 99.1 98.2 (98.69) \pm 0.39 0.7%

4.6% (175.7) 184.3 185.0 183.7 185.8 (186.5) - 1.5% 0.4%

95.2 94.4 (94.8) \pm 0.4% 0.4%

187.4 182.1 (182.9) \pm 2.5% 0.4%

Stage chart outputs of monitor signals on PPS - all verified within
 \pm 6.1% of absolute output voltage.

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DATE/COMPONENT/EVENT	DESCRIPTION
31 DEC 85 FIMS INSTRUMENT - PRIOR TO VERIFICATION TESTS	.032 - .036 AMPS DUAL STATE CURRENTS TO CEP DURING FUNCTIONAL [501] .165 - .185 A FOR BOTH PPS'S DURING FUNCTIONAL
2 JAN 86 Calibration for M281M30 ② 102642115 ev.	CONFIRMED NITROGEN @ 1.023 eV + 2133 eV CONFIRMED OXYGEN @ " " INSTRUMENT ACTIONED PER PRESSURE OF 2.9x10 ⁻⁷ T. Energy PPS still overloading at limited. Still has unexplained problem. DEFLECTION VOLTAGES APPEARED TO DROP OUT
2 JAN 86 FIMS ENERGY PPS VERIFICATION EFFORT - FLIGHT EPROMS.	INSPECTION OF EPPS PROM CARD REVEALED SEVERAL FUEL PADS NOT SOLDERED TO EPROM PINS, HIGH PROBABILITY THAT SIGNALS ARE INTERMITTENT, RESULTING IN APPARENT DROP OUT OF PPS VALVES.
" " "	12 PADS ARE EFFECTED. EPPS WILL BE RE TESTED AFTER PINS HAVE BEEN SOLDERED
3 JAN 86 FIMS INSTRUMENT PRIOR TO SHIPMENT.	ALL PPS BOARDS REMOVED. LOOSE CAPACITORS RTVD IN PLACE. ALL BOARDS CRATED + VACUUM BAKED CEP BOARDS WAO ALL COMPONENTS SOLDERED + WERE CRATED.
" " "	PRE SHIPMENT FUNCTIONAL .5 AMPS ON CEP PUMP UP. CEP FUNCTIONAL (CURRENT IS DRAWN WITH INTERFERENCE TO SC-1). PPS CURRENT ~.165 AMPS
" " "	CHART RECORD SHOWS TEPBS + -EPBS MONITORS IN SOME EPROMIOUS STATE BUT THEY RECOVER UPON RECEIPT OF FIRST COMMAND. Ran new E PPS test with corrected pressure - absolutely no change! - Modified CEP time on time constant to 0.4 seconds.
4 JAN 86 SCOPFI FIMS INSTRUMENT FINAL TEST. PRIOR TO SHIPMENT.	SYSTEM PUMPED DOWN OVERNIGHT TO 10 ⁻⁶ Torr (VACUATION) BEAM OF 1KEV DETECTED. PROPER PPS STEP + PROPER CHANNEL. BEAM INCREASED TO AFTER SYSTEM RESTORED CORRECTLY. SYSTEM POWER OFFED (VACUUM) PACKED FOR SHIPMENT

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DATE/COMPONENT/EVENT

DESCRIPTION

10 JAN 86 ANDOYA TEST RANGE
PRE-LAUNCH TEST. ~~TEST~~ WITH
GSE

11 JAN 86 ANDOYA TEST RANGE
PRE-LAUNCH END TO END TAI CHECK.

STIM INPUT TEST USING HP FUNCTION GENERATOR.
MONITOR PCM DECOM.
INVERT PEAK INPUT. FREQUENCY ADJUSTED TO FILL
COUNTER. PCM COUNT MATCHED GEDU.EREQ.

SEED LAUNCH SEQUENCE TEST COMPLETED.

RECORDED DATA LOOKS GOOD ON PLAYBACK

TEST OPERATED FOR EQUIVARIANT TIME OF FLIGHT, 800SEC

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APPENDIX B
Software Listing

SERIES-III 8086/8087/8088 MACRO ASSEMBLER V1.0 ASSEMBLY OF MODULE TABLE
OBJECT MODULE PLACED IN FIRST TABLE:DDU
INVOCATION LINE CONTROLS: TITLE(17:00:00 23 DEC.85)

LOC	OBJ	LINE	SOURCE
1		1	*****
2		2	*****
3		3	*****
4		4	*****
5		5	*****
6		6	*****
7		7	*****
8		8	*****
9		9	*****
10		10	*****
11		11	*****
12		12	*****
13		13	*****
14		14	*****
15		15	*****
16		16	*****
17		17	*****
18		18	*****
19		19	*****
20		20	*****
21		21	*****
22		22	*****
23		23	*****
24		24	*****
25		25	*****
26		26	*****
27		27	*****
28		28	*****
29		29	*****
30		30	*****
31		31	*****
32		32	*****
33		33	*****
34		34	*****
35		35	*****
36		36	*****
37		37	*****
38		38	*****
39		39	*****
40		40	*****
41		41	*****
42		42	*****
43		43	*****
44		44	*****
45		45	*****
46		46	*****
47		47	*****
48		48	*****

TABLE:ASM

PROCEDURE NAME:TABLE

PROGRAMMER:W.C. GIBSON

LAST REVISION:23 DEC.85

INPUTS:NONE

OUTPUTS:LOADS JUMP ADDRESS FOR INTERRUPT TYPE CODE 20

DESTROYS:NOTHING

FUNCTIONS:THE PURPOSE OF THIS PROCEDURE IS TO PASS THE

THE INTERRUPT JUMP ADDRESS FOR IRQ OF THE

SLAVE 8259 INTERRUPT CONTROLLER.

NAME:TABLE

SEGMENT

DATA

ENDS

CODE

SEGMENT

ASSUME CS:CODE,DS:DATA

EXTRN MAJORFRAME

PUBLIC TABLE

NEAR

PROC

PUSH DS

PUSH BX

PREPARE WORKING REGISTERS

MOV AX,0

ABSOLUTE MEMORY SEG.

MOV DS,AX

1SEG. WHERE VECTOR IS LOADED

MOV BX,00B4H

YOFFSET FOR TYPE CODE OF 20

MOV AX,OFFSET MAJORFRAME

MOV BXJ,AX

SEND OFFSET OF INT. ROUTINE

MOV AX,CODE

LOAD SEGMENT OF INT.ROUTINE

MOV BX+2J,AX

LOAD SEGMENT

POP AX

PRESTORE

POP BX

POP DS

RESTORE EVERYCODE

RET

ENDP

ENDS

TABLE

CODE

END

ORIGINAL PAGE IS
OF POOR QUALITY

LOC	OBJ	LINE	SOURCE
000F	26E705F900	51	MOV WORD PTR ES:DI,00F9H
0014	C70600000000	52	MOV WORD PTR MINOR_FRAME, 0000H
001A	E80000	53	CALL GET_SCIENCE_DATA
001B	893E000001	54	CMP WORD PTR MODE,1
0022	7503	55	JNE NORMAL_MODE
0024	E96706	56	JMP SPECIAL_MODE ;IF MODE=1,SPECIAL MODE
0027	8306000002	57	ADD WORD PTR ENERGY_COUNTS,2
002C	635E000058	58	CMP WORD PTR ENERGY_COUNTS,58
0031	7015	59	JGE ENERGY_MAX_NORMAL ?
0033	8B1E0000	60	MOV DX,WORD PTR ENERGY_COUNTS
0037	2E8B7000	61	MOV AX,WORD PTR ENERGY_COMMANDS[DX]
003C	F7D0	62	NOT AX
003E	8B3E0000	63	MOV DI,WORD PTR PP32STR
0042	265905	64	MOV ES:DI,DI,AX ;SEND ENERGY FPS CMD.
0045	EB1490	65	JMP MASS_NORMAL ;CONTINUE WITH MASS
0048	C70600000000	66	MOV WORD PTR ENERGY_COUNTS,0
004E	2EA10000	67	MOV AX,WORD PTR ENERGY_COMMANDS
0052	F7D0	68	NOT AX
0054	8B3E0000	69	MOV DI,WORD PTR PP32STR
0058	265905	70	MOV ES:DI,DI,AX ;SEND ENERGY 1ST CMD.
005B	8306000002	71	ADD WORD PTR MASS_COUNTS,2
0060	813E00000602	72	CMP WORD PTR MASS_COUNTS,516
0066	7015	73	JGE MASS_MAX_NORMAL
0068	8B1E0000	74	MOV DX,WORD PTR MASS_COUNTS
006C	2E8B7000	75	MOV AX,WORD PTR MASS_COMMANDS[DX]
0071	F7D0	76	NOT AX
0073	8D3E0000	77	MOV DI,WORD PTR PP32STR
0077	265905	78	MOV ES:DI,DI,AX ;SEND NORMAL MASS CMD.
007A	E9A100	79	JMP MAJOR_FRAME_OUT
007D	C70600000000	80	MOV WORD PTR MASS_COUNTS,0
0083	8906000001	81	ADD WORD PTR PASS_COUNT,1
0088	835E000000A	82	CMP WORD PTR PASS_COUNT,10
008B	720F	83	JD NOT_SPECIAL_MODE
008F	C706000000100	84	MOV WORD PTR MODE,1 ;SET MODE SPECIAL
0095	C70600000000	85	MOV WORD PTR PASS_COUNT,0
0098	EB1190	86	JMP SPECIAL_MODE ;ENTER SPECIAL MODE
009E	2EA10000	87	MOV AX,WORD PTR MASS_COMMANDS
00A2	F7D0	88	NOT AX
00A4	8B3E0000	89	MOV DI,WORD PTR PP32STR
00A8	265905	90	MOV ES:DI,DI,AX ;SEND 1ST MASS CMD.
00AB	EB1190	91	JMP MAJOR_FRAME_OUT
00AE	8306000002	92	ADD WORD PTR ENERGY_COUNTS,SPECIAL,2
00B3	813E00000603	93	CMP WORD PTR ENERGY_COUNTS,SPECIAL,75
00B9	7015	94	JGE ENERGY_MAX_SPECIAL
00BB	8B1E0000	95	MOV DX,WORD PTR ENERGY_COUNTS,SPECIAL
00BF	2E8B7000	96	MOV AX,WORD PTR SPECIAL_MODE_ENERGY[DX]
00C4	F7D0	97	NOT AX
00C6	8B3E0000	98	MOV DI,WORD PTR PP32STR
00CA	265905	99	MOV ES:DI,DI,AX ;SEND SPECIAL ENERGY
00CB	EB1490	100	JMP MASS_SPECIAL
00B0	C70600000000	101	MOV WORD PTR ENERGY_COUNTS,SPECIAL,0
00B6	2EA10000	102	MOV AX,WORD PTR SPECIAL_MODE_ENERGY
00BA	F7D0	103	NOT AX
00BC	8B3E0000	104	MOV DI,WORD PTR PP32STR
00E0	265905	105	MOV ES:DI,DI,AX ;SEND MAX SPECIAL ENR.

LOC OBJ LINE SOURCE

```
00E9 0906000002 E 106 MASS-SPECIAL: ADB WORD PTR MASS-COUNTS-SPECIAL,2
00EA 013E00001A0C E 107 CMP WORD PTR MASS-COUNTS-SPECIAL,3098
00EB 7B15 108 JOE MASS-MAX-SPECIAL
00EC 0B1E0000 E 109 MOV DX,WORD PTR MASS-COUNTS-SPECIAL
00ED 2E0B070000 E 110 MOV AX,WORD PTR SPECIAL-MODE-MASSIDIX
00EE F7B0 111 NOT AX
00EF 0B3E0000 E 112 MOV DI,WORD PTR PP31STR
00F0 260905 113 MOV ESI,DI;TAX TSEND MASS-SPECIAL
00F1 EB1A90 114 JMP MAJOR-FRAME-OUT
00F2 070600000000 E 115 MOV WORD PTR M0BE,0
00F3 2EA10000 E 116 MOV WORD PTR MASS-COUNTS-SPECIAL,0
00F4 F7B0 117 NOT AX
00F5 0B3E0000 E 118 MOV DI,WORD PTR PP31STR
00F6 260905 119 MOV ESI,DI;TAX TSEND MASS-SPECIAL MAX
00F7 5B 120 POP AX
00F8 5B 121 POP DX
00F9 5F 122 POP DI
00FA CF 123 INEB
00FB 124
00FC 125 MAJOR-FRAME-INTERRUPT ENDF
00FD 126 CODE ENDS
00FE 127 END
```

ASSEMBLY COMPLETE, NO ERRORS FOUND

ORIGINAL PAGE IS
OF POOR QUALITY

SERIES-III 8086/8087/8088 MACRO ASSEMBLER V1.0 ASSEMBLY OF MODULE MINOR
 OBJECT MODULE PLACED IN IFS:MINOR.OBJ
 INVOCATION LINE CONTROLS: TITLE(10:25:00 17 OCT.85)

LOC	OBJ	LINE	SOURCE
		1	*****
		2	MINOR.ASH
		3	*****
		4	*****
		5	*****
		6	*****
		7	*****
		8	*****
		9	*****
		10	*****
		11	*****
		12	*****
		13	*****
		14	*****
		15	*****
		16	*****
		17	*****
		18	*****
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		45	*****
		46	*****
		47	*****
		48	*****
		49	*****
		50	*****

ORIGINAL PAGE IS
OF POOR QUALITY

LOC OBJ LINE SOURCE

LOC	OBJ	LINE	SOURCE
002E	5B	51	POP DX
002F	5F	52	POP DI
0030	6F	53	IRET
		54	MINOR_FRAME_INTERRUPT
		55	CODE
		56	END

ASSEMBLY COMPLETE, NO ERRORS FOUND

ORIGINAL PAGE 13
OF POOR QUALITY

SERIES-III 8086/8087/8088 MACRO ASSEMBLER V1.0 ASSEMBLY OF MODULE DATAN
OBJECT MODULE PLACED IN 1761DATA.OBJ
INVOCATION LINE CONTROLS: TITLE(09:20:00 17 OCT.85)

LOC OBJ LINE SOURCE

1 ***** DATA.ASH *****

2 ***** DATA.ASH *****

3 ***** DATA.ASH *****

4 ***** DATA.ASH *****

5 ***** DATA.ASH *****

6 ***** DATA.ASH *****

7 ***** DATA.ASH *****

8 ***** DATA.ASH *****

9 ***** DATA.ASH *****

10 ***** DATA.ASH *****

11 ***** DATA.ASH *****

12 ***** DATA.ASH *****

13 ***** DATA.ASH *****

14 ***** DATA.ASH *****

15 ***** DATA.ASH *****

16 ***** DATA.ASH *****

17 ***** DATA.ASH *****

18 ***** DATA.ASH *****

19 ***** DATA.ASH *****

20 ***** DATA.ASH *****

21 ***** DATA.ASH *****

22 ***** DATA.ASH *****

23 ***** DATA.ASH *****

24 ***** DATA.ASH *****

25 ***** DATA.ASH *****

26 ***** DATA.ASH *****

27 ***** DATA.ASH *****

28 ***** DATA.ASH *****

29 ***** DATA.ASH *****

30 ***** DATA.ASH *****

31 ***** DATA.ASH *****

32 ***** DATA.ASH *****

33 ***** DATA.ASH *****

34 ***** DATA.ASH *****

35 ***** DATA.ASH *****

36 ***** DATA.ASH *****

37 ***** DATA.ASH *****

38 ***** DATA.ASH *****

39 ***** DATA.ASH *****

40 ***** DATA.ASH *****

41 ***** DATA.ASH *****

42 ***** DATA.ASH *****

43 ***** DATA.ASH *****

44 ***** DATA.ASH *****

45 ***** DATA.ASH *****

46 ***** DATA.ASH *****

47 ***** DATA.ASH *****

48 ***** DATA.ASH *****

49 ***** DATA.ASH *****

50 ***** DATA.ASH *****

ORIGINAL PAGE IS
OF POOR QUALITY

LOC OBJ LINE SOURCE

```
000F 268905 51 CLEAR_A0_A1: MOV ES:DIJ,AX
0012 E84500 52 CALL DELAY IWAIT 2 MICRO.
0015 E80600 53 MOV AX,0006H ICLEAR A1
0018 268903 54 READ_FIRST_WORD: MOV ES:DIJ,AX
001B E83C00 55 CALL DELAY IWAIT AGAIN
0021 E83900 56 CALL DELAY
0024 268904 57 MOV AX:WORD PTR ES:SIJ IFETCH DET.
0027 F7D0 58 NOT AX
002A 268905 59 MOV DI:WORD PTR DATASTR
002D 268906 60 MOV WORD PTR ES:DIJ,AX ISEND DATA
0031 B80700 61 MOV DI:WORD PTR PFI_CONTROL
0034 268903 62 MOV AX,0007H ISET A1
0037 E82000 63 CALL DELAY IWAIT
003A E80200 64 MOV AX,0002H ICLEAR A0
003B 268903 65 MOV ES:DIJ,AX ICLEAR A0
0040 E81700 66 CALL DELAY
0043 E81400 67 CALL DELAY
0046 268904 68 MOV AX:WORD PTR ES:SIJ IFETCH 2ND
0049 F7D0 69 NOT AX
004B 268905 70 MOV DI:WORD PTR PFI_PORTA
004F 268903 71 MOV WORD PTR ES:DIJ,AX
0052 A30200 72 MOV WORD PTR SCIENCE_DATA+2,AX ISTORE 2ND WRD
0053 5B 73 POP AX IPRESTORE
0056 5B 74 POP DX
0057 5E 75 POP BI IPRESTORE
0058 5F 76 POP DI IPRESTORE
0059 C3 77 RET IDONE
005A 90 78 PROC NEAR
005B C3 79 GET_SCIENCE_DATA: ENDP
005C 90 80 DELAY: PROC NEAR
005D C3 81 RET IDONE
005E 90 82 RET IDONE
005F C3 83 DELAY: PROC NEAR
0060 90 84 CODE: ENDP
0061 90 85 END
```

ASSEMBLY COMPLETE, NO ERRORS FOUND

ORIGINAL PAGE IS
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SERIES-III 8086/8087/8088 MACRO ASSEMBLER V1.0 ASSEMBLY OF MODULE FIMS
OBJECT MODULE PLACED IN :F6:FIMCAL.OBJ
INVOCATION LINE CONTROLS: TITLE(CAL ONLY 31 DEC.85)

LOC OBJ LINE SOURCE

```
1 *****
2 1* FCAL_ASM *****
3 *****
4 1* *****
5 1* PROCEDURE NAME:FIMS_MAIN *****
6 1* PROGRAMMER:W.C. GIBSON *****
7 1* LAST REVISION:30 DEC. 1985 *****
8 1* INPLETS:NONE *****
9 1* *****
10 1* OUTPUTS:CONFIGURES 8259,8255 AND EPS INTERFACES *****
11 1* *****
12 1* DESTROY:ALL REGISTERS *****
13 1* *****
14 1* FUNCTIONS:THIS ROUTINE IS THE MAIN PROGRAM ELEMENT OF *****
15 1* THE FIMS=C INSTRUMENT CENTRAL ELECTRONICS *****
16 1* *****
17 1* PROCESSOR THIS ROUTINE FIRST INITIALIZES *****
18 1* THE SEGMENT REGISTERS THEN THE VARIOUS *****
19 1* HARDWARE ELEMENTS HAVING INITIALIZED ALL OFF *****
20 1* THE DEVICES REQUIRING INITIALIZATION-THIS *****
21 1* ROUTINE HALTS AND WAITS FOR A MAJOR OR MINOR *****
22 1* FRAME INTERRUPT. *****
23 1* *****
24 1* *****
25 *****
26 ***** NAME FIMS *****
27 *****
28 DATA SEGMENT PUBLIC
29 PERIPHERALS DW 0200H
30 PROM_SEGMENT DW 0F000H
31 PROM_OFFSET DW 0C0C0H
32 RAM_SEGMENT DW 0000H
33 RAM_OFFSET DW 0000H
34 STACK_SEGMENT DW 0000H
35 STACK_POINTER DW 0F00H
36 PPI_PORTA DW 0030H
37 PPI_PORTB DW 0034H
38 PPI_PORTC DW 0032H
39 PPI_CONTROL DW 0036H
40 PIC_ICW1 DW 0040H
41 PPS1STR DW 0000H
42 PPS2STR DW 0010H
43 DATASTR DW 0020H
44 ***** PERIPHERALS *****
45 PUBLIC FROM_SEGMENT
46 PUBLIC FROM_OFFSET
47 PUBLIC RAM_SEGMENT
48 PUBLIC RAM_OFFSET
49 PUBLIC STACK_SEGMENT
50 PUBLIC STACK_POINTER
```

ORIGINAL PAGE 12
OF POOR QUALITY

LOC OBJ LINE SOURCE

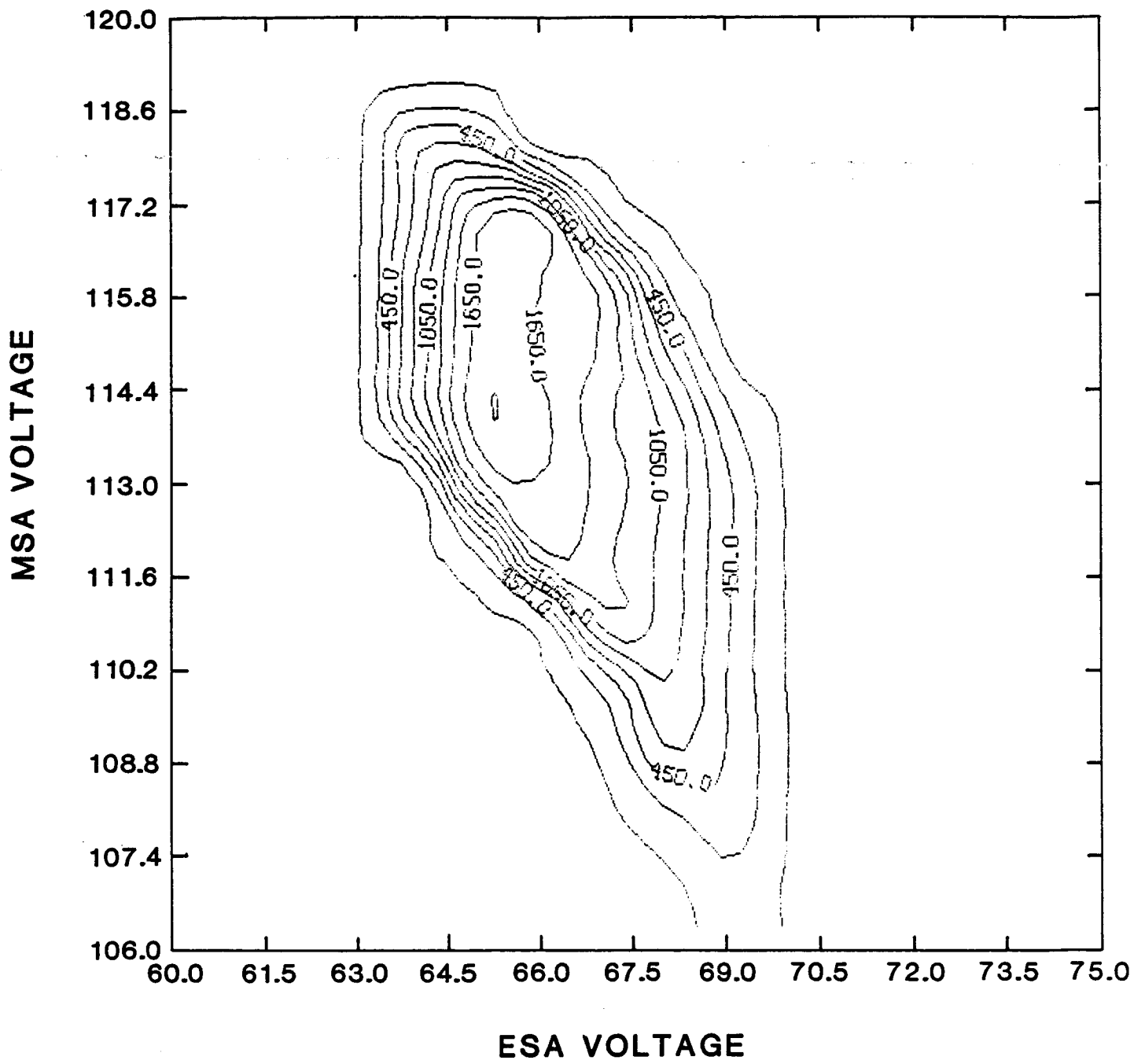
```
0026 B8----- R 97 MOV AX, DATA
0029 SED8----- R 98 MOV DS, AX
002B A10000 R 99 MOV AX, WORD PTR PERIPHERALS
002E SEC0----- R 100 MOV ES, AX
0030 A10A00 R 101 MOV AX, WORD PTR STACK_SEGMENT
0033 SED0----- R 102 MOV SS, AX
0035 A10C00 R 103 MOV AX, WORD PTR STACK_POINTER
0038 8BE0----- R 104 MOV SP, AX
003A E80000 E 105 CALL S8259
003D E80000 E 106 CALL S8255
0040 E80000 E 107 CALL JTABLE
0043 8D3E1A00 R 108 MOV DI, WORD PTR PPS2STR
0047 2EA10000 R 109 MOV AX, WORD PTR ENERGY_COMMANDS
004B F7D0----- R 110 NOT AX
004D 268205----- R 111 MOV ES:DI, AX
0050 883E1800 R 112 MOV DI, WORD PTR PPS1STR
0054 2EA10800 R 113 MOV AX, WORD PTR MASS_COMMANDS
0058 F7D0----- R 114 NOT AX
005A 268205----- R 115 MOV ES:DI, AX
005D 883E1C00 R 116 MOV DI, WORD PTR DATASTR
0061 B80000----- R 117 MOV AX, 0
0064 268905----- R 118 MOV ES:DI, AX
0067 EB----- R 119 STI
0068 E4----- R 120 HLT
0069 EBEC----- R 121 JMP TURN_ON_INTERRUPTS
----- R 122 ENDS
----- R 123 END
```

ASSEMBLY COMPLETE. NO ERRORS FOUND

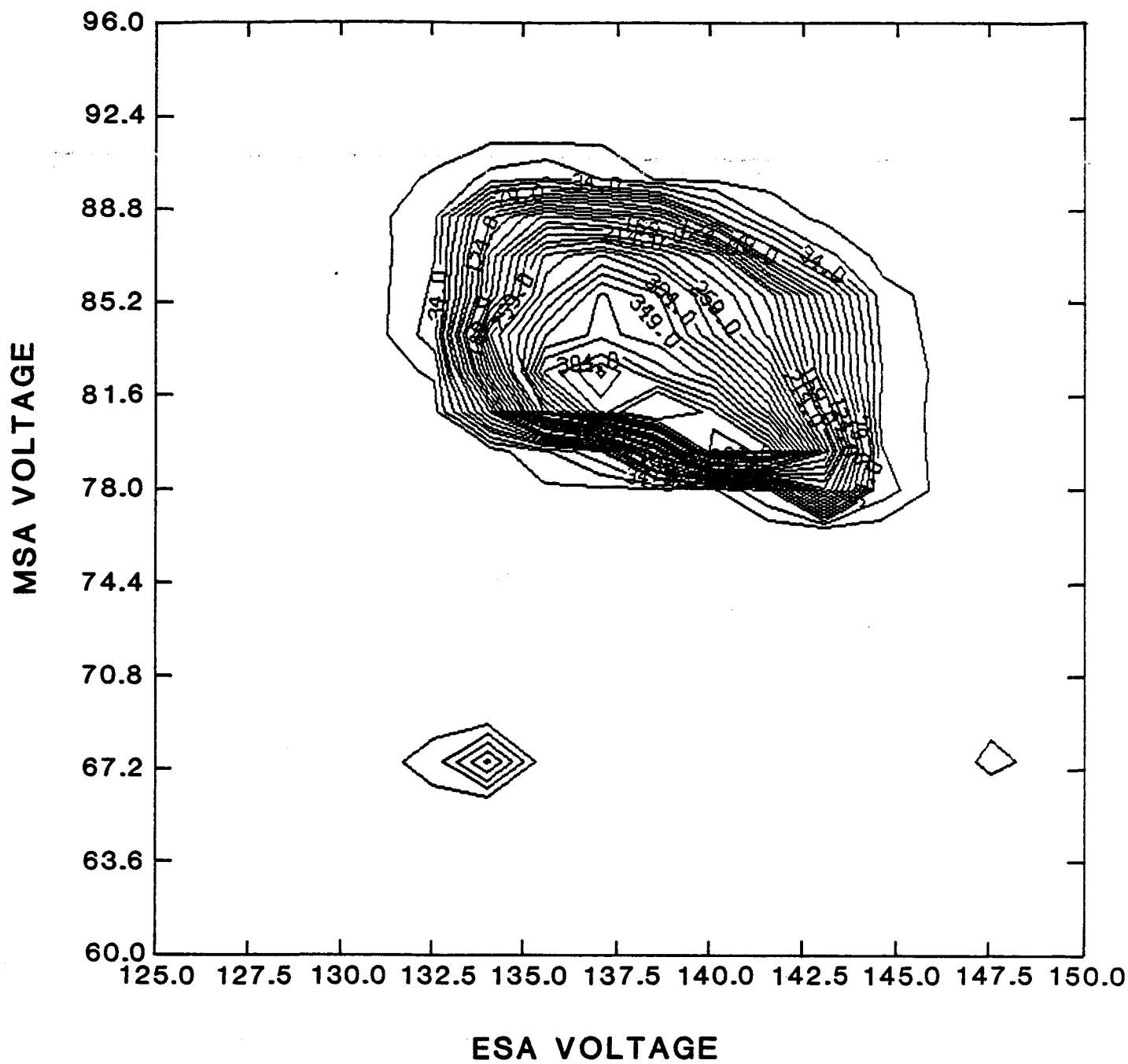
ORIGINAL PAGE IS
OF POOR QUALITY

APPENDIX C
Lab Data Plots

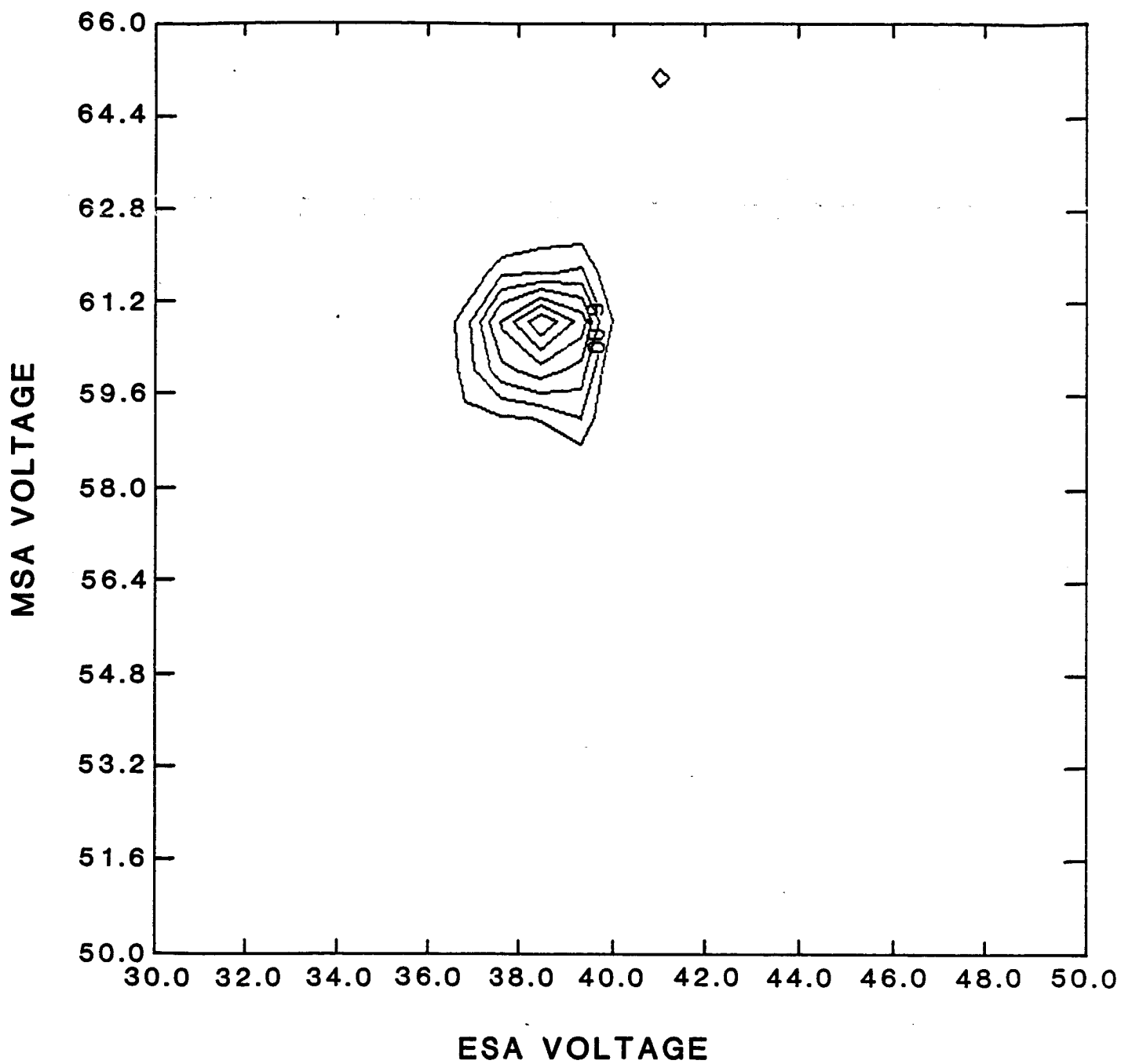
INNER CHANNEL H₂ 1 KeV



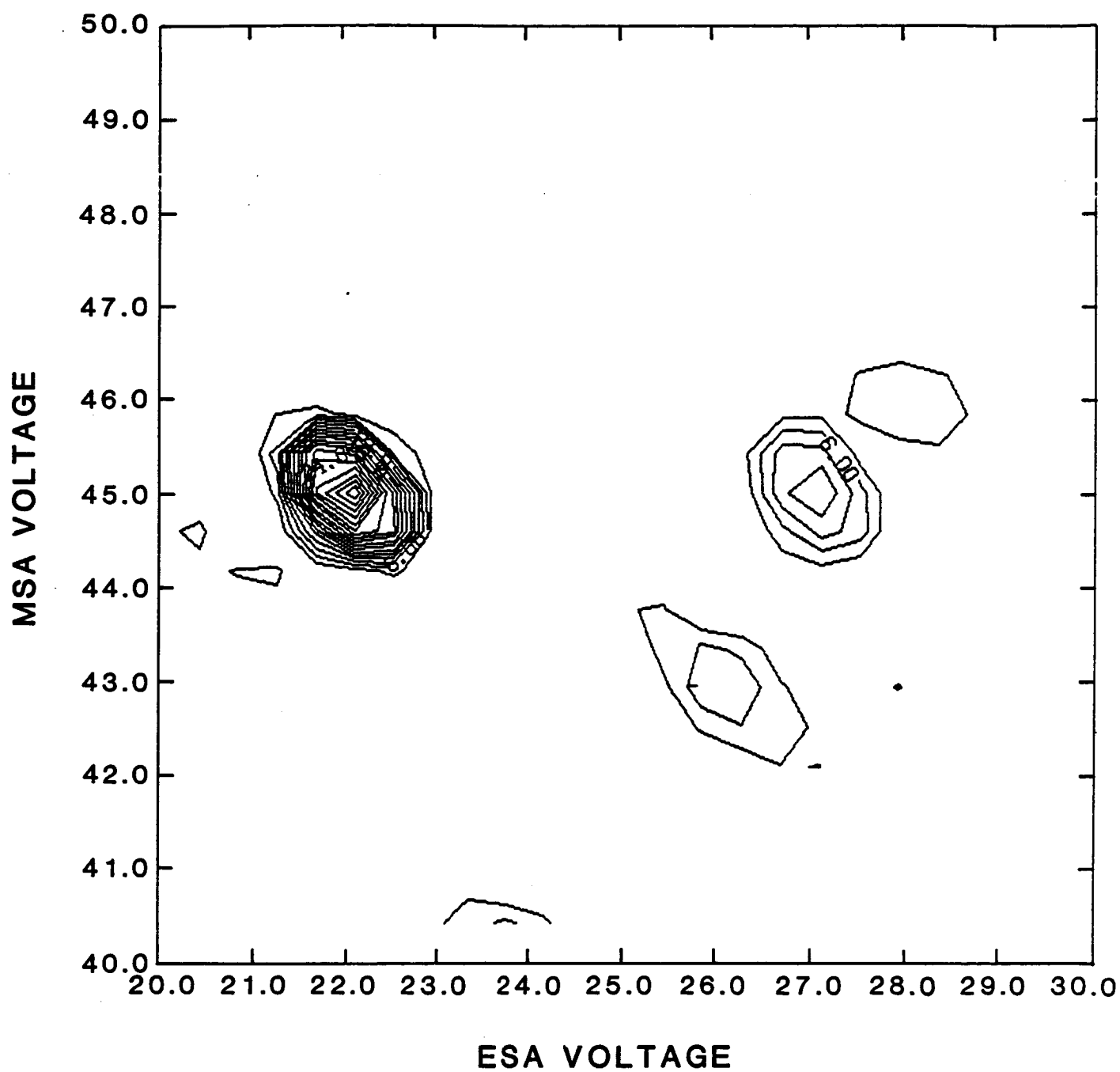
OUTER CHANNEL H₂ 1 KeV



OUTER CHANNEL N₂ 500eV



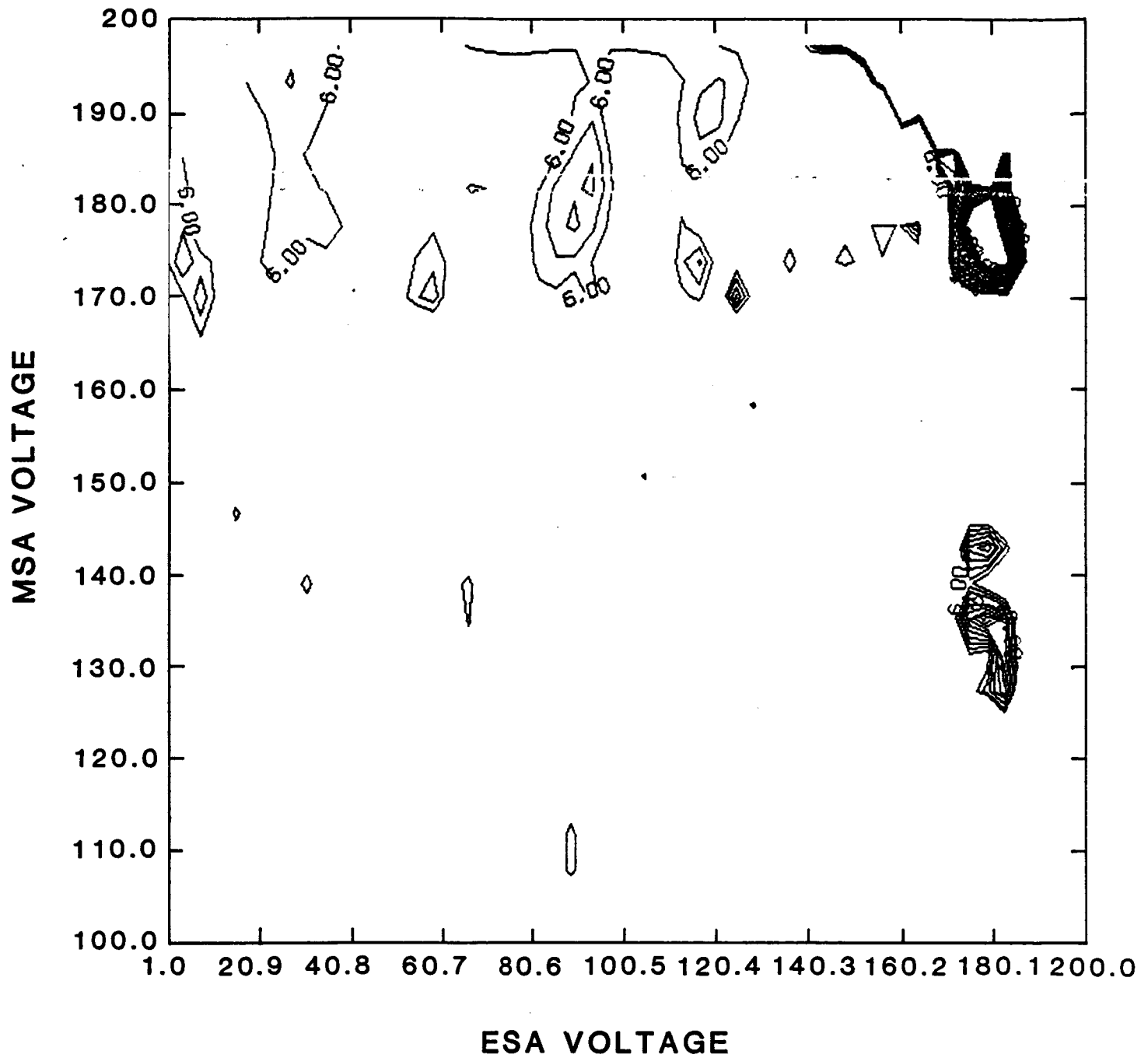
OUTER CHANNEL N₂ 300eV



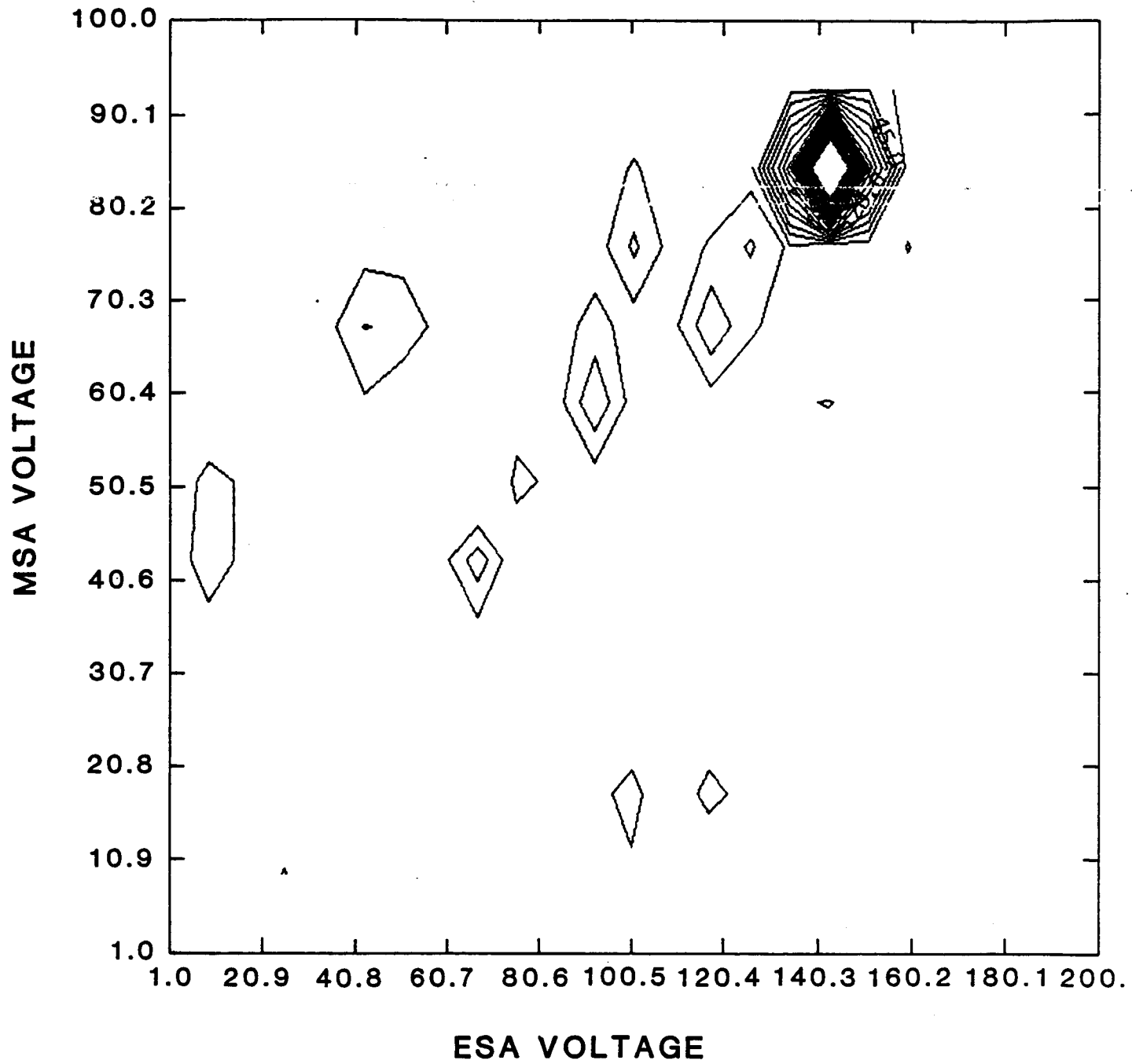
OUTER CHANNEL N_2 1 KeV



2 KeV N₁⁺ N₂⁺
WIDE SCAN FOR GHOST PEAKS



INNER CHANNEL H₂ 2 KeV



C-2

APPENDIX D

Listing of Power Supply Settings

FIMS COMMAND TABLES

ENERGY	ENERGY PPS	MASS PPS	PROM 1	PROM2	SPECIES
1.000	.150	25.063	0000	0203	30
1.200	.150	25.077	0000	0203	30
1.439	.150	25.092	0000	0203	30
1.727	.150	25.111	0000	0203	30
2.073	.173	25.134	000F	0203	30
2.487	.207	25.161	0021	0203	30
2.985	.249	25.193	0033	0203	30
3.582	.298	25.232	0046	0204	30
4.298	.358	25.279	0058	0204	30
5.158	.430	25.336	006A	0204	30
6.189	.516	25.403	007D	0204	30
7.427	.619	25.484	008F	0205	30
8.913	.743	25.582	00A1	0205	30
10.695	.891	25.679	00B4	0205	30
12.834	1.070	25.839	00C6	0206	30
15.401	1.283	26.008	00DB	0207	30
18.481	1.540	26.210	00EB	0207	30
22.177	1.848	26.453	00FD	0208	30
26.613	2.218	26.746	010F	0209	30
31.935	2.661	27.097	0122	020B	30
38.322	3.194	27.519	0134	020C	30
45.987	3.832	28.027	0146	020E	30
55.184	4.599	28.638	0158	0210	30
66.221	5.518	29.373	016B	0213	30
79.465	6.622	30.258	017D	0216	30
95.358	7.947	31.324	018F	0219	30
114.430	9.536	32.610	01A2	021D	30
137.316	11.443	34.162	01B4	0222	30
164.779	13.732	36.035	01C6	0227	30
197.734	16.478	38.299	01D9	022E	30
237.281	19.773	41.038	01EB	0234	30
284.738	23.728	44.354	01FD	023C	30
341.685	28.474	48.374	0210	0245	30
410.022	34.169	53.251	0222	024F	30
492.027	41.002	59.175	0234	0259	30
590.432	49.203	66.377	0247	0265	30
708.518	59.043	75.138	0259	0271	30
850.222	70.852	85.804	026B	027F	30
1020.266	85.022	98.796	027E	028D	30
1224.320	102.027	114.624	0290	029C	30
1469.184	122.432	133.913	02A2	02AB	30
1763.020	146.918	157.416	02B5	02BC	30
2115.624	176.302	186.048	02C7	02CC	30
1.000	.150	23.077	0000	01FB	28
1.200	.150	23.089	0000	01FB	28
1.439	.150	23.105	0000	01FB	28
1.727	.150	23.123	0000	01FB	28
2.073	.173	23.145	000F	01FB	28
2.487	.207	23.172	0021	01FB	28
2.985	.249	23.204	0033	01FB	28
3.582	.298	23.242	0046	01FB	28
4.298	.358	23.288	0058	01FC	28
5.158	.430	23.344	006A	01FC	28
6.189	.516	23.410	007D	01FC	28
7.427	.619	23.490	008F	01FC	28
8.913	.743	23.585	00A1	01FD	28
10.695	.891	23.700	00B4	01FD	28
12.834	1.070	23.838	00C6	01FE	28
15.401	1.283	24.003	00DB	01FF	28
18.481	1.540	24.202	00EB	01FF	28
22.177	1.848	24.441	00FD	0200	28
26.613	2.218	24.728	010F	0203	28

31 Dec 85

Fims-C

For 815V float

Fims 3 Program

Cal Prom

(94.4-182.1) m -
(99.1-183.7) m +
(95.2-184.4) m +
(98.2-185.8) m -

Cal Prom

(86.4-185.0) E -
(86.2-184.3) E +

ORIGINAL PAGE IS
OF POOR QUALITY

38.322	3.194	25.487	0134	0205	28
45.987	3.832	25.986	0146	0207	28
55.184	4.599	26.586	0158	0209	28
66.221	5.518	27.308	016B	020C	28
79.465	6.622	28.178	017D	020F	28
95.358	7.947	29.226	018F	0212	28
114.430	9.536	30.490	01A2	0217	28
137.316	11.443	32.015	01B4	021C	28
164.779	13.732	33.858	01C6	0221	28
197.734	16.478	36.085	01D9	0228	28
237.281	19.773	38.781	01EB	022F	28
284.738	23.728	42.047	01FD	0237	28
341.685	28.474	46.008	0210	0240	28
410.022	34.169	50.817	0222	024A	28
492.027	41.002	56.661	0234	0255	28
590.432	49.203	63.769	0247	0261	28
708.518	59.043	72.423	0259	026E	28
850.222	70.852	82.966	026B	027B	28
1020.266	85.022	95.816	027E	028A	28
1224.320	102.027	111.483	0290	0299	28
1469.184	122.432	130.588	02A2	02A9	28
1763.020	146.918	153.884	02B5	02B9	28
2115.624	176.302	182.282	02C7	02CA	28
1.000	.150	.150	0000	0000	2
1.200	.150	.150	0000	0000	2
1.439	.150	.150	0000	0000	2
1.727	.150	.150	0000	0000	2
2.073	.150	.150	0000	0000	2
2.487	.166	.150	000B	0000	2
2.985	.199	.150	001D	0000	2
3.582	.239	.150	002F	0000	2
4.298	.287	.150	0042	0000	2
5.158	.344	.150	0054	0000	2
6.189	.413	.150	0066	0000	2
7.427	.495	.150	0079	0000	2
8.913	.594	.150	008B	0000	2
10.695	.713	.150	009D	0000	2
12.834	.856	.150	00AF	0000	2
15.401	1.027	.150	00C2	0000	2
18.481	1.232	.150	00D4	0000	2
22.177	1.478	.150	00E6	0000	2
26.613	1.774	.150	00F9	0000	2
31.935	2.129	.150	010B	0000	2
38.322	2.555	.150	011D	0000	2
45.987	3.066	.150	0130	0000	2
55.184	3.679	.150	0142	0000	2
66.221	4.415	.150	0154	0000	2
79.465	5.298	.150	0167	0000	2
95.358	6.357	.150	0179	0000	2
114.430	7.629	.150	018B	0000	2
137.316	9.154	.150	019E	0000	2
164.779	10.985	.150	01B0	0000	2
197.734	13.182	.150	01C2	0000	2
237.281	15.819	.150	01D5	0000	2
284.738	18.983	.150	01E7	0000	2
341.685	22.779	.150	01F9	0000	2
410.022	27.335	.150	020C	0000	2
492.027	32.802	.150	021E	0000	2
590.432	39.362	.150	0230	0000	2
708.518	47.235	.150	0243	0000	2
850.222	56.681	.150	0255	0000	2
1020.266	68.018	.150	0267	0000	2
1224.320	81.621	.150	027A	0000	2
1469.184	97.946	.150	028C	0000	2
1763.020	117.535	.150	029F	0000	2

ORIGINAL PAGE IS
OF POOR QUALITY

7.5

2115.824	141.042	26.861	0000	020A	32
1.000	.150	26.874	0000	020A	32
1.200	.150	26.890	0000	020A	32
1.439	.150	26.910	0000	020A	32
1.727	.150	26.933	000F	020A	32
2.073	.173	26.960	0021	020A	32
2.487	.207	26.993	0033	020A	32
2.985	.249	27.033	0046	020B	32
3.582	.298	27.081	0058	020B	32
4.298	.358	27.138	006A	020B	32
5.158	.430	27.207	007D	020B	32
6.189	.516	27.289	008F	020B	32
7.427	.619	27.388	00A1	020C	32
8.913	.743	27.507	00B4	020C	32
10.695	.891	27.650	00C6	020D	32
12.834	1.070	27.821	00D8	020D	32
15.401	1.283	28.027	00EE	020E	32
18.481	1.540	28.274	00FD	020F	32
22.177	1.848	28.571	010F	0210	32
26.613	2.218	28.928	0122	0211	32
31.935	2.661	29.357	0134	0213	32
38.322	3.194	29.873	0146	0215	32
45.987	3.832	30.494	0158	0217	32
55.184	4.599	31.241	016B	0219	32
66.221	5.518	32.140	017D	021C	32
79.465	6.622	33.223	018F	021F	32
95.358	7.947	34.529	01A2	0223	32
114.430	9.536	36.104	01B4	022B	32
137.316	11.443	38.005	01C6	022D	32
164.779	13.732	40.302	01D9	0233	32
197.734	16.478	43.079	01EB	0239	32
237.281	19.773	46.441	01FD	0241	32
284.738	23.728	50.514	0210	0249	32
341.685	28.474	55.454	0222	0253	32
410.022	34.169	61.451	0234	025D	32
492.027	41.002	68.736	0247	0268	32
590.432	49.203	77.594	0259	0274	32
708.518	59.043	88.372	026B	0282	32
850.222	70.852	101.492	027E	028F	32
1020.266	85.022	117.466	0290	029E	32
1224.320	102.027	136.920	02A2	02AE	32
1469.184	122.432	160.611	02B5	02BE	32
1763.020	146.918	189.455	02C7	02CE	32

ORIGINAL PAGE IS
OF POOR QUALITY

1.000	.150	4.156	0000	014E	16
1.200	.150	4.167	0000	014F	16
1.439	.150	4.179	0000	014F	16
1.727	.150	4.195	0000	014F	16
2.073	.173	4.213	000F	0150	16
2.487	.207	4.235	0021	0150	16
2.985	.249	4.261	0033	0151	16
3.582	.298	4.292	0046	0152	16
4.298	.358	4.330	0058	0152	16
5.158	.430	4.375	006A	0153	16
6.189	.516	4.430	007D	0155	16
7.427	.619	4.495	008F	0156	16
8.913	.743	4.573	00A1	0158	16
10.695	.891	4.668	00B4	015A	16
12.834	1.070	4.781	00C6	015C	16
15.401	1.283	4.917	00D8	015F	16
18.481	1.540	5.080	00EB	0163	16
22.177	1.848	5.277	00FD	0166	16
26.613	2.218	5.513	010F	016B	16
31.935	2.661	5.797	0122	0170	16
38.322	3.194	6.139	0134	0176	16
45.987	3.832	6.551	0146	017C	16